



# DESIGN AND INSTRUMENTATION OF AN EARTH SHOCK TUBE

**Final Report** 

Ву

Atlantic Research Corporation

A Division of The Susquehanna Corporation

February 1968

Submitted to

The United States Army
Ballistics Research Laboratory
Aberdeen Proving Ground
Maryland

Contracts

DA-36-034-509-ORD-3116RD DA-18-001-AMC-877(X) DDDC

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#### ABSTRACT

This report, which is divided into two parts, covers the work done under Contracts DA-36-034-509-ORD-3116RD and DA-18-001-AMC-877(X).

The major objective of this work was to design an instrumented earth shock tube for one dimensional wave propagation studies. in a shock tube the lateral strains imposed when the load is applied should not interfere with the axial motion of the soil particles. However, the frictional effects at the walls of the confining tube are so great that the ideal situation is difficult to achieve. an attempt was made to reduce these frictional effects by placing Teflon powder between the soil sample and the confining tube. Testing of this system indicated that imbedded stress gage behavior should be investigated before continuing further testing of the earth shock tube. two laboratory devices, the dynamic oedometer and the dynamic triaxial apparatus, were designed for investigating imbedded stress gage behavior as well as for other studies involving the behavior of soils under dvnamic loadings. Interesting results were obtained from all the testing completed; however. it is concluded that further investigations are required to complete the evaluation of the above systems.

During the course of these investigations an earth strain gage and a horizontal displacement meter were developed. However, further testing of these units is required to complete their design.

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# PART I DESIGN OF AN INSTRUMENTED EARTH SHOCK TUBE

Contract DA-36-034-509-ORD-3116RD

#### 1.0 INTRODUCTION

This report, written in two parts, describes the work performed by Atlantic Research Corporation for the Ballistic Research Laboratories under two separate contracts; namely, Contract Nos. DA-36-034-509-ORD-3116RD and DA-18-001-AMC-877(X). The second contract was actually a continuation of the work initiated under the first contract. The time period covered by the two contracts ranged from December 1959 to October 1967.

The effort expended during this time period covered several areas which developed from the investigations into the development of an earth shock tube. For purposes of clarity, the major areas investigated under these contracts are reported herein as follows:

- 1. Development of an earth strain gage.
- 2. Design, instrumentation and test of an earth shock tube.
- Design, instrumentation and evaluation of a dynamic oedometer. (1)
- 4. Development of a horizontal displacement meter. (2)
- Design, instrumentation and evaluation of a dynamic triaxial apparatus. (1)

The dynamic triaxial apparatus was never assembled and evaluated because of the limited funding available.

<sup>(1)</sup> The design of the dynamic oedometer and the dynamic triaxial apparatus was done under the direction of Dr. Werner Heierli and Alva Matthews of the Paul Weidlinger firm in New York City.

<sup>(2)</sup> A separate report was published covering the development of the horizontal displacement meter; however, for purposes of continuity, it is reproduced here in its entirety.

#### 2.0 EARTH STRAIN GAGE

#### 2.1 Design

The design of the earth strain gage is shown in Figure 1. The general configuration is two large discs connected axially by a stem of small cross-sectional It was hoped that this configuration would reduce soil disturbances caused by the presence of the gage. Encased in the stem is a Bourns Model 141 linear-motion potentiometer, whose resistance element is a carbon film. The manufacturer claims that the associated electronic equipment limits the resolution which can be obtained with this potentiometer. The stem of the transducer is encased within a cylinder which is allowed to slide over the transducer housing, and soil is prevented from entering into the sliding area by a flexible plastic tube cemented to the metal parts at each end. The force needed to overcome the inertia of the system is minimized by the use of aluminum for most metal parts. The nominal transducer slider friction is 4 ounces. The only other restraining force will be contributed by the soil contained between the two discs of the gage. Density-matching of the gage and soil can be accomplished by adding metal to, or taking metal from, the gage housing.

#### 2.2 Circuitry

The gage can be incorporated into several types of electrical circuits. It can be used as a voltage divider, of course, with the slider position indicated as a variable voltage-drop. In addition, it can be considered a variable resistor and used in an a.c. or a d.c. resistance bridge. Hence the earth strain gage can be readily used in several standard measurement and recording systems.

#### 2.3 Evaluation

These gages were utilized in the tests conducted in the earth shock tube. Test results are tabulated and discussed in the following section.

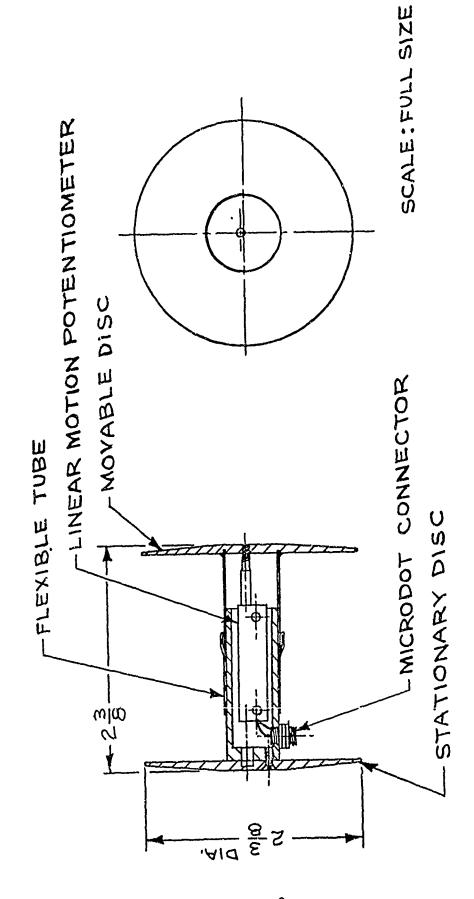


Figure 1. Earth Strain Gage.

#### 3.0 EARTH SHOCK TUBE

#### 3.1 Friction Investigation

In earlier work on the dynamic behavior of soils; e.g., Reference (1), consideration has been given to wave propagation studies using shock tube techniques. Ideally, the lateral strains imposed when the load is applied should not interfere with the axial motion of the particles. However, the frictional effects at the walls of the confining tube were so great that the ideal situation was difficult to achieve. It had been reported that lubrication reduces the wall friction; so it seemed worthwhile to investigate various lubrication techniques. If a large reduction in wall friction could be achieved, the earth shock tube could take a simple form.

#### 3.1.1 Sliding Plate Experiments

In these experiments, a metal plate with various surface treatments was caused to slide over the test surface. The forces required to start the motion and to maintain the motion were measured with a Hunter Force Indicator, Model L-20. Moisture content and compaction of the soil were not controlled, although a new surface of sand was exposed for each sliding test. The measurements, reported as the average of a number of trials along with the maximum observed deviation from the average, are consistent enough to warrant the conclusion that differences in observed frictional forces greater than 10% are probably real.

In the first series of experiments, a metal plate with different surface treatments was caused to slide across a surface of dry sand. The plates are described in Table I, and the measured frictional

<sup>(1)</sup> Massachusetts Institute of Technology; the Behavior of Soils under Dynamic Loadings; III Final Report on Laboratory Studies; Contract DA-49-129-ENG-227, Office of the Chief of Engineers, AFSWP-118; August, 1954.

### TABLE I

### Description of Plates for Table II

Plate 1	Cold rolled steel - frictional surface as received from stock - dimensions 1/2 inch by 6 inches by 6 inches with a 1/2-inch radius on leading edge; weight 5.1 pounds
Plate 2	Cold rolled steel - frictional surface 16 microinch ground finish - dimensions 1/2 inch by 6 inches by 6 inches with a 1/2-inch radius on leading edge; weight 5.0 pounds
Plate 5	Plate 2 with a coat of buffed Dyer's general purpose paste wax.
Plate 6	Plate 1 with a coat of buffed Dyer's general purpose paste wax.
Plate 9	Cold rolled steel - frictional surface Teflon with approximately a cold rolled finish - dimensions 1/2 inch by 6 inches by 6 inches with a 1/2-inch radius on leading edge; weight 5.0 pounds
Plate 10	Plate 2 with a light coat of dry molybdenum disulfide.
Plate 11	Plate 1 with a light coat of dry molybdenum disulfide.
Plate 12	Plate 2 with a light coat of powdered graphite.
Plate 13	Plate 1 with a light coat of powdered graphite.
Plate 14	Plate 2 with a light coat of S-73 plasticizer.

forces are given in Table II. These data indicate the following: Treatment of the metal surface alone can reduce friction by about 50%; the treatments tested were not effective unless the metal surface was polished.

In the second series of experiments, four metal surfaces were tested by sliding over a thin layer of granular material spread over the sand to form a lubricating layer. The metal surfaces are described in Table I and the lubricating layers in Table III. The measurements are reported in Table IV. The data indicate that the friction can be reduced by two-thirds or more by using a lubricating layer of Teflon powder inserted between the soil and the confining wall; a Teflon-coated rough metal surface was almost as effective as a polished metal surface, so far as friction reduction was concerned.

Use of a polished metal surface or a Teflon-coated metal surface to confine the earth in a shock tube might not be practical from the cost point of view. Hence, a series of experiments was carried out with the sliding plate to test the frictional effects of various surface coatings, using cold rolled steel in the "as-received" condition as the base material. The results, given in Table V, show that cellophane sheet and two of the lacquers give as low frictional resistance as the polished metal plate.

#### 3.1.2 Tube Experiments

Friction forces in tubes with an inside diameter of 2.5 inches were next investigated, as an extension of the sliding-plate experiments. Samples of dry and of moist sand were compacted in tubes of various surface finishes, with and without lubricating layers of Teflon powder. The length-to-diameter ratio of the sand column was varied from 5.9 to 1.4 by filling a longer section of the same tube.

TABLE II

Effect of Surface Treatment of Metal Plate
on Frictional Resistance with Dry Sand

		Stat	tic	Dynamic		
Plate No.	No. of Trials	Ave. Static Frictional Resistance (lbs)	Maximum Deviation from Ave. (lbs)	Ave. Dynamic Frictional Resistance (1bs)	Maximum Deviation from Ave. (1bs)	
1	10	2.3	0.2	2.3	0.1	
2	10	1.6	0.2	1.3	0.2	
6	10	1.9	0.3	1.7	0.2	
5	10	1.2	0.2	1.0	0.2	
9	25	2.2	0.2	2.1	0.3	
10	25	1.1	0.2	1.1	0.2	
11	25	2.1	0.2	2.1	0.3	
12	25	1.1	0.1	1.1	0.2	
13	25	2.2	0.3	2.1	0.3	
14	30	1.6	0.4	1.5	0.5	

<sup>(1)</sup> See Table I for plate description.

#### TABLE III

#### Lubricating Layer Descriptions and Notes

- 1. Glass bead layer, 1/16" thick, 15 micron to 25 micron size range.
- 2. Glass bead layer, 1/16" thick, 350 micron to 500 micron size range.
- 3. Sand-graphite mixture, 1/8" thick layer, 5% graphite by volume.
- 4. Glass bead-molybdenum disulfide mixture, 1/16" thick layer, glast bead size range 350 micron to 500 micron, 1% molybdenum disulfide by volume.
  - 5. Teflon powder layer, 1/8" thick, grain size unknown.
- 6. Teflon powder layer, same as Number 5. Layer was placed on moist sand in a shallow pan and then sealed with aluminum foil. The assembly remained intact for 3 days before the tests were conducted. Initial and final weight checks of the pan revealed no loss of weight after the 3-day period, within the accuracy of the measurement method. However, the surface of the lubricating layer may have dried out during the friction measurements.

TABLE IV

Effect of Lubricating Layers on Frictional Resistance

		Static		Dynamic			
Plate (1	) <sub>Layer</sub> (2) No.	No. of Trials	Sand Condition	Ave. Static Frictional Resistance (1bs)	Maximum Deviation from Ave. (lbs)	Ave. Dynamic Frictional Resistance (1bs)	Maximum Deviation from Ave. (lbs)
1	1	10	Dry	2.1	0.1	2.0	0.1
2	1	10	Dry	1.9	0.1	1.8	0.1
9	1	10	Dry	2.0	0.2	2.0	0.1
10	1	10	Dry	1.7	0.1	1.7	0.1
1	2	10	Dry	1.5	0,1	1.3	0.1
2	2	10	· Dry	1.2	0.1	1.1	0.1
9	2	10	Dry	1.4	0.1	1.4	0.1
10	2	10	Dry	1.1	0.1	1.1	0.1
1	3	10	Dry	2.5	0.1	2.4	0.2
2	3	15	Dry	1.7	0.2	1.7	0.2
9	3	10	Dry	2.0	0.1	2.0	0.1
1	1;	10	Dry	1.5	0.1	1.5	0.1
2	4	10	Dry	0.9	0.2	0.9	0.2
9	4	10	Dry	1.1	0.1	. 1.1	0.1
10	4	10	Dry	0.9	0.2	0.9	0.2
1	5	10	Dry	1.0	0.2	0.9	0.1
2	5	10	Dry	$\mathbf{o.}\epsilon$	0.1.	0.6	0.1
9	5	10	Dry	0.7	0.3	ο.€	0.1
9	6	10	Moist	1.0	0.1	0.9	0.1
1	$\epsilon$	10	Moist	0.9	0.2	0.8	
2	6	10	Moist '	0.6	0.1	0.6	
9	6	10	Moist	0.7	0.3	0.7	0.1

<sup>(1)</sup> See Table I for plate description.

<sup>(2)</sup> See Table III for layer description.

TABLE V

#### Effect of Surface Treatment of Metal Plate on Frictional Resistance with Dry Sand and Teflon Powder Layer

Cold rolled steel with "as received" surface
Dimensions: 1/2 inch by 6 inches by 6 inches with 1/2

inch radius on leading edge; weight 5.0 lbs

Send Condition: Dry

Lubricating Layer: Teflon Powder No. 1

(600 micron grain size)

		Static		Dynami	C	
Frictional Surface	No. of Trials	Ave.Static Frictional Resistance (129)	Maximum Deviat. fr. Ave. (1bs)	Ave. Dynamic Frictional Resistance (1bs)	Maximum Deviat. fr. Ave. (1bs)	Remarks
Acetate sheet	70	0.8	0.1	0.8	0.2	0.007" thick surface glossy
Cellophane sheet	10	0.6	0.1	0.6	0.1	0.002" thick, surface glossy
Mylar sheet	10	1.3	0.1	1.2	0.1	0.001" thick, surface glossy with wrinkles
Mylar sheet	10	0.8	0.2	0.8	0.1	0.005" thick, surface glossy
Polyethylene sheet	10	0.9	0.1	0.9		0.005" thick, surface semi-glossy
Plexiglas sheet	10	0.8	0.1	0.8	0.1	0.020" thick, surface glossy
Polystyrene sheet	10	1.4	0.2	1.1	0.2	0.022: thick, surface satin
Aluminum foil	10	0.8	0.1	0.8	0.1	0.002" thick, surface bright with wrinkles
Areidite	10	1.5	0.3	1.5	0.3	Cured 1 hour at 150° F and 1 hour at 75° F.
Araldite	10	1.9	0.2	2.0	0.3	Preceding plate tested after 5 days.
"Krylon" Clear Acrylic Resin No. 1302	10	2.7	1.3	0.9	0.4	Three coats cured as per directions.

TABLE V (continued)

		Static		Dynamic		
Frictional Surface	No. of Trials	Ave.Static Frictional Resistance (lbs)	Maximum Deviat. fr. Ave. (1bs)	Ave.Dynamic Frictional Resistance (1bs)	Maximum Deviat. fr. Ave. (1bs)	Remarks
"Kiylon" Enamel No. 1605	10	0.9	70 AV SF	0.7	0.3	Four coats cured for 16-1/2 hours at room temperature
Lacquer, Glidden No. 2838	10	0.5	0.1	0.5	0.1	Two coats cured for 3 hours at room temperature.
Lacquer, Glidden No. 2838	10	0.7	0.1	0.8	0.1	Four coats cured for 2 hours at room temperature. Surface aust laden.
Lacquer, Glidden No. 2838	10	0.7	0.1	0.7	0.1	Two coats cured for 16 hours at room temperature. Surface dust laden.
Lacquer, Spray, Illinois Bronze Powder Co., No. 130	10	0.6	0.2	0.6		Two coats cured for 2 days at room temperature.
Glyptol No. 1202	10	1.2	0.1	1.2	0.1	Cured for 3 days at room temperature Surface dust laden.

#### Note:

The average static frictional resistance for a cold rolled steel plate with a 16-microinch finish sliding on a lubricating layer of Teflon powder, measured under comparable conditions, is 0.6 lbs; see Table IV.

To install the Teflon layer, a thin-walled aluminim tube with an outside diameter of approximately 2 1/8 inches was inserted along the axis of the larger tube and was held in place by a plug inserted in the bottom of the larger tube. The annular gap was filled with Teflon powder and the inside tube with sand. The smaller tube was then withdrawn and the contents of the larger tube compacted.

The first experiments with tubes were qualitative, to learn whether the general aspects of the sliding-plate experiments would be confirmed in tubes. After the compaction process, the tube was lifted vertically upwards, the bottom plug was removed, and the tendency of the sand column to fall out of the tube was observed. The results, given in Table VI, are consistent with the results of the sliding plate experiments and indicace that the Teflon powder layer confined by a polished metal surface does provide lubrication. There is some indication of an effect of particle size, although the corrosion of the polished steel surface may have influenced the results. Visual inspection of the cross sections of the sand columns showed a uniform layer of Teflon powder, indicating that the installation technique was satisfactory.

In the second series of experiments with tubes, an attempt was made to obtain semi-quantitative data concerning the frictional resistance. The tube was held rigidly in place, vertically, and the Hunter Force Indicator was attached to the bottom retaining plug. The bottom plug was then lowered until its top surface was level with the bottom of the tube, in order to eliminate friction between the plug and the tube. The force of the sand column on the plug was observed. The results are given in Table VII. About 70% of the weight

TABLE VI Results of Experiments on Frictional Effects in Tubes

Tube Finish	L/D(2)	Sand Condition	Lubricating Layer	Results
Cold rolled	5.8	Dry and highly compacted	None	Column fell freely out of tube.
Cold rolled	4.5	Water-saturated and highly compacted	None	Column remained intact in tube.
Cold rolled	2.3	Moist and highly compacted	None	Column remained intact in tube.
Cold rolled	1.5	Moist and highly compacted	None	Column remained intact in tube.
Cold rolled	1.4	Moist and highly compacted	Teflon powder(1)	Column remained intact in tube
16-microinch	1.5	Moist and highly compacted	Teflon powder(1)	Column slowly slid out of tube.
16-microinch	5.4	Moist and highly compacted	Teflon powder(1)	Column fell freely out of tube.
16-microinch	1.2	Moist and highly compacted	None	Column remained intact in tube.
16-microinch, pirted	5.4	Moist and highly compacted	Teflon 5	Broke in half; one half fell out of tube and other half remained intact in tube.
16-microinch, pitted	5.3	Saturated and highly compacted	Teflon 5(1)	Column slowly slid intact out of tube.
16-microinch, pitted	5.0	Moist and highly compacted	Teflon 7(1)	Column remained intact in tube.

(1) Grain sizes:
Teflon powder: size unknown
Teflon 5: 350 micron Teflon 7: 35 micron

(2) L/D: length-diameter ratio

TABLE VII

Effect of Lubricating Layer on Behavior of Sand Column in Vertical Tubes

•	Tube(1)	<u>L/D<sup>(2)</sup></u>	Teflon(3) Powder	Condition(	Column Weight (1bs)	Measured Force on Bottom of Column (1bs)	Column Weight on Bottom (%)	Remarks
3	1.	5.6	5	MC	4.0	2.2	55	
<b>.</b> I	1	5.6	5	MC	4.0	2.0	50	Column above was recompacted and test was rerun.
	2	4.8	5	MC	3.4	2.4	71	
	2	5.6	5	MC	3.6	2.4	67	
	2	5.6	5	MC	3.6	3.1	86	Side of tube was lightly tapped with hammer.
	2	5.4	1	MC	4.2	1.7	40	***
	5	4.6	1	MC	3.7	1.3	35	****
	2	3.9	1	MC	2.4	1.1	46	
	2	2.9	1	MC	1.8	0.9	50	***
<b>)</b>	2	5.2	5	DC	2.9	2.5	86	Compaction of column proved to be difficult.
	2	5.9	••	DC	4.3	0.4	9.3	Compaction of column proved to be difficult.
	3	5.5	5	MC	3.4	1.0	29	# # <b>*</b> # # #
	3	5.5	5	MC	3.4	1.9	56	Side of tube was lightly tapped with hammer.
	3	5.5	5	MC	3.4	2.5	74	Measurement was made after 1-1/2 hours.
	4	5•5	5	MC	4.0		₩.**	Column did not slide out of tube.
	4	5.5	5	МС	4.0	1.3	32	Column was push- ed slightly to start.
					17			

#### TABLE VII (continued)

 Tube(1)	L/D(2)	Teflon(3) Powder	Condition (4	Column Weight (1bs)	Measured Force on Bottom of Column (1bs)	Column Weight on Bottom (%)	Remarks
4	5.5	5	MC	4.0	1.4	35	Measurement made after 1/2 hour.
4	5.5	5	MC	4.0	1.4	35	***
5	5.5	5	SC	3.6	0.6	17	Teflon powder wet
5	5.8	5	MC	3.4	1.5	1111	
5	5.8	5	MC	3.4	1.4	41	
5	5.8	5	MC	3.4	1.7	50	A0 40 40 40 40 40
5	5.8	5	MC	3.4	1.8	53	a, 20 ay 40 ay 44
5	5.8	5	MC	3.4	1.4	41	
5	5.8	5	MC	3.4	1.6	47	49 40 40 40 40

#### Notes:

More		
(1)	Tube No.	Description
	1	Cold rolled steel, inside surface Teflon coated, tube originally honed to 16-microinch surface
	2	Cold rolled steel, inside surface honed to a 16-microinch finish
	3	Cold rolled steel, inside surface honed to a 16-microinch surface and then bright chrome plated, thickness 0.00001 inch.
	4	Cold rolled steel, inside surface honed to a 16-microinch surface and then copper and nickel plated as per Federal Specification QQ-N-290, Class I, Type II, and then chrome plated as per Federal Specification QQ-C-320 Class I, Type I. Frictional surface had a smooth bright chrome finish.
	5	Stainless steel, Type 304; inside surface honed to a 13-microinch surface.

- (2) L/D: Length-diameter ratio of column
- (3) Teflon powder grain sizes
  Teflon 1: 600 microns
  Teflon 5: 350 microns
- (4) MC: Moist and compacted
  DC: Dry and compacted
  - SC: Saturated and compacted

of a column of moist and compacted sand was supported by the bottom plug, when a layer of Teflon powder No. 5 was inserted between the sand and the polished cold rolled steel confining wall. Without the lubricating Teflon layer, there would have been no force on the bottom plug.

Finally, three tests were conducted to measure the pushing force required to move a column of moist compacted sand, with and without the Teflon powder layer, in a horizontal tube. The tube assemblies were prepared in the manner described above. The bottom plug was removed, and the cavity end of the tube was placed against a rigid retaining wall. The plug, with Hunter Force Indicator attached, was then used to start motion from the other end of the tube. The observations are presented in Table VIII. For these particular experimental conditions, which may not be optimum for low friction, 'he Teflon powder lubricating layer reduced the friction to less than 10% of the value measured without the lubricating layer.

#### 3.1.3 Dynamic Tests

A few tests under dynamic conditions were carried out with an accelerometer buried in the sand column in the 2 1/2-inch-diameter tube. A falling weight applied a force pulse to the sand column, and the output of the accelerometer was displayed on an oscilloscope. The wave form was fairly well reproduced in successive trials. However, the amplitudes varied somewhat, attributed to variations in the compaction of the sand and in the nature of the applied pulse. It seemed unlikely that this technique could be used successfully in evaluating the effectiveness of the lubricating layer, so the tests were discontinued. It would probably be better to use a tube large enough so that several accelerometers, placed at different locations with respect to the wall, could be used simultaneously.

TABLE VIII
Frictional Resistance Effects in Horizontal Tubes

Tube(1)	<sub>L/D</sub> (2)	Teflon(3) Powder	Condition(4) of Sand	Column Weight (1bs)	Static Pushing Force (1bs)
4	5.9	NONE	MC	4.5	Exceeded 20 <sup>(5)</sup>
5	€.9	NONE	MC	4.7	Exceeded 20 <sup>(5)</sup>
5	5,8	5	MC	3.4	1.8

#### Notes:

- (1) See Table VII for tube description.
- (2) L/D: Length-diameter ratio of column.
- (3) Teflon 5: 350 micron grain size.
- (4) MC: moist and compacted.
- (5) Limit of gage is 20 pounds.

#### 3.2 Force Application Hyge Shock Tester

The earth shock tube must include a method for applying a force to the soil surface. The force should rise to its maximum value in a few milliseconds and maintain that value approximately constant for a minimum of 30 milliseconds and preferably for a much longer time. The pressure s ould be about 500 psi. The Hyge Shock Tester seemed to meet these requirements.

#### 3.2.1 Principle of Operation

A diagram of the Hyge Shock Tester is shown in Figure 2. The Tester is basically a cylinder divided into two chambers by an orifice plate. Chamber A is sealed from Chamber B by two sealing rings, one at the periphery of the thrust piston and the other at the relatively small orifice leading into Chamber B. When Chamber A is pressurized to a low value the thrust piston seats itself against the orifice opening, thus exposing a small piston area to Chamber B. High values of pressure, therefore, are needed in Chamber B to balance the forces on the piston produced by the low gas pressure in Chamber A. A slight increase in pressure over the equilibrium pressure moves the thrust piston in Chamber A, and the seal at the orifice is destroyed. The high pressure in Chamber B then acts on the entire surface of the thrust piston and rapidly moves the piston further into Chamber A. The acceleration and deceleration of the thrust piston is precisely controlled by metering the flow of gases. The thrust pulse is transferred to the thrust column which can be attached to any desired object.

The Model HY6407 Hyge Shock Tester, available at Atlantic Research Corporation, can produce a thrust of 40,000 pounds with a thrust column travel of a fraction of an inch. This force applied to a column of soil ten inches in diameter would give a

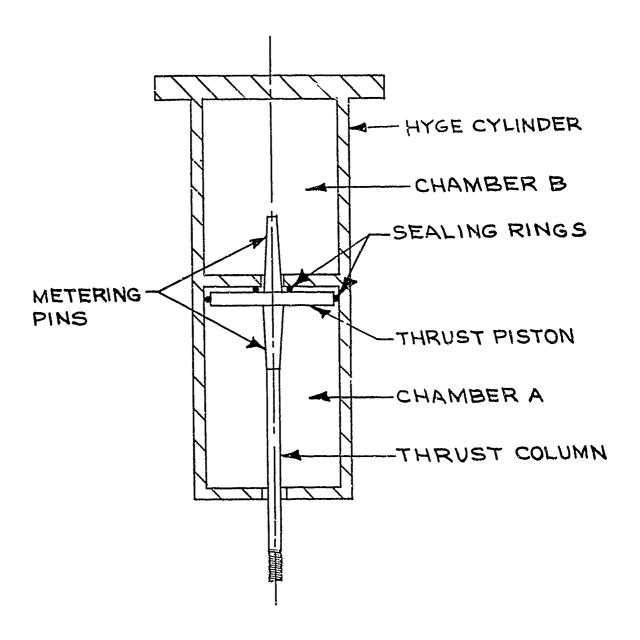


Figure 2. Hyge Shock Cylinder.

pressure somewhat greater than 500 psi. The rise times of thrust pulses produced by this model have been reported to be between 5 milliseconds and 10 milliseconds; however, shorter rise times were possible if a specially designed metering pin was used. These performance characteristics appeared suitable for the purpose described here.

#### 3.3 Modification and Mounting of the Hyge Shock Tester

#### 3.3.1 Hyge Cylinder

The cylinder end of the Hyge unit was securely fastened to a rigid cross-member composed of two 12-inch I-beams (Figure 3). Under a dynamic load of 40,000 pounds, the theoretical deflection of this composite beam, at the Hyge unit, was 0.03 inch. A Baldwin-Lima-Hamilton, Type C, wire strain gage load cell was attached to the thrust column to record the wave form of the input pulse. The cell was capable of measuring 50,000 pounds of thrust within an accuracy, as stated by the manufacturer, of ± .25% of full-scale output at 70° Fahrenheit. The loading button of the load cell bears on a replaceable heat treated plate which is attached to the semi-floating piston assembly.

#### 3.3.2 Structural Steel Components

All structural steel material used in the structure conformed to the American Society for Testing Materials "Standard Specifications for Structural Steel for Bridges and Buildings, Serial Designation A-7." The slenderness ratios of all columns for both compression and tension were well within the values specified by the American Institute Of Steel Construction. Plumbing of the main columns and the earth shock tube support was facilitated by use of metal plates imbedded in the concrete. Lateral motions of the cross-member were minimized by rods attached to the 12-inch I-beams on the inside of the main columns. Removal of the cross-member

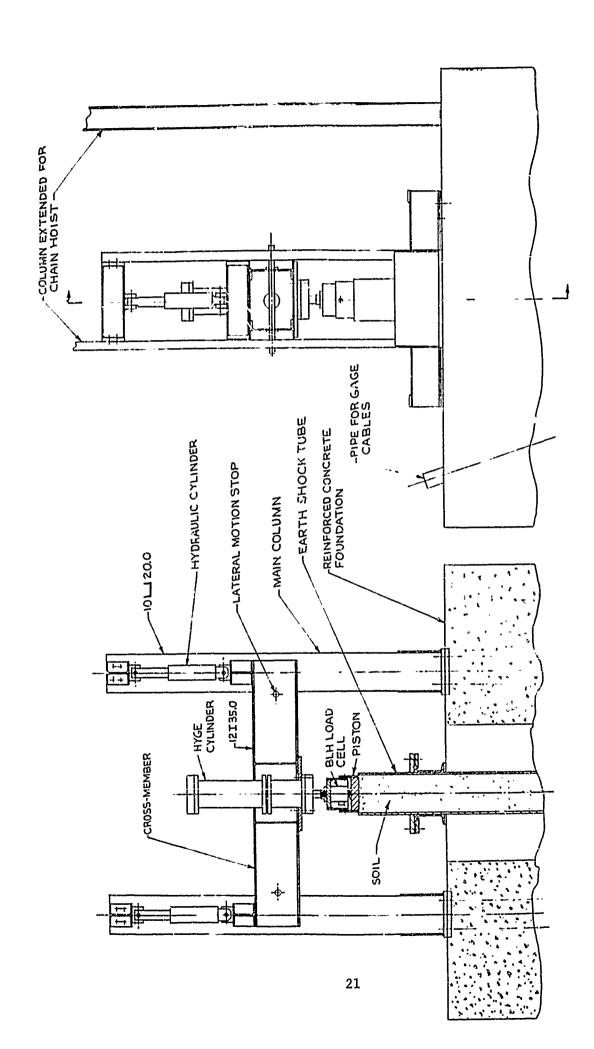


Figure 3. Hyge System for Soil Measurement Program.

from the structure was accomplished by removing the lateral motion stops and disconnecting the lower hydraulic cylinder pins. The Hyge cylinder could then by removed from the cross-member. A structural arrangement (not completely shown in the sketch) supported a chain hoist which was used to handle the heavier components of the system.

#### 3.3.3 Hydraulic Jacks

Two Dynex, Inc., high-pressure, double-acting cylinders (Figure 4) were used both to transmit the Hyge load to the main columns and to adjust vertically, within 9 inches, the Hyge cylinder to accommodate different soil sample heights in the shock tube. These cylinders could be used to preload the soil in the shock tube to simulate conditions at various depths in the earth. Each cylinder was capable of withstanding 22 tons of force in the push direction and 6 tons of force in the opposite direction at a cylinder pressure of 6000 psi. The manufacturer stated that the cylinder would not collapse under load when fully extended.

To move the cylinders, a Dynex double-acting hydraulic hand pump was used. With a fluid displacement of 0.294 cubic inch per stroke, 22 strokes of the pump were required to move the cylinder 9 inches. Handle effort was determined by the acting pressure, with a maximum of 79.2 pounds at a pressure of 6000 psi.

#### 3.3.4 Concrete Foundation

The foundation for the shock tube and Hyge Tester was built in accordance with suggestions made by the Consolidated Electrodynamics Corporation who supplied the Hyge Tester. It contained more than 800 cubic feet of reinforced concrete and was in the snape of a cube 10 feet on each edge with a hole in the center 3 feet in diameter to accept the

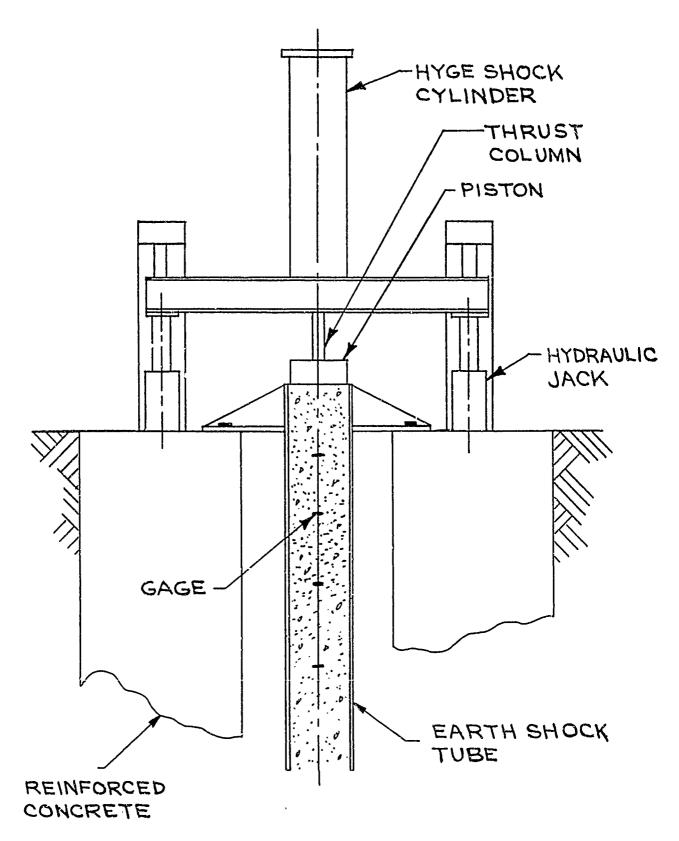


Figure 4. Hyge Shock System.

shock tube. Seven feet of the block was below the earth's surface. To accommodate gage cables from the instruments in the shock tube, a 4-inch pipe or conduit was imbedded in the concrete with one end extending above the top surface of the concrete block and the other end intersecting the 3-foot diameter hole at the lower surface.

### 3.3.5 Shock Tube

The 10-foot long shock tube, which has an inside diameter of 10 inches, was securely fastened to the concrete foundation by means of two 8-inch structural channels straddling the 3-foot diameter hole. Plumbing of the tube was facilitated by bolting the assembly to steel plates imbedded in the concrete. It was planned to follow the procedure previously used in the laboratory experiments to assemble the tube, sand, Teflon powder (when used) and instruments.

# 3.4 Test Series No. 1

With the assistance of BRL technical personnel, a preliminary test was carried out to check the performance of the earth shock tube system which had been constructed. The system performed generally as expected, and it appeared that only minor modifications and adjustments were required.

#### 3.4.1 Experimental Arrangement

The input force pulse generated by the Hygo cylinder was measured by a Baldwin-Lima-Hamilton Type C load cell. Soil stress was measured by modified BRL (oil-coupled) stress gages, and acceleration by Wiancko Type A 1023 accelerometers. Atlantic Research Corporation earth strain gages were installed, but gave no records because their range was exceeded during the first one or two trials before the recorders were operating.

The shock tube, with no liner, was filled with moist, clean building sand, the gages being

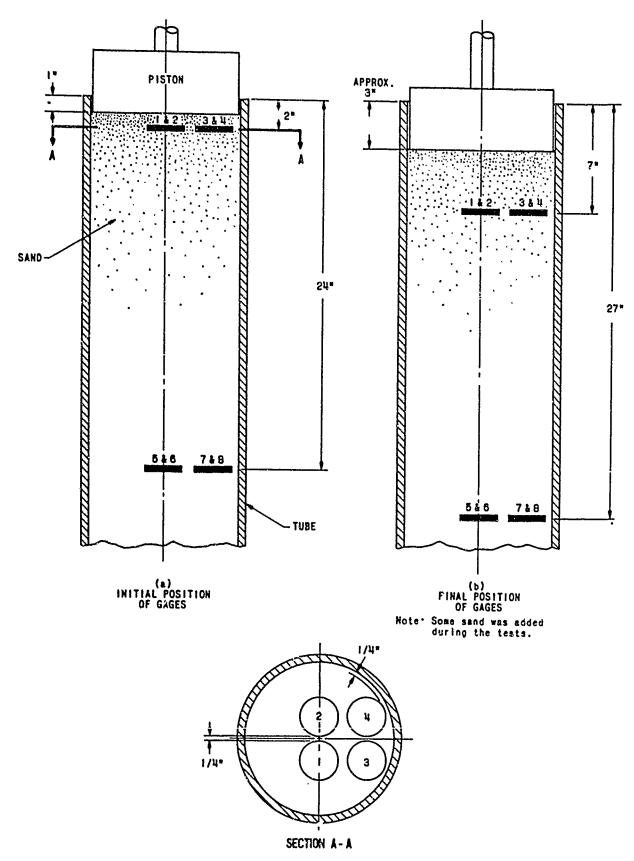
installed and carefully hand-tamped at the chosen levels during the filling. Two gages of each type, one at the center and one at the edge of the tube, were installed at different levels. No particular difficulty was experienced in locating and connecting the gages. The initial and final locations of the stress gages and of the accelerometers are shown in Figure 5. The strain gages were installed at different levels and are not shown in the figure.

## 3.4.2 Experimental Procedure

The Hyge cylinder was aligned with the earth shock tube by individual operation of the hydraulic jacks. Leveling of the semi-floating piston was accomplished by tapping on the appropriate side. After alignment was completed, the Hyge cylinder was lowered until the load cell button made contact with the bearing plate attached to the piston and resting on the soil surface. The amount of preload on the piston was measured by observing the deflection of the recorder galvanometer connected to the load cell. The Hyge cylinder was then charged and fired and the gage outputs recorded by the CEC recorder.

The sand column was compacted several inches by the first two or three applications of force, to a level beyond the range of adjustment of the hydraulic jacks; therefore, additional sand was placed in the tube to bring the surface to approximately one inch from the top of the tube. Sand was not added after the first two or three shots, but the piston was lowered to the displaced soil surface for the next force application.

The gages were not disturbed during the tests. Afterwards, the sand was carefully removed and the locations of the gages with reference to the top of the tube were measured.



GAGES:

STRESS 1,3,5,7 ACCELEROHETERS 2,4,6,8

Figure 5. Position of Gages - Test Series Number 1.

#### 3.4.3 Observations

## 3.4.3.1 Equipment Performance

This preliminary investigation indicated that this system for the rapid application of force to a column of earth performed satisfactorily. Minor problems that required solution were the following: adjustment and measurement of the preload on the piston; centering the piston in the shock tube; and removal of air trapped in the hydraulic cylinders. At high forces, a relatively small damped oscillation was evident in the applied force.

The firing pressure of the Hyge cylinder depended on the preload on the piston. If this load was too great, the piston would not fire; if it was zero, the load cell button impacted on the bearing plate. Control and measurement of the preload needed improvement. In later work, the load cell was kept in contact with the piston at all times by a spring device.

The piston was about 3/32 inch off center. Examination of the piston and the tube after the tests showed that there was no binding between the components during the tests. However, the condition was corrected before further tests.

As the Hyge cylinder fired, it was noted that the cross-member attached to the cylinder moved upwards. The maximum movement, on Shot No. 3, was about 1/2 inch. This action was attributed to air trapped in the hydraulic cylinders, a condition which was corrected before further tests were performed.

## 3.4.3.2 Gage Records

Several good records were obtained in these preliminary tests. As an illustration of the kind of data which can be expected, Figure 6 shows the records obtained from the load cell and from the two stress gages at the upper level in Shot No. 3 and Shot No. 5. The maximum stresses recorded in these two tests are tabulated below.

Observed Peak Stress

Shot No. 3 Shot No. 5
Applied Stress: 242 psi Applied Stress: 74 psi

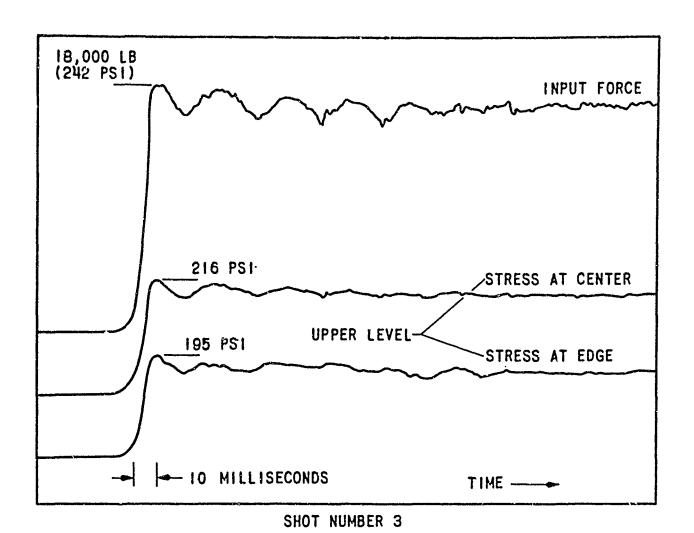
	Center (psi)		Center (psi)	Edge (psi)
Upper Level	216	195	71	59
Lower Level	24	25	1.9	1.6

Records were obtained from the accelerometers but not from the strain gages.

#### 3.4.3.3 Instrument Performance

The stress gages and the accelerometers seemed to have performed satisfactorily. Visual observation of the deflections of the galvanometers during the first one or two force applications indicated that the strain gages were operating initially but failed when their range was exceeded. It was later determined that the stop for the linear motion potentiometer, the transducing element of the strain gage, was damaged. As a result of this test, the design of the strain gage was modified.

The gages were excavated carefully after the tests. They appeared to have moved parallel to the axis of the tube, without turning. Within the error of observation, gages near the edge of the tube



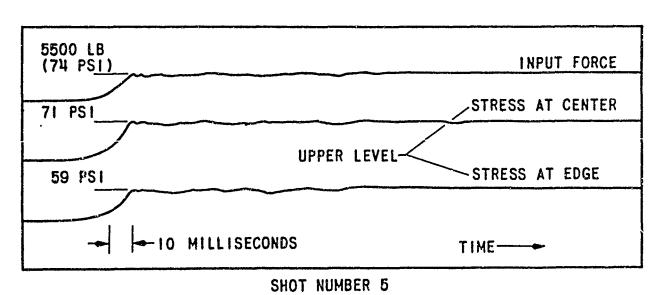


Figure 6. Stress Data Obtained from Earth Shock Tube.

moved the same distance as gages at the same initial level near the center.

# 3.5 Test Series No. 2: Simple Sand-Filled Tube

The purpose of Test Series No. 2 was to obtain data with a simple sand-filled tube to assist in the investigation of the proposed friction-reduction system using Teflon powder.

# 3.5.1 Experimental Arrangement

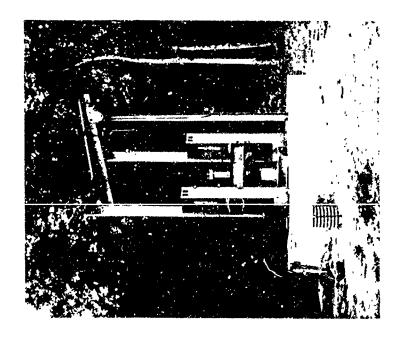
Centering the piston was not accomplished because the easily available adjustments were not sufficient; however, careful observation indicated that the piston was not enough off center to come in contact with the side of the tube during operation. Photographs of the installation are given in Figure 7.

Because of their smaller size, CEC Type 4-202 accelerometers were substituted for the Wiancko accelerometers used previously. Otherwise the instrumentation was the same: Baldwin-Lima-Hamilton load cell, modified BRL (oil-coupled) stress gages, and Atlantic Research Corporation strain gages. The gages were located at two levels. In the first two groups of tests, stress and acceleration gages were located at different levels to avoid crowding the gages. In the third group of tests, the stress and acceleration gages were located at the same levels. The original and final locations of the gages are shown in Figures 8 and 9.

# 3.5.2 Experimental Procedure

Test Series No. 2 was carried out in three groups. Before each group, sand was excavated to a depth of about 27 inches and the tube was filled with fresh sand, tamped by hand. During excavation, the distance of each gage from the top of the tube was measured. During the filling operation, the gages were installed at the selected levels.





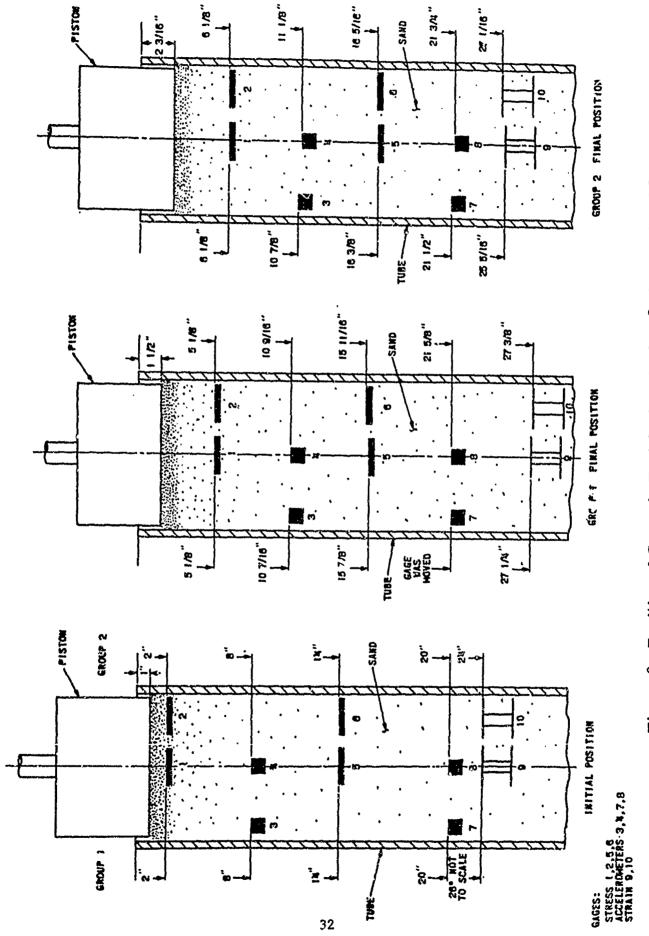


Figure 8. Position of Gages in Test Series Number 2, Groups 1 and 2.

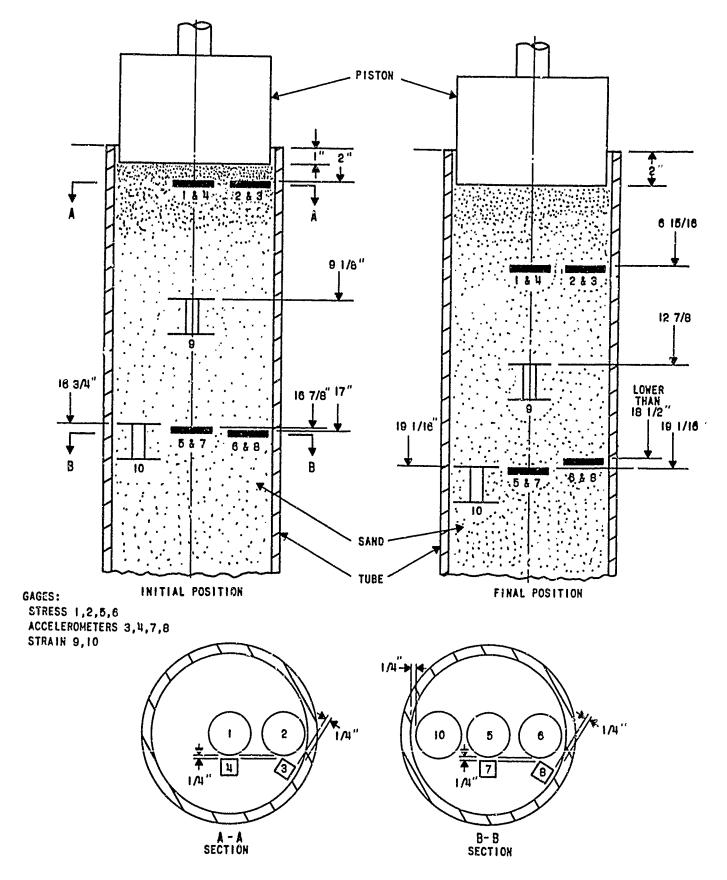


Figure 9. Position of Gages in Test Series Number 2, Group 3.

Each group of tests consisted of a number of successive force applications, and the gage outputs were recorded for each force application. After each of the first few force applications, fresh sand was added to bring the surface back to the original level; otherwise the surface would have been outside the displacement range of the hydraulic cylinders. Sand was not added after the sand column had been compacted to the extent that the piston motion was small, 1/8 inch or so.

Before each shot, the piston was adjusted to be about 1/16 inch above the sand surface. Occasion-ally the 0-ring at the orifice of the Hyge cylinder would not seal properly, with the result that force application would occur slowly instead of suddenly. This condition could be recognized by watching the load cell galvanometer. At a load of about 200 pounds, the pressure was then released, and the Hyge cylinder was readjusted. This event is termed a misfire and is believed to be caused by malfunction of a check valve.

The location of the sand surface with reference to the top of the tube was measured after each shot. The density and moisture content of the sand were not measured.

## 3.5.3 Observations

## 3.5.3.1 Stress

Table IX gives the measurements of applied stress and of stress in the sand column during the initial phase of the compaction process, when sand was added before the force application. Measurements are given for the two levels, both at the center and at the edge. During the initial compaction process, stresses observed at the upper level, both at the center and at the edge, were high relative to the applied

TABLE IX

Stress Measurements: Initial Compaction

Test Series No. 2

Note: Sand was added to tube before each of the following applications of force.

		Ann1	;o4(3)				Str	ess (p	si)		
Applied(3) Stress				Upper	Level	(4)		Lower L	evel (4	)	
Group (1	l) <sub>Shot</sub> (2	·) (p	si)	Ce	nter	Edge		Center		Edge	
No.	No.	Peak	100 ms	Peak	100 ms	Peak	100 ms	Peak	100 ms	Peak	100 ms
1	2 7	198 190	129 145	306 168	171 104	277 246	150 159	81 62	19 14	59 42	17 15
2	1 2 3 4	111 174 171 190	44 121 160 185	228 239 190 186	134 197 164 170	236 208 156 166	111 125 144 149	70 55 44 51	37 22 3 <sup>1</sup> + 43	64 43 32 42	42 26 27 37
3	1 2 3	132 202 204	76 179 191	248 230 212	144 186 185	250 207 198	128 162 167	71 41 33	36 33 24	52 18 18	32 18 14

- Notes: (1) Between each group of tests, sand was excavated and the tube was filled with fresh sand to a depth of about 27 inches from the top of the tube.
  - (2) Missing shot numbers represent a recording failure or a misfire.
  - (3) Force is measured. Stress is computed assuming uniform distribution of force over the area of the piston, 74.6 square inches.
  - (4) See Figures 2 and 3 for gage locations.

stress. With one or two exceptions, stresses measured at the edge were less than stresses measured at the center.

Table X lists the stress measurements during the final compaction process for the three groups of tests. Sand was not added during these tests. At the upper level, the center peak values were not greatly different from the applied peak stresses, and edge values were usually less than center values. Attenuation between the upper and lower levels is marked.

The oscillograph traces were similar to those shown in Figure 6.

#### 3.5.3.2 Acceleration

The only data taken from the acceleration records were the peak values. These are given in Table XI for the initial compaction phase and in Toble XII for the final compaction phase.

# 3.5.3.3 Sand Surface and Gage Displacement

The displacement of the sand surface caused by each force application is given in Table XIII. Note that sand was added during the early shots of each group.

The initial and final locations of the gages are given in Figures 8 and 9. It is interesting to compare the gage displacement with the sum of the surface displacements.

TABLE X
Stress Measurements: Final Compaction
Test Series No. 2

Note: No sand was added during these tests.

	ed (3)		Stress (psi)									
	••	~	ess	Upper Level (4)					Lower Level (4)			
Group(1) Shot(2) (psi)			si)	Ce	nter	E	dge	Center		Edge		
No.	No.	Peak	100 ms	Peak	100 ms	Peak	100 ms	Peak	100 ms	Peak	100 ms	
1	9 10 11 12	142 143 160 171	145 132 149 151	154 152 177 218	106 114 139 166	176 176 182 191	128 133 136 147	38 41 49 50	20 25 28 25	27 25 29 30	16 17 21 20	
2	5 6 7 8 9 10 12	198 218 238 286 338 303 352	196 213 238 264 250 30 283	217 214 242 270 344 294 344	185 195 220 220 238 262 256	1.64 1.69 1.69 2.58 329 2.62 311	138 151 168 211 203 232 221	54 59 66 75 91 78 99	41 52 55 52 30 62 51	45 49 59 63 79	38 45 51 50 34 -	
3	4 5 7 9	246 308 348 409	240 269 325 370	242 318 343 369	210 247 296 278	228 296 336 388	193 220 284 313	3°i 51 60 67	29 28 39 41	19 27 40 43	16 17 29 32	

Notes: See Table IX.

TABLE XI

Acceleration Measurements: Initial Compaction

Test Series No. 2

Note: Sand was added to tube before each of the following applications of force.

Group (1	) <sub>Shot</sub> (2) 	Appli Str (p Peak	ess si) 100 ms	Fea Upper Le Center	77: \	Lower	(g) evel <sup>(4)</sup> Edge
1	2 7	198 190	129 145	100 59	97 41	28	- 49
2	1	111	44	125	91	82	73
	2	174	121	39	22	19	11
	3	171	160	9	13	8	6
	4	190	185	13	17	12	8
3	1	132	76	87	113	116	92
	2	202	179	14	13	6	8
	3	204	191	13	11	10	11

Notes: See Table IX.

TABLE XII

Acceleration l'easurements: Final Compaction
Test Series No. 2

Note: No sand was added during these tests.

Group(1)	Shot(2)	Applie Stre (pa Peak	ed (3) ess si) 100 ms	Peal Upper Le	Oi Y	Lower Le	(g) evel <sup>(4)</sup> Edge
1	9 10 11 12	142 143 160 171	145 132 149 151	51 15 23 15	38 15 15 11	16 12 10	10 8 8
2	5 6 7 8 9 10 12	198 218 238 286 338 303 352	196 213 238 264 250 300 283	18 16 28 27 - 27 61	27 21 29 16 - 29 31	10 16 17 13 - 24 29	20 9 14 11 - 15
3	4 5 7 9	246 308 348 409	240 269 325 370	27 69 41 56	26 39 34 39	17 - 41	14 - 42 -

Notes: See Table IX.

TABLE X1II

Displacement of Sand Surface (inches)

Test Series No. 2

Shot No.	Group 1	Group 2	Group 3
1	2	3	3 1/4
2	9/16 <sup>(2)</sup>	3/8 <sup>(2)</sup>	1/4 <sup>(2)</sup>
3	1/4 <sup>(2)</sup>	1/8 <sup>(2)</sup>	1/16 (2)
4	(1,2)	1/8 <sup>(2)</sup>	3/16
<b>5</b> ,	(1,2)	1/8	1/8
6	(1)	1,'8	1/8
7	3/16	1/8	3/8
8	1/8 <sup>(2)</sup>	1/8	1/16
9	1/16	3/16	1/16
10	1/16	3/16	
11	1/4	1/8	
12	1/16	1/16	
13	1/16		

<sup>(1)</sup> Misfires; displacement too small to measure.

<sup>(2)</sup> Fresh sand added to tube before firing.

# Sum of Surface Stress Gage Displacement, Displacements (inches) Center, Upper Level (inches)

Group 1 3 4/8 3 1/8
Group 2 4 11/16 4 1/8
Group 3 5 4 15/16

The general trend of these displacements is about what one would expect.

#### 3.5.3.4 Strain

The strain gages were installed with the two discs fully extended. Unfortunately, however, the bridge could not be balanced at this value of the resistance.

# 3.6 Test Series No. 3: Friction-Reduction System

The purpose of Test Series No. 3 was to test proposed installation techniques and to obtain measurements to assist in the evaluation of the system.

#### 3.6.1 Experimental Arrangement

A four foot long polished stainless steel cylindrical liner was placed in the top portion of the tube. The liner fitted the tube snugly.

A separator was used to install a layer of Teflon powder. The separator is a one foot long, thin-walled cylinder with an outside diameter that is 1/2 inch less than the inside diameter of the liner, leaving an annular space 1/4 inch thick. Four small angles are placed along the outside cylindrical surface, parallel to the longitudinal axis, 90 degrees apart, to maintain the annular separation. All surfaces of the separator were polished with a size 400 grit carborundum paper.

The installation proceeded as follows: The tube was excavated to a depth of 4 feet. The liner was installed. The separator was lowered to the 4-foot depth. The annular gap was filled with Teflon powder No. 5, which was lightly compacted during the filling operation by tapping the tube

with a hammer. The inside of the separator was filled with sand, hand-tamped. The separator was then carefully raised to the next level and the operation repeated until the tube was full. Gages were installed at the selected levels during the filling.

The installation of the Teflon layer was carried out without difficulty. Examination of the layer after the first excavation showed the layer to be uniform and continuous. It was consolidated enough to stand by itself, but a piece broken off could be crembled easily in the hand.

The initial and final gage locations are shown in Figures 10 and 11. Accelerometers were omitted from the second Setup because of the high accelerations observed in the first setup.

# 3.6.2 Experimental Procedure

Test series No. 3 was carried out in four groups. After Group No. 1 was completed, the setup was left intact over night and force applications, Group No. 2, were continued the next day. The sand and Teflon powder were then excavated to a depth of about 31 inches and a new setup made for Group No. 3. Group No. 4 was a continuation of Group No. 3 after a layover of one night. This procedure was adopted to investigate whether moisture permeating the Teflon layer would cause an observable change.

The procedure for Test Series No. 3 was similar to that described for Test Series No. 2. Whenever column compaction was appreciable, fresh sand and Teflon powder were added.

## 3.6.3 Observations

## 3.6.3.1 Equipment Performance

The amplifier attenuations in the first shot were set as for Test Series No. 2, expecting satisfactory traces to be obtained

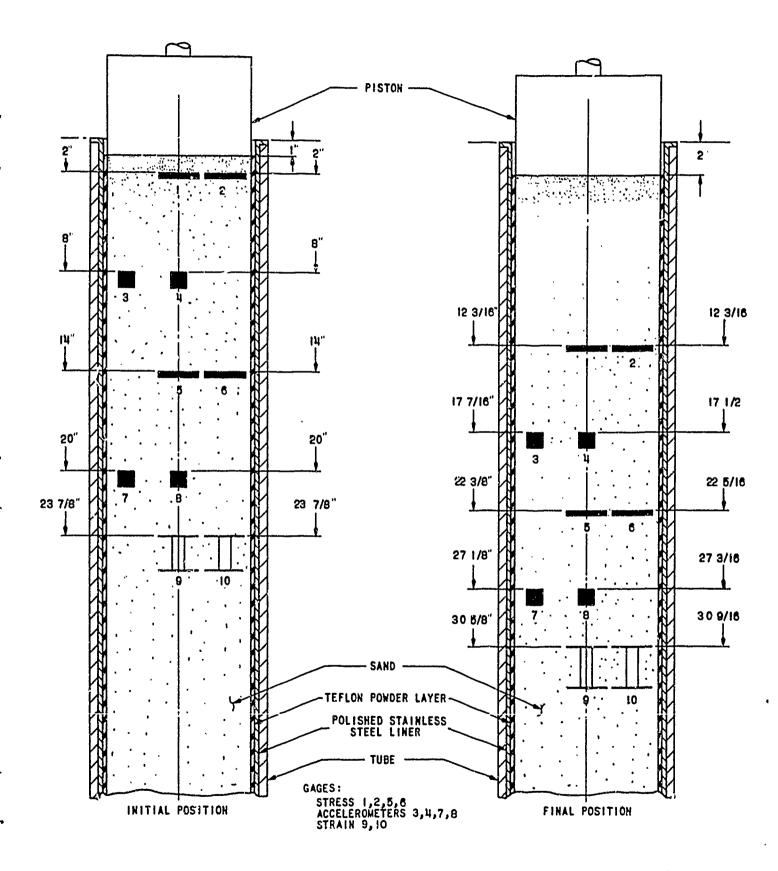


Figure 10. Position of Gages in Test Series Number 3, Groups 1 and 2.

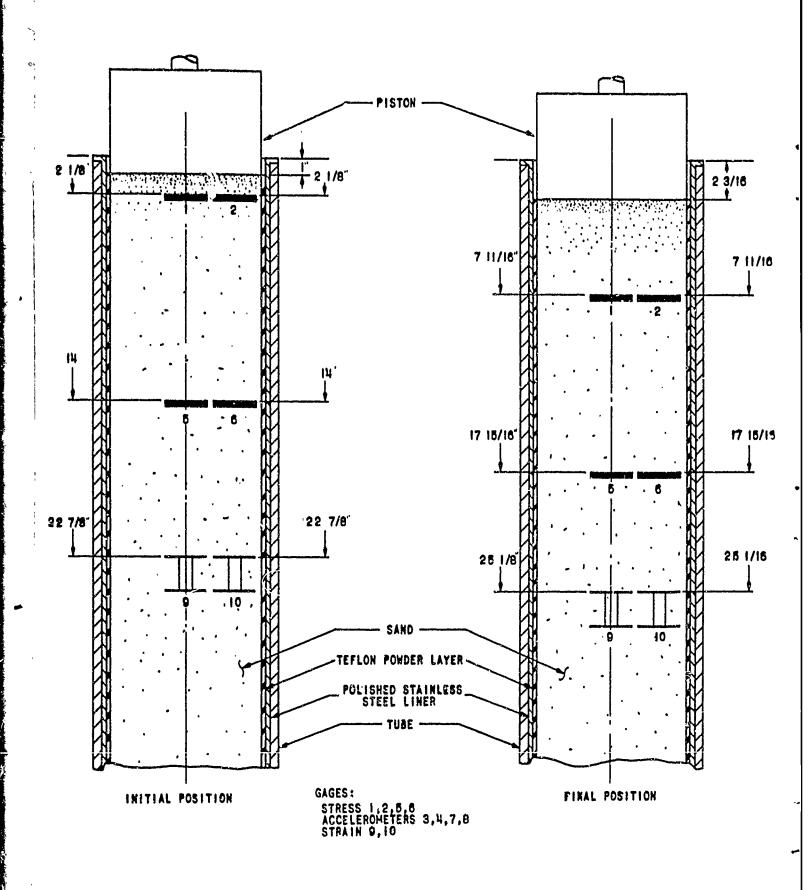


Figure 11. Position of Gages in Test Series Number 3, Groups 3 and 4.

even if transducer outputs were two or three times the previous value. Some of the traces went off the paper, and the attenuation had to be increased. An examination of two or three traces showed amplitudes beyond the range of the gage; so the possibility exists that Test Series No. 3 should be repeated, with different gages or gage locations, to get results which can be compared with the results of Test Series No. 2.

## 3.6.3.2 Sand Surface and Gage Displacement

The initial and final gage locations are shown in Figures 10 and 11. The displacements of the sand surface in successive shots of Test Series No. 3 are listed in Table XIV. It is to be noted that the surface and the gage displacements are much greater than in Test Series No. 2. end of Group No. 2 tests, the upper level gages had been displaced downward 10 3/16 inches, whereas the sum of the sand surface displacements in Group No. 1 and Group No. 2 was 11 11/16 inches. Similar figures for Group No. 3 and Group No. 4 are: qaqe displacement, 4 9/16 inches; sum of sand surface displacement, 6 13/16 inches. The numerical evidence that compaction occurred to a greater depth is corroborated by the fact that excavation, literally by the hand, was much more difficult between Group No. 2 and Group No. 3 of Test Series No. 3 than between the groups of Test Series No. 2.

Recovery phenomena were observed. With pressure maintained on the sand surface, the distance between the top of the tube and the sand surface was determined. The

TABLE XIV

Displacement of Sand Surface

Test Series No. 3

	Grou		Group	2(2)	Cro	up 3	Group	<sub>4</sub> (3)
Shot	Applied (1) Stress (psi)	Displ. (inches)	oplied (3) Stress (psi)	Displ. (inches)	pplied (1) Stress (psi)	Displ.	pplied (1) Stress (psi)	Displ. (inches)
1	136	3 5/8	-	5,'16 <sup>(14</sup> )(5)	_	9/16(4)	174	7/16 <sup>(5)</sup>
2	2.74	2 1/2 <sup>(5)</sup>	174	5/1€	-	1/16(4)	_	0 (4)
3	266	1 7/8 <sup>(5)</sup>	174	0	174	2 1/2	-	o (4)
1+	266	1/2	174	1/8	266	1 3/4 (5)	174	5/16
5	360	9/16	174	0	360	1/2	183	3/16
6	454	ì					183	1/4
7							183	1/16
8							183	1/8
9							183	1/16

## Notes:

- (1) Applied stress is an approximation estimated from Pyge cylinder operating pressure and Test Series No. 2 data.
- (2) Group 2 is a continuation of the Croup 1 shock tube setup which was left intact overnight.
- (3) (roup 4 is a continuation of the Group 3 shock tube setup which was left intact overnight. The sand and Teflon powder were excavated to a depth of about 31 inches after croup 2 and a complete new setup made for Group 3.
- (4) Misfire; small static load applied.
- (5) Fresh sand and Teflon powder added to tube before firing.

The measurement was repeated after the piston had been removed from the tube. The difference was usually small, but sometimes amounted to 1/2 inch or more. The measurements reported on Table XIV were made after recovery had taken place.

#### 3.6.3.3 Stress

Irregularities were observed in the records of Group No. 4, and it was evident that the ranges of the gages had been exceeded. Two accelerometers were obviously damaged. Laboratory tests of stress-gage performance were made after the field tests were completed. It was found that the stress gages performed satisfactorily in static calibrations but were very sensitive to acceleration. Furthermore, the type of irregularity observed in the records of Group No. 4 could be demonstrated in the laboratory by moving the gages. Hence it was decided to repeat Test Series No. 3 with gages of greater range.

Irregularities were not observed in the records from the first three groups of this test series, and it may be that the readings were valid. Consequently, the observations are reported here in Tables XV-XVIII, but no conclusions will be made until Test Series No. 3 is repeated.

Table XV gives the stress measurements, both the peak and the value at 100 milliseconds, made during the initial compaction of the sand column, when sand and Teflon powder were added to the tube before each force application. Stress measurements made during the final compaction of the sand column, when no sand was added to

#### TABLE XV

Stress Measurements: Initial Compaction

Test Series No. 3

Friction-Reduction System

Note: Sand and Teflen powder were added to tube before each of the following

applications of force.

See text for discussion of gage performance.

		Appli	ed (3)				Str	ess (psi)			
Stress			Upper Level (4)					Lower I	evel (4	)	
Group (1	) <sub>Shot</sub> (2)	<b>(</b> p	<b>si</b> )	Ce	Center Edge		dge	Ce	nter	Edge	
No.	No.	Peak	100 ms	Peak	100 ms	Peak	100 ms	Peak	100 ms	Peak	100 ms
1	1	58 138	20 62	172	70 234	109 285	41 191	58	19	60	20
	2	138	62	-	234	285	191	191	113	162	113
3	3	130 247	77 182	375 615	225 412	358 386	204	181	96	182	99 218
	4	247	182	615	412	386	259	305	217	258	218

- Notes: (1) Group 2 is a continuation of the Group 1 shock tube setup which was left intact overnight. The sand and Teflon powder were excavated to a depth of about 31 inches after Group 2 and a complete new setup made for Group 3.
  - (2) Missing shot numbers represent a recording failure or a misfire.
  - (3) Force is measured. Stress is computed assuming uniform distribution of force over the area of the piston, 73.6 square inches.
  - (4) See Figures 10 and 11 for gage locations.

# TABLE XVI

Stress Measurements: Final Compaction

Test Series No. 3

Friction-Reduction System

Note: No sand or Teflon powder was added during these tests.

See text for discussion of gage performance.

		Ann1d	.ed(3)			Stress (psi)						
40.5	•	Str	ess		Upper	Level (4	.)		Lower Level (4)			
Group(1)	Shot(2		si)	Center		Edge		Center		Edge		
No.	No.	Peak	100 ms	Peak	100 ms	Peak	100 ms	Peak	100 ms	Peak	100 ms	
1	4 5 6	261 302 364	188 250 313	585 556 720	384 392 486	565 1070	328 389 846	298 334 412	215 242 277	256 268 289	217 242 267	
2	2 3 4 5	181 204 217 171	120 134 152 123	344 384 399 333	215 254 282 235	321 298 302 284	229 244 268 224	181 226 171	145 162 166 136	192 222 167	154 171 175 145	
3	5	348	284	776	585	787	585	375	281	281	268	

Notes: See Table XV.

# TABLE XVII

Acceleration Measurements: Initial Compaction

Test Series No. 3

Friction-Reduction System

Note: Sand and Teflon powder were added to tube before each of the following applications of force.

See text for discussion of gage performance.

	A	pplied(3) Feak			eration (	
Group (1)	Shot(2)	Stress	Upper Le	vel <sup>(4)</sup>	Lower Le	vel <sup>(4)</sup>
No.	No.	(psi)	Center	Edge	Center	Edge
1	1 2	58 138	108 <i>6</i> 6	77 44	23 83	57 83
3	3 4	130 247	No acce used	leromet in this		

Note: See Table XV.

# TABLE XVIII

Acceleration Measurements: Final Compaction

Test Series No. 3

Friction-Reduction System

Note: No sand or Teflon powder was added during

these tests.

See text for discussion of gage performance.

	A	pplieď(3)	Pea	k Acceler	ation (	g)	
Group(1) No.		Peak	Upper Le	vel <sup>(4)</sup> <u>Edge</u>	Lower L Center	eveî (4)	
1	4 5 6	261 302 364	- 135	200	-	52 127 130	
2	2 3 4 5	181 204 217 171	- 59 52	- 46 42	- - -	- - -	
3	5	348	No acce	leromet <del>e</del> :	rs used	in this	group.

Note: See Table XV.

the tube, are given in Table XVI. Acceieration measurements are given in Tables XVII and XVIII. Strain measurements are not reported because the amplifier was overloaded in most cases.

# 3.7 Earth Shock Tube - Test Series No. 4

Test Series No. 4 was conducted with the assistance of ERL technical personnel.

## 3.7.1 Experimental Arrangement

The installation and experimental procedure for Test Series No. 3, described previously, was followed in Test Series No. 4. The bottom of the earth shock tube was filled with the building sand used in the previous tests to within 32 inches of the top. Standard Ottawa sand was used to fill the tube to the top. In this test series measurements were made in the Ottawa sand.

Because of the high accelerations noted in the previous tests and the unavailability of higher range, CEC Type 4-202, aluminum-housing accelerometers, accelerations were measured with three Wiancko, Type A1023, accelerometers and one CEC, Type 4-202, stainless-steel-housing accelerometer. A Wiancko accelerometer with a range of 500g was placed at the upper level at the center of the tube and the 250g CEC accelerometer was placed at the edge at the same level for Group Nos. 1 and 2. In Group No. 3, the CEC accelerometer, designated by No. 4 in Figure 12, was placed at the second level at the center of the tube. The other Wiancko accelerometers, with a range of 100g, were used at the lower levels.

The BRL stress gages were reconstructed using CEC, Type 4-313 pressure elements with ranges of 1000 psia and 500 psia. The 1000 psia elements were used at the upper level and the 500 psia elements at the lower level. One of the 1000 psia

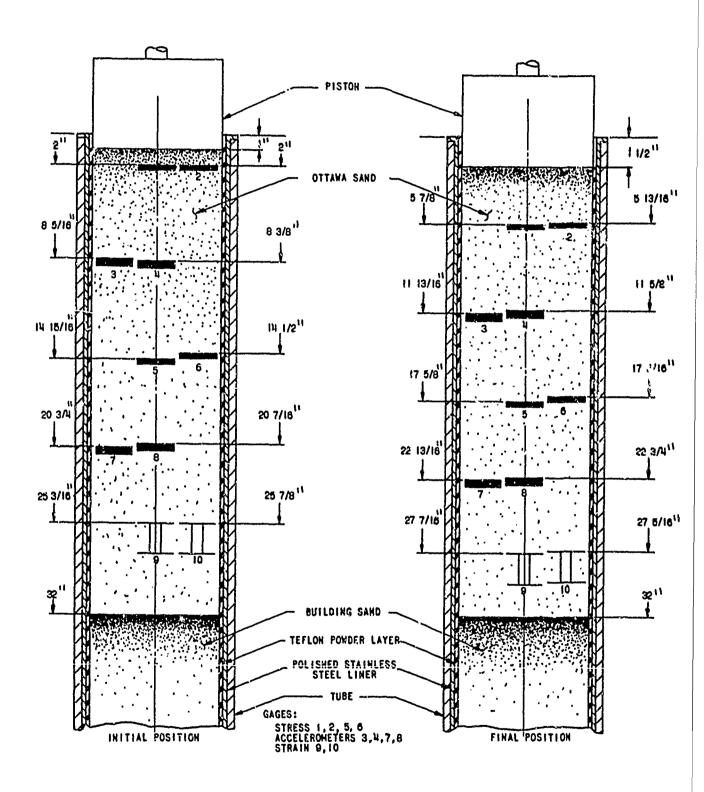


Figure 12. Position of Gages in Test Series Number 4, Group 1.

elements designated as No. 2 in Figure 12, was damaged on the first shot of Group No. 1. The manufacturer informs us that these pressure elements cannot withstand more than 100g's impact loading. Therefore, upon the recommendation of CEC, several Type 4-327 pressure elements, which are capable of withstanding 1000g's for 1 millisecond, are being adapted to the BRL stress gage.

The Atlantic Research Corporation earth strain gage designated as No. 9 in Figures 12, 13, and 14 used the modified bridge circuit to prevent amplifier overloading. The sensitive element of this strain gage was a wirewound linear potentiometer with a resolution of 0.0014 inch. The other strain gage used an infinite-resolution carbon film resistance element as the transducer.

The initial and final locations of the gages for the three groups of tests are given in Figures 12, 13, and 14. In Group No. 1 and Group No. 2, similar gages were placed at the same level, one at the center and one at the edge. Stress Gage No. 2 was damaged, and the bridge could not be balanced for Accelerometer No. 8; so no data are reported for these two gages. In Group No. 3, different types of gages were placed at the same level, as described in Figure 14.

#### 3.7.2 Observations

The experimental arrangements for Test Series No. 4 were different from those for previous tests in the following respects: Ottawa sand was used; the sand was tamped by hand, using a two-inch diameter rod, as the tube was filled and the gages installed; a different tetering pin was used for the Hyge Shock Tester, to give faster rise times. Because of these changes, quantitative comparisons of the results of Test Series No. 4 with the results of previous tests may not always be justified.

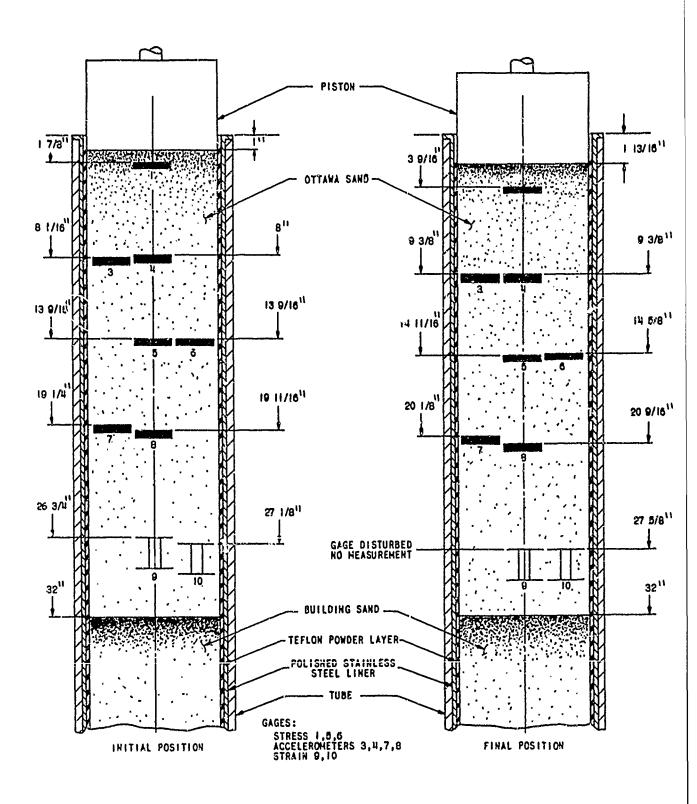


Figure 13. Position of Gages in Test Series Number 4, Group 2.

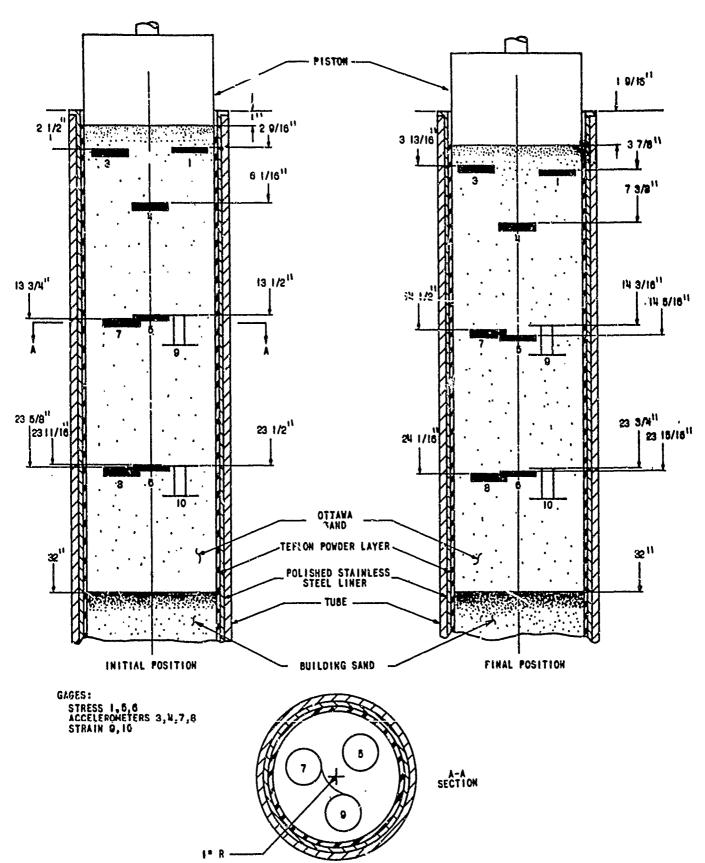


Figure 14. Position of Gages in Test Series Number 4, Group 3.

# 3.7.2.1 Wave Form

Figure 15 is a tracing of the records given by the load cell, two accelerometers, two stress gages, and a strain gage from Shot No. 13 of Group No. 2. Rise times are much faster than in previous tests, two or three milliseconds for pressure and less than one millisecond for acceleration.

An interesting feature of the pressure records is the pulse which occurs on the otherwise almost flat top of the pressure The time of arrival of the pulse at the various gages is consistent with the view that a reflection occurs at the bottom of the liner, four feet down. The sand from the four-foot depth to the ten-foot depth had not been disturbed during the four series of tests, but the whole ten-foot length of the tube was excavated after Series No. 4 in order to relocate the cables. A noticeably compacted layer in the building sand, able to support its own weight over the span of ten inches, was found at the four-foot depth at the bottom of the liner and probably served to reflect the pulse.

During the first few shots of a group, while the sand is being compacted, the wave forms are different from those observed from shots toward the end of the group. Figure 16 is a tracing of records from the first shot of Group No. 2. It will be noted that the rise times are long, ten milliseconds or so for the pressure records and about two milliseconds or so for the acceleration records. A rough indication of the status of compaction seems to be given by the displacement of the sand surface, Table XIX. Speaking generally,

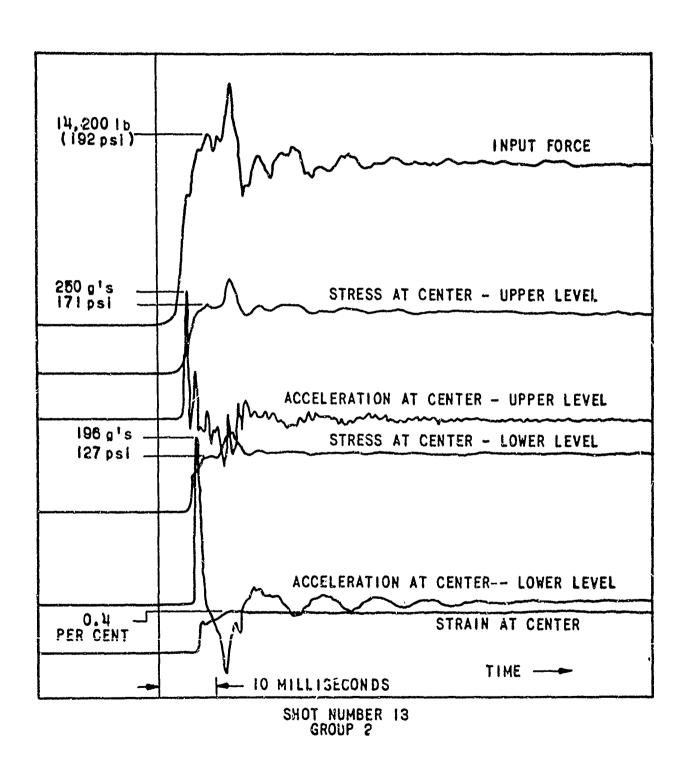
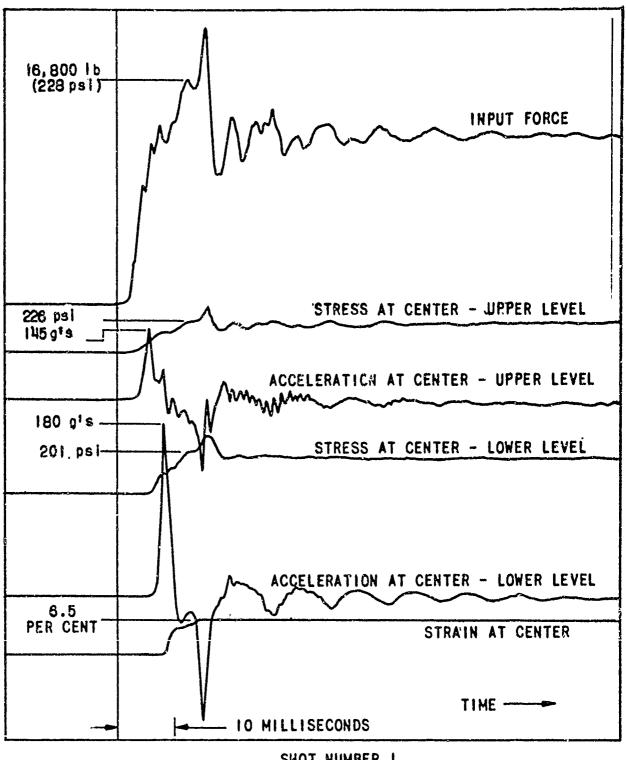


Figure 15. Typical Records from Test Series Number 4 After Compaction.



SHOT NUMBER 1 GROUP 2

Figure 16. Typical Records from Test Series Number 4 During Initial Compaction.

TABLE XIX

Displacement of Sand Surface

Test Series No. 4

	Displac	Displacement (Inches)				
Shot No.	Group 1	Group 2	Croup 3			
1	1 1/4	5/8	7/8			
5	3/8 (1)	1/8	1/8			
3	7/16	1/8	3/8 (1)			
4	1/8	1/8	1/16			
5	7/16 (2)	1/16	1/8			
6	1/4 (1,2)	0				
7		1/8				
8		1/4 (1)				
9		3/16				
10		1/8				
11		1/16	•			
12		o				
13		1/16				
14		1/16 (2)				

### Notes:

- (1) Sand was added before this shot.
- (2) Full tank pressure was applied after the Hyge cylinder fired, and thus a gradually increasing load was applied to the sand column for about eighty seconds. In usual operation, the Hyge cylinder is exhausted right after firing, so that the load on the sand column is reduced in a few seconds.

short rise times are observed when the displacement of the sand surface or successive force applications becomes and remains small.

#### 3.7.2.2 Acceleration

Peak accelerations are recorded in Tables XX and XXI. The accelerations are high, much higher than they were in Test Series No. 2 (building sand, no Teflon powder) or in Test Series No. 3 (building sand, Teflon powder lubricating layer).

In each of the three groups, the acceleration at the lower level is greater than
the acceleration at the upper level for the
first shots in the group and is less for
the last shots of the group. The reason for
this apparent anomaly is not evident, although
it might be related to differences in the
initial state of compaction at the two levels.

The data of Table XX allow a comparison of upper level accelerations at the edge of the tube with the accelerations at the center. Of the 18 pairs of observations, 8 show equal or smaller acceleration at the edge and 10 show greater acceleration at the edge. In Group No. 2, for which all the applied stresses were about 200 psi, the average peak acceleration at the edge was 191g and the average peak acceleration at the center was 181q. It would appear from these average data and from an examination of the individual data that the acceleration of the sand at the the side of the tube is not significantly different from the acceleration of the sand at the center.

TABLE XX

Acceleration Measurements

Test Series No. 4

Peak Acceleration (g) Peak Applied Upper Level Lower Level Group Shot Stress No. No. Center (psi) Edge Center Edge (1) No records obtained ---(1) \_115 .123 6 No records obtained (1) 14 

### Notes:

(1) Sand was added before this shot.

TABLE XXI
Acceleration and Stress Measurements

Test Series No. 4

Group No. 3

	Peak Applied	Peak Accel. (g) (1)		Stres	Stress (psi) (1,2)		
No.	Stress (psi)	Gage 3	Gage 4	Gage 7	Cage	Gage 5	Cage 6
1	291	159	216	241	458	304	274
2	202	134	135	173	264	164	136
3 (3)	20 <sup>1</sup> t	144	140	168	319	141	130
4	207	99	107	1.60	248	136	120
5	204	98	91	84	228	130	120

### Notes:

- (1) See Figure 14 for gage location; the levels for stress gages and accelerometers were not the same.
- (2) See text for discussion of these measurements.
- (3) Sand was added before this shot.

In the final four shots of Group No. 2, the amplitude of the acceleration pulse is reduced an average of 21% in traveling from the upper level to the lower level, a distance of 11 3/16 inches.

### 3.7.2.3 Stress

Stress measurements are reported in Tables XXI and XXII. In most cases the value reported was that of the plateau before the peak, because the peak was taken to be a reflection. For a few of the early shots in a group, the peak was reported, because the rise was continuous and smooth without any indication of a plateau.

The stresses reported at the upper level for Group No. 1, Group No. 3, and two shots of Group No. 2 were greater than the applied stress. Similar inconsistencies occurred in the values reported for the lower level. The explanation for these anomalies is not vet apparent, although it is believed that they cannot be attributed to faulty calibration of the gages or the recording system.

Certain dvnamic effects on the gages can be expected. Inertial effects of a moving column of sand would increase the recorded pressure, and this might be particularly important during the compaction process. The transducer is sensitive to acceleration. According to the manufacturer, the acceleration sensitivity is 0.05% of full range per g. At accelerations of 100g, this effect might be significant, as much as 50 psi for the gages at the upper level and 25 psi for the gages at the lower level. Correction has not been attempted for acceleration sensitivity, because the accelerometers were not at the

same level as the stress gages (except for Group No. 3). Errors of dynamic origin would be a maximum during the first one or two milliseconds of a record and may not be important for the stress data given in the table (except, possibly, during the early stages of compaction).

The static recordings of the stress gages are probably also in error, or, better said, are probably not representative of the undisturbed medium. The stress values at 100 milliseconds given for Test Series No. 3 are inconsistent with the values reported for the applied stress. Similar anomalies were found in the one or two static measurements (not reported) made in connection with Test Series No. 4. Clearly, the characteristics of the stress gages should be investigated further.

Although the numerical values of the stress are questioned, relative values may give valid information. The data of Table XXII allow a comparison of stress at the edge with stress at the center, both at the lower level. Of the 18 pairs of observations, 8 show the stress at the edge to be equal to or less than the stress at the center and 10 show the stress at the edge to be greater than the stress at the center. In Group No. 2, for which all applied stresses were about 200 psi, the average of the stress readings at the edge is 158 psi and the average stress at the center is 155 psi. We conclude that stress measured at the edge is not significantly different from stress measured at the center.

TABLE XXII

Stress Measurements

Test Series No. 4

Stress (psi) (1) Peak Applied Upper Level Lower Level Shot Group Stress No. No. (psi) Center Center Edge (2) 4 No records obtained (2) 1 2 34 56 78 9 10 No records obtained (2) 14 

#### Notes:

- (1) See text for discussion of these measurements.
- (2) Sand was added before this shot.

In the final four shots of Group No. 2, the amplitude of the recorded stress step is reduced an average of 20% in traveling from the upper level to the lower level, a distance of 11 1/8 inches. The stress attenuation is thus approximately equal to the peak acceleration for these four shots. Additional evidence regarding attenuation is given by the fact that the amplitude of the reflected stress pulse, when it returns to the upper level after a travel of 7 1/2 feet, is 30% to 40% of the amplitude of the incident stress step (also measured at the upper level).

#### 3.7.2.4 Strain

A sequence of satisfactory strain records was obtained, for the first time, in Group No. 2 of Test Series No. 4. The peak strains are given in Table XXIII.

The quantity actually recorded is the change in distance between the two discs, which is converted to strain by dividing by the actual distance between the discs. The distance between the discs was not measured before each shot and cannot be computed from the observed change in distance because of the recovery which takes place when the force applied to the sand is removed. In Table XXIII the base length is assumed to be the maximum distance between the discs, and the actual strains may be up to 1/5 greater than those reported.

The resistive transducer in the center location was wire-wound with an advertised resolution of less 1.4 mils. The transducer at the edge location was a carbon film with a resolution claimed to depend only on the amplifying and recording equipment. The

TABLE XXIII

Approximate Strain Measurements (1)

Test Series No. 4 - Croup No. 2

	Feak Strain	(percent)
Shot No.	Center	Edge
1	6.5	3.2
2	2.8	1.8
3	2.5	1.5
ļt	0.8	1.4
5	0.6	1.4
6	** **	··· ·
7	0.5	1.3
8	0.6	1.4
9	0.5	1.2
10	0.5	1.2
11	0.6	1.4
12	0.5	1.0
13	0.4	1.0
14	0.5	1.0

### Notes:

(1) The maximum distance between the discs of the Atlantic Research strain gage was used as the base length for computing the strain. Therefore, excluding Shot No. 1, these data can be in error by a maximum of 20% of the reported strain.

smallest observed change in distance was ten mils; so resolution errors can be no greater than 14%.

The form of the strain wave front is often similar to the form of the stress wave front; most of the strain occurs in the first millisecond, and a gradual increase up to the final value occurred during the next several milliseconds. Table XXIII records the final value. According to these data, strain at the edge is less than strain at the center during initial compaction but becomes greater as compaction proceeds. Additional measurements in future work will be necessary to check this first set of strain observations.

### 3.7.2.5 Conclusions

Measurements of acceleration made in the earth shock tube with the Teflon powder lubricating layer are much higher than accelerations measured in the earth shock tube, and the attenuation of stress and acceleration is reduced. Measurements of stress and acceleration at a given level are approximately the same whether made at the edge or at the center.

Readings of the stress gages appear to be anomalous and further investigation is required. A series of strain records has been obtained, but additional work is necessary to test the gage and to develop the technique of using it.

## 3.8 Additional Proposed Test Series

Since the Teflon powder friccion-reduction system seemed to have achieved a measure of success, plans were made to extend the system to the full 10-foot length of the shock tube. In addition, provisions were made to measure the frictional force at the sidewalls of the confining tube

during each application of force (Figure 17).

If these measurements had been attempted, they may have established what fraction of the applied energy was lost as friction.

Another series of earth shock tube tests was conducted. In this test series, the Teflon powder friction-reducing system was placed in the entire length of the 10 foot shock tube.

The data seemed to indicate that the friction-reducing technique did not perform as expected. However, further investigation revealed that the Teflon powder layer between the sand column and the confining tube was not continuous. There were several large areas in the tube, especially at approximately 55 inches from the top of the tube, where no Teflon powder was present and the sand came into direct contact with the confining tube. The total contact area was considered to be of sufficient size and so oriented that the friction-reducing technique could not be evaluated. Therefore, the data obtained from this series of tests will not be reported and the series should be repeated.

A new technique, using electric vibrators, was proposed to introduce the Teflon powder between the sand and the confining tube. A series of tests utilizing this procedure was never carried out.

# 3.9 Additional Instrumentation Testing

In tests with the earth shock tube anomalously high readings were reported from stress gages imbedded in the soil under both static and dynamic conditions. This over-registration of earth pressure cells is found in the static and quasidynamic measurements of soil mechanics (1) and has recently been reported (2) in dynamic soil measurements similar to those

<sup>(1)</sup> See, for example, the following summary paper: J. J. Hamilton; Earth Pressure Cells, Design, Calibration and Performance; National Research Council (Canada), Division of Building Research, Technical Paper No. 109; Ottawa, November 1960.

<sup>(2)</sup> T. Winston and J. R. Stagner; Free-Field Stress Gauge and Test Results in a New 1000 psi Dynamic Pressure Tank; Noise Control, Shock and Vibration, 7, 4-10 (November-December 1961).

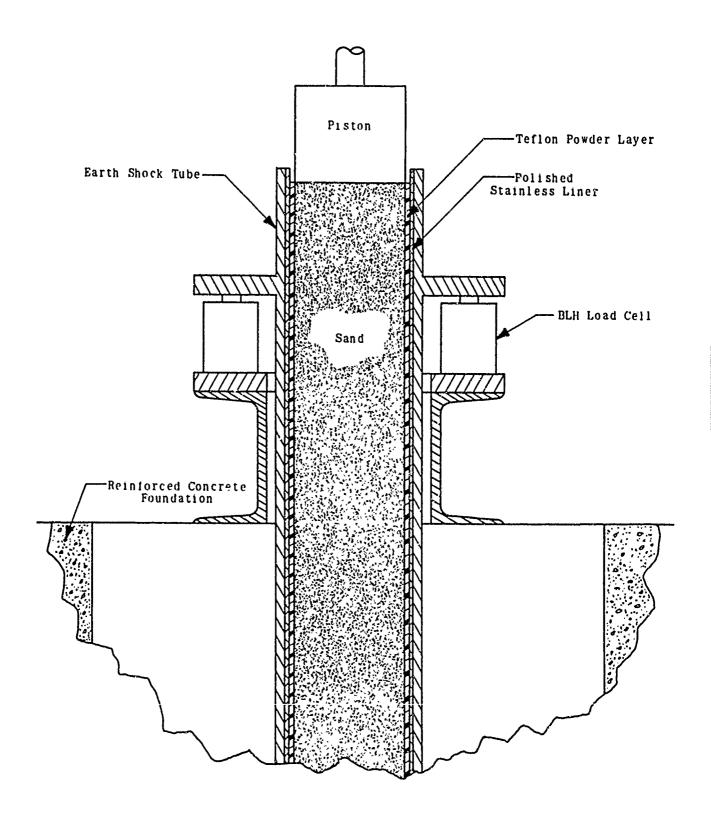


Figure 17. System to Measure Friction at Sidewall of Earth Shock Tube.

made in this project.

A series of simple preliminary static tests was conducted at the Ballistic Research Laboratories, observed by Atlantic Research Corporation personnel, to investigate stress gage behavior in sand. Several stress gages similar to those used in our dynamic tests were imbedded at several levels in a container of ordinary sand. The container was approximately 12 inches in diameter and 2 feet high. The top gage was placed approximately 1/2 inch below the top surface of the sand. A load was applied to the confined sand by means of a piston loaded with a manually-operated hydraulic system. A load cell placed between the piston and the hydraulic actuator measured the applied load. Some of the general observations made during these tests are reported here, by permission of the BRL Technical Supervisor.

It was found that the Imbedded stress gages could be made to register any value, within the limits of the system, by simply adjusting the sand density about the active face of the stress gage. When a small mound of sand was placed above the stress gage, as shown (exaggerated in Figure 18(A), the application of force to the sand column produced high stress indications by stress gages 1 and 3. If a small depression was made over the stress gage, as shown (exaggerated) in Figure 18(B), the stress gages would indicate low stresses, and in some cases very nearly zero stress, for maximum input stress. After each test, the piston was carefully removed from the container and the top sand surface observed. The surface was flat, and there was no indication that an irregularity had been present on the surface.

It is obvious from these tests that stress gages imbedded in sand can indicate values which are not in agreement with the average input stress. Therefore, an investigation of stress gage behavior in soils was undertaken to determine if reliable stress measurements can be made with imbedded gages.

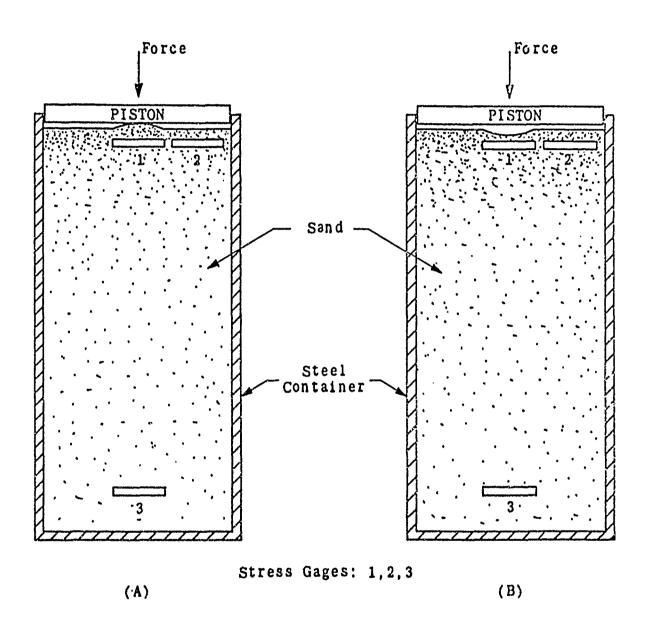


Figure 18. Experimental Arrangement for Static Tests to Observe Stress Gage Behavior in Sand.

#### 4.0 DYNAMIC OEDOMETER

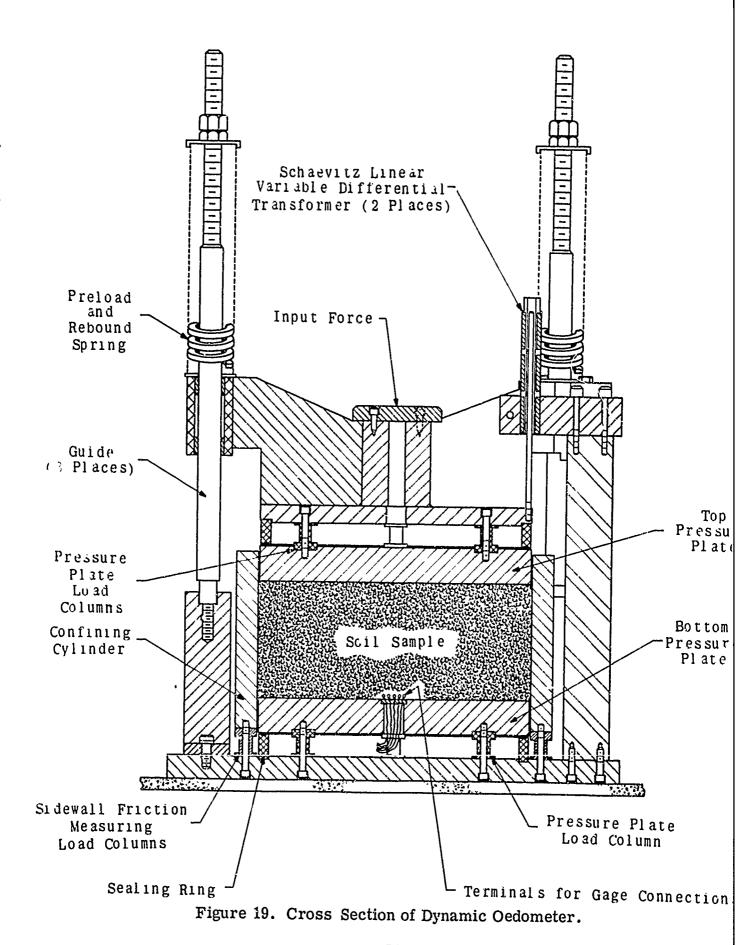
It was evident from earlier shock tube tests that the imbedded stress gages measured anomalously high values of stress under both static and dynamic conditions. This over-registration of earth pressure cells is not uncommon in soil measurement programs; however, the phenomenon is n.t quantitatively under-Therefore, it was decided to extend the program to investigate soil compaction and imbedded stress gage behavior before further shock tube tests were carried out. To carry out this program, a laboratory device was designed by Atlantic Research Corporation under the direction of Dr. Werner Heierli and Dr. Alva Matthews of the Paul Weidlinger Consulting Engineering Firm. device, named the dynamic oedometer by Dr. Heierli, was designed to be used in experiments on constrained soil samples subjected to one-dimensional loading. In addition to the stress gage investigation program, it is hoped that experiments may be conducted to determine accurate values of Poisson's ratio, constrained modulus and other soil constants useful for propagation predictions.

The dynamic oedometer is a laboratory device designed for the investigation of soil compaction and embedded stress gage behavior. It was expected that the device would be useful in one-dimensional wave propagation studies. A cross section of the device is shown in Figure 19 and a photograph of the assembled unit is shown in Figure 20. This device is designed for a 1,000 psi maximum input pressure pulse.

### 4.1 Design

## 4.1.1 General

The dynamic oedometer is very similar to the device used for static tests. However, the rate of load application in this device can be controlled from a very slow rate, essentially a static load, to a very rapid rate, depending upon the loading device. It was planned to use the pneumatic device, the "Hyge" unit as a force applicator. This unit was described briefly in paragraphs 3.2 and 3.3 of this report.



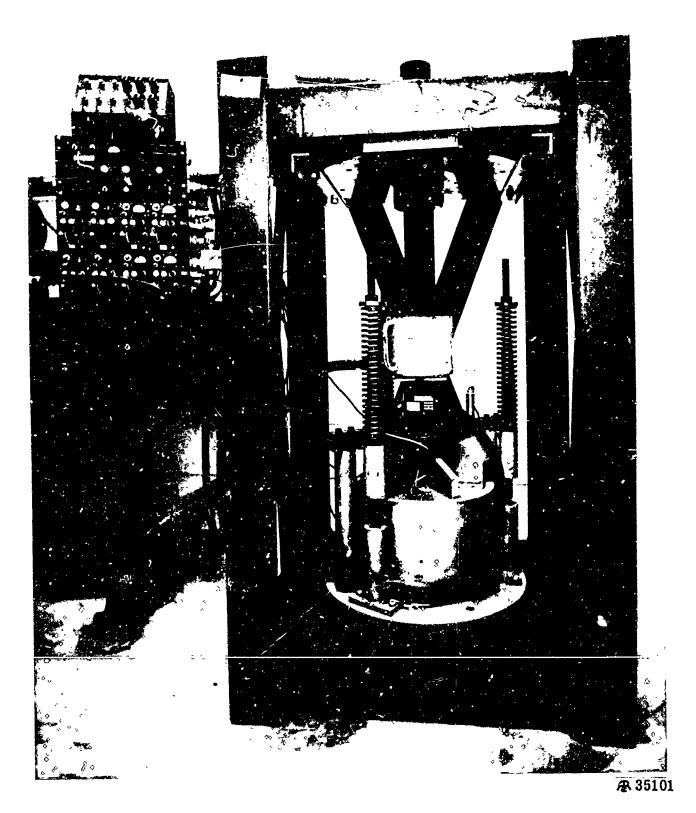


Figure 20. Dynamic Oedometer.

The Hyge unit is capable of producing peak forces of 40,000 pounds with a rise of time of a few milliseconds.

The oedometer consists of three main members the confining cylinder and a top and bottom pressure plate. The soil sample is packed in the very rigid smooth confining cylinder which allows essentially no lateral deformation. The sample is further confined by the top and bottom pressure plates as shown in Figure 19. A pressure pulse is introduced into the sample through the guided top pressure plate. The displacement of this plate is measured by two diametrically opposed Schaevitz LVDT gages. pulse is measured both by the load column arrangement in the top plate and a Baldwin Lima Hamilton load cell which is located between the source and the top plate assembly. As the pressure rulse passes through the sample, it is measured by the load columns located in the bottom pressure plate assembly which is identical to the top plate arrangement.

The movement of the top pressure plate is guided by three equally spaced guide rods. The arms extending from the plate to the guide rods contain oilite bushings which minimize the frictional force on these moving members. To either preload the soil sample or to control to some degree the rebound of the top pressure plate, a coil spring is used at each guide rod.

It is expected that with a soil sample of a height-diameter ratio of 1/4, the side friction between the soil and the confining cylinder can be reduced to a negligible amount. However, to establish the exact side friction-time history during a test, the confining cylinder is supported by four equally spaced load columns which are instrumented with strain gages.

Extremely small circumferential strains in the cylinder which are assumed not to affect the soi? measurements will be measured by strain gages placed

in two diametrically opposed longitudinal grooves in the cylinder wall.

An important requirement of the dynamic oedometer is that the state of stress in the soil sample during a test should be quasistatic; i.e., that at any one time during the dynamic loading of the sample, the stresses within that sample must be the same for different locations. One of the initial tests planned for the device is to determine whether or not such a quasistatic condition exists.

#### 4.1.2 Pressure Plates

The average pressure at the top and bottom of the soil sample is measured by identical arrangements of a pressure plate supported by four equally spaced load columns. The plates which form part of the pressure plate assembly are designed to have sufficiently small deflections and a high natural frequency. Calculations show that the plate supported by the four load columns will have a maximum deflection in the order of 1/2000 of the diameter. The natural frequency of the pressure plate and the supporting load columns is in the order of 3200 cps.

Also, to minimize the effect of inertial forces due to the acceleration of the top pressure plate, it was constructed of an aluminum alloy. Assuming a 1000 psi peak pressure input into the sample with a rise time of 4 milliseconds and a parabolic deflection of the soil, calculations show the inertial loss due to the mass of the plate is in the order of 25 psi. As can be seen in Figure 19, the mass above the supports does not influence this inertial pressure loss.

To facilitate electrical connections to imbedded stress gages, a terminal arrangement consisting of a Teflon plug and wire pins is inserted into the bottom pressure plate. To eliminate any interference between the movement of soil particles and the Teflon plug, it is mounted flush with the pressure plate surface.

To eliminate additional frictional forces due to soil particles jamming between the pressure plates and the confining cylinder, the clearance between these two members is in the order of several thousandths of an inch which is smaller than the grain size of the proposed samples.

# 4.1.3 Load Columns

The average pressure at the top and bottom of the soil sample and the side frictional load on the confining cylinder are measured by load columns. load columns are instrumented with standard resistive foil elements which are connected in a Wheatstone bridge arrangement. Due to the maximum input pressure of 1000 psi and the low sensitivity of foil strain gages, two sets of load columns for the top and bottom pressure plates were considered. A small diameter cylindrical spool type load column is used for the lower stress levels while a solid cylindrical column is used for the higher stress levels. Piezoresistive strain elements were considered to instrument the columns because of their high gage factor; however, it was decided that their high temperature coefficients would produce many drift problems.

Load columns which were used were made of 15-7 Mo wrought stainless steel. They were heat treated to a hardness above Rockwell 40C and a tensile strength of 210,000 to 240,000 psi.

Special Constantan foil strain elements (Dentronics No. MH234TT-C6) possessing low hysteresis and high linearity characteristics were attached to each load column with RP-43 cement and cured for one hour at 350°F. Electrical connections were made to the foil elements and a gage coat of RP-43 was applied. The assembly was then post cured at 350°F for 24 hours.

Each instrumented load column was cycled with compressive loads twelve times with a Tinius Olsen testing machine. The peak force was 15,000 pounds. Three of the fourteen columns exhibited open circuit gages after testing and had to be reconditioned. A check on the linearity and hysteresis of the gage elements after testing indicated that they were still within the manufacturer's specifications.

The foil elements were then potted in a RTV silicone rubber compound to minimize the temperature drift problems.

The load columns supporting the confining cylinder which are used to measure the side friction are identical in design to the columns used in the pressure plates.

To eliminate the introduction of foreign matter into the top and bottom pressure load column cavity, a sealing ring is inserted between the pressure and the back-up plate. This ring is also used to offer some shielding to the unshielded strain gage leads.

## 4.1.4 Cylinder

The steel cylinder used to confine the soil sample has the following dimensions: 12-inch inside diameter, 1-inch wall thickness, and 7 3/4-inch length. It is sufficiently rigid to give a practically constrained test. Calculations, based upon the postulate that the lateral strain in the soil sample in the oedometer should be no more than 1/500 of the lateral strain of the soil in a confined test, indicate that the circumferential strain will be in the order of 0.01 X 10<sup>-3</sup> in/in for each 100 psi increment of axial stress. This amount of strain is considered to be satisfactory for the contemplated tests.

Since an attempt was to be made to determine the dynamic Poisson's ratio of the soil sample, it was desirable to incorporate into the design of the cylinder a method of measuring this small circumferential

strain. A strain gage attached to the cylinder wall would produce too small an output to be useful; therefore, to produce stress concentrations in the cylinder to produce sufficient gage output, two sets of diametrically opposed grooves are incorporated into the cylinder design. The grooves are placed longitudinally on axis, both from within and without the cylinder in a symmetrical arrangement so as to preserve the neutral axis. Therefore, using narrow grooves and assuming that the material behaves elastically, there will be no deformation from the circular shape of the cylinder. To produce a smooth surface on the inside of the cylinder, the grooves are filled with Araldite and machined to the inside diameter of the cylinder. The araldite has a low modulus of elasticity compared to that of steel. Strain gages are cemented into both outside grooves in the loaded area of the cylinder. Since the cylinder is loaded on only part of its length, careful consideration had to be given to detailed calibration of the strain gages placed in the grooves before accurate Poisson's ratio measurements can begin. A calibration technique similar to the actual loading of the cylinder must be used.

The natural frequency of the cylinder for cylindrically symmetrical vibrations (breathing mode) (neglecting the grooves) is 4600 cps.

#### 4.1.5 Pressure Input

Initially a pneumatic loading device, the "Hyge" unit used on the earth shock tube was selected to be used as the force applicator for the dynamic oedometer. The Hyge unit allows peak forces up to 40,000 pounds with a rise time of about 4 milliseconds to be applied to the oedometer.

### 4.1.6 Friction Problems

If side friction measurements indicated excessive values, it would become necessary to introduce a friction reducing system (such as silicone grease,

Teflon powder, etc.) between the soil sample and the confining tube.

Also, tests of friction between the soil sample and the pressure plates should be made. Friction on these end plates would greatly influence the lateral pressure measurements. If tests indicate that excessive friction is present on these plates, friction reducing systems will be incorporated into this area.

# 4.2 Testing

The dynamic oedometer was statically calibrated to a maximum force of 50,000 pounds.

No further work on the oedometer was performed under this contract.

The experiments run in the dynamic oedometer were funded under Contract No. DA 18-001-AMC-877X. A final report on this program has been incorporated as Part II of this report.

#### 5.0 HORIZONTAL DISPLACEMENT METER

### 5.1 Development

This instrument was originally conceived by personnel of the Ballistics Research Laboratories at the Aberdeen Proving Ground(1). A laboratory model was constructed at Aberdeen Proving Ground to demonstrate the principle. Atlantic Research Corporation was selected as contractor under Contract Number DA-36-034-ORD-3116-RD, a then existing contract, to design, fabricate and laboratory-test an instrument embodying those principles demonstrated that would be suitable for field application and routine manufacture. Atlantic Research initiated design work on this project in June of 1962. Design goals for the instrument housing dictated that it be watertight and have the structural strength to withstand the pressure and shock environment expected in association with 6-inch displacements in rock Intimately related to the housing design is the and soils. mechanism for precisely leveling the instrument in the longitudinal direction (propagation path of expected displacement wave) after it has been firmly imbedded at the test station. The housing design employed two concentric cylinders. outer cylinder, shown in Figure 21, provides protection from environmental damage and the inner cylinder; Figure 22, provides a structural base for the working parts of the seismometer. The inner cylinder is mounted to the outer cylinder at three points, Figure 22. Two of these points are designed to transmit about 99 percent of the acceleration loading due to the expected shock. These two points are located on the central horizontal diameter of the cylinder and use ball bearings to support the inner cylinder assembly. point which is designed to transmit about 1 percent of the inertial loading is a steel tongue attached to the inner cylinder with an electrical heating element wrapped around its periphery that is electrically and thermally insulated from the tongue, but is mechanically bonded to it. projects into a cup attached to the outer cylinder. This cup

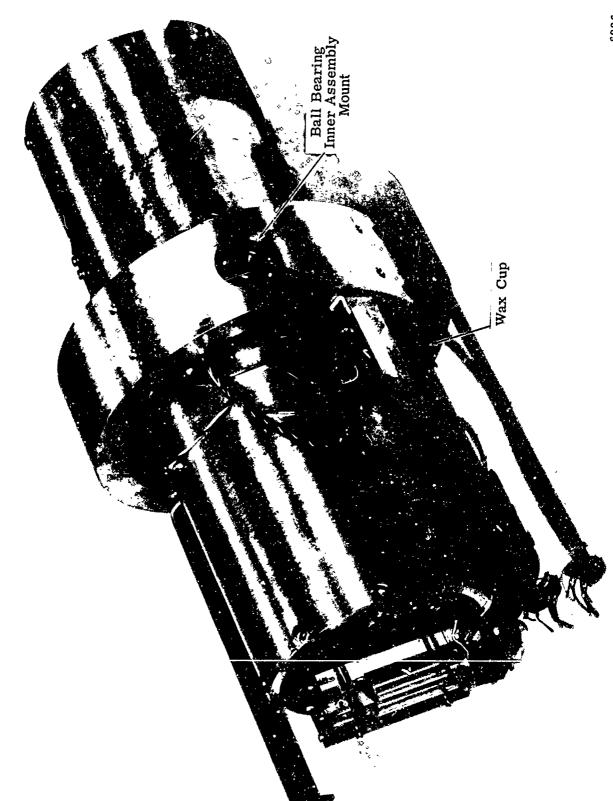
<sup>(1)</sup> Patent 3, 164, 983, B. Perkins, Jr., et. al., Horizontal Displacement Meter.

Control Box



Figure 21. Horizontal Seismometer Assembly.

Figure 22. Inner Cylinder of Horizontal Seismometer.



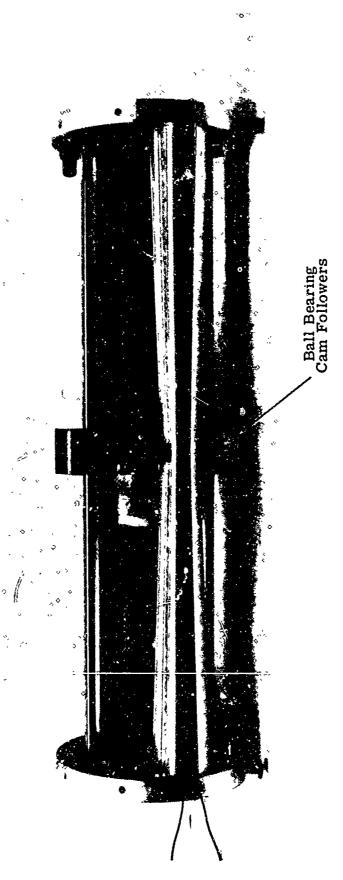
is filled with a wax which, when at ambient temperatures, is quite firm and easily capable of transmitting the designed 1 percent of the total inertial loading on the inner cylinder. There is sufficient imabalance provided in the inner cylinder assembly that, when the wax in the pot is molten, the inner cylinder will level itself to within 4 minutes of angle.

Since the instrument is essentially unaffected by transverse position, no provision is made for self leveling in this direction. There is, however, incorporated a position indicating switch which indicates remotely via warning light: on the control box, Figure 21, when the deviation from horizontal exceeds ± 30 minutes of angle in the longitudinal direction and ± 2 degrees of angle in the transverse direction. The warning light arrangement also indicates which direction from horizontal the instrument is tilted if the limits are exceeded. The warning light system was designed to be used only during installation of the instrument at the test station and, therefore, the control box with indicator lamps, was situated at the end of a 25-foot length of waterproof armored cable connected to the seismometer. Obviously, this is of no value during the actual test. Located at the control box also, is the switch to activate the heater, a shorting plug receptacle to release the mass, and cable connectors for battery power and for data output.

The seismic mass must be locked in its central position in order for the instrument to level itself. Provision is made for remotely releasing the mass (at the control box) after the wax has been melted, the instrument leveled, and the wax frozen again.

The seismometer proper, shown in Figure 23, is comprised of an inertial mass supported by ball bearings on two parallel steel rods which extend to the full length of the inner cylinder assembly (about 13 inches). Attached to the seismic mass are two ball bearing cam followers which are spring loaded in such a manner that they are pulled toward one another. These cam followers track on the opposite sides of a biconical cam extending the full length of the inner

Figure 23. Horizontal Seismometer Inertial Mass Instrumentation.



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Cylinder. This spring and cam arrangement provides the restoring force to return the mass to the center position after it has been displaced by the shock phenomena being investigated. This device provides a small spring deflection for a large mass deflection (a ratio of 12). The cam follower arms have a dashpot mounted between them which is filled with 400 cs silicone oil. The dashpot assembly is adjusted to provide 69 percent of critical damping to the seismic mass.

Also attached to the seismic mass are four electrical wiper contacts which traverse a deposited carbon resistance element and a gold plated copper strip. Both the resistance element and the metal strip extend approximately the length of the inner cylinder assembly. Electrical connections are made to each end of the carbon strip and to the metal strip. This arrangement forms a potentiometer which can be used as a voltage divider or as one or two arms of a bridge. The carbon resistance element was manufactured by the United Electrodynamics Corporation for this instrument and has a total resistance of about 1700 ohms. It was the sponsor's expressed desire that the seismometer be compatible with the Consolidated Electro-dynamics Corporation System "D" carrier-bridge amplifier system for field operations. In order to match this instrument to the System " ,", it was necessary to build a matching box, shown schematically in Figure 24. This device also adds the capability of remotely "zero" balancing the electrical output of the instrument and of generating calibration steps equivalent to plus or minus 2-inch displacements for calibration purposes.

The original plans for field testing this instrument called for its installation in a horizontal bore 10 to 15 feet in length, drilled in the side of an underground tunnel. To meet this requirement, an insertion tool was designed and fabricated with which one can place the instrument in bores up to 20 feet deep and remotely detach it from the case.

Laboratory tests of the instrument under simulated shock environments were marginally successful. The limitation imposed by the laboratory setup was that the required energy to accelerate the 40-pound mass of the instrument to rates

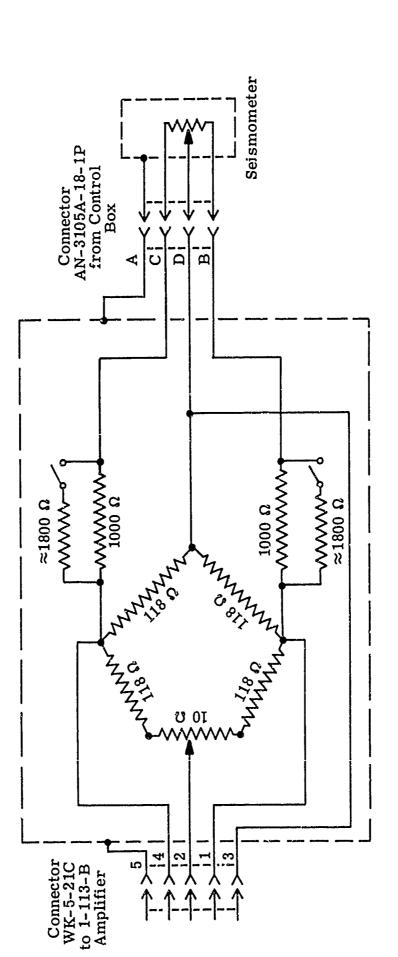


Figure 24. Matching Box for Six-Inch Horizontal Displacement Meter to CEC System "D" Amplifier Type 1-113-B.

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much higher than those associated with its natura. period was not readily available.

Laboratory tests on the improvised shake table indicated that the instrument would faithfully transduce horizontal displacements up to 6 inches if the period of the impressed motion was one second or less.

Tests using an approximately sinusoidal displacement input at a frequency slightly above the natural frequency of the instrument shown in Figure 25, indicate good fidelity of response.

Results of tests of natural period and damping ratio are shown in Figure 26.

# 5.2 Field Test

On July 1, 1964, the instrument was taken to the Suffield Experimental Station, Alberta, Canada, by Mr. Willis Jackson of BRL to record the ground motions caused by the detonation of 500 tons of H.E. The location assigned for this instrument was 200 feet from the center of the charge. This would place it just outside of the anticipated crater. See Figure 27.

Unfortunately, no other supporting instrumentation was placed at this location. The instrument was buried 5 feet deep at the designated station, leveled, checked out and The entire channel was then turned over to the calibrated. field crew to check periodically until the time of the shot and to make it the final / cording of the blast. Subsequent checks of the calibration prior to the shot indicated decreasing sansitivity for reasons unknown. Troubleshooting techniques were suggested by phone, but the results of these tests are not known. As a result, the calibration is meaningless, and tests since recovery of the instrument indicate that the resistance of the carbon trip may have been affected by the passage of excessive current during diagnostic tests. absolute magnitude of displacements therefore could not be determined in this test.

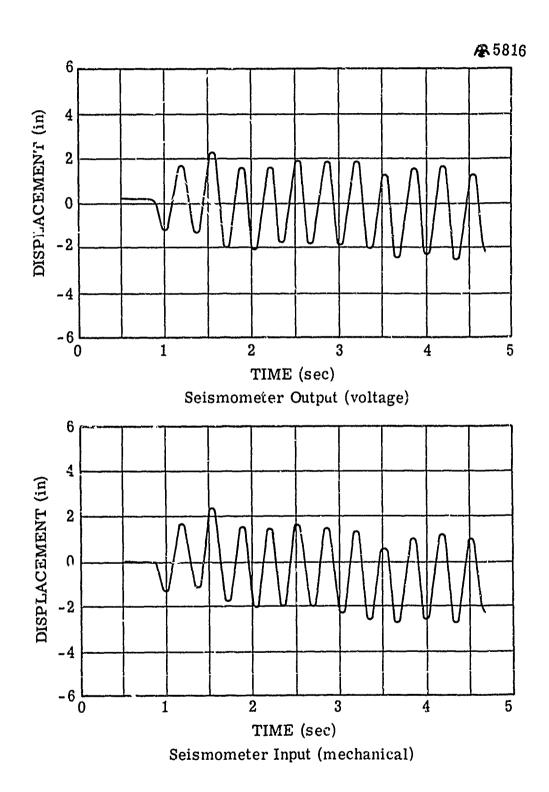
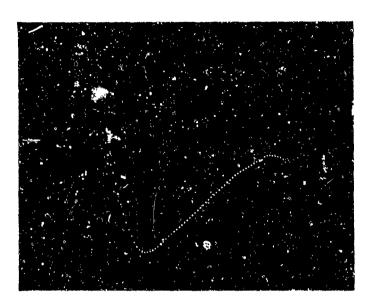
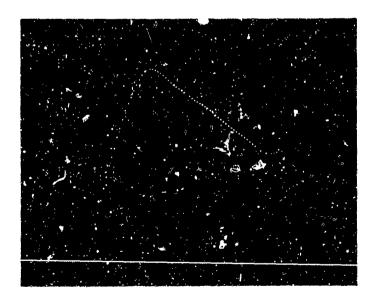


Figure 25. Fidelity Response of Horizontal Seismometer.

Timing Marks  $\cong$  100 msec (X Axis) Y Axis 4 cm  $\cong$  6 Inch Displacement Ratio of Successive Amplitudes  $\cong$  20:1 Per Cent Critical Damping  $\cong$  69



P = 1.5 Sec.



P = 1.5 Sec.

Figure 26. Natural Period and Damping Ratio.

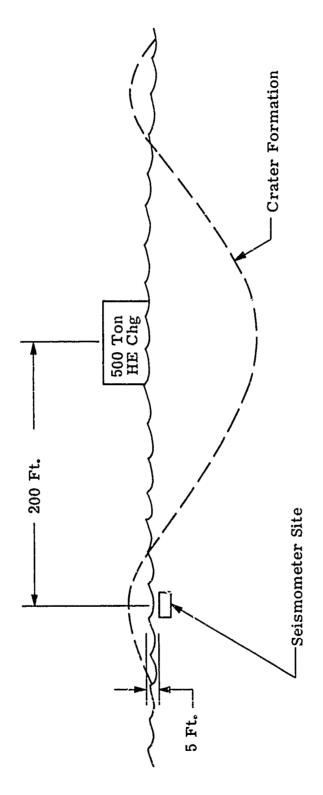


Figure 27. General Orientation of Test Station at Suffield Experimental Station.

## 5.3 Evaluation

Analysis of this record indicates that the airblastinduced displacement arrived at the instrument at T = 0.024second after the shot. The noise following the airblast wave is probably due to electrical causes. The surface layer and the instrument were probably set in vibration by the airblast. At 0.157 second, the initial displacement due to the arrival of the compressional "P" wave is shown. At 0.298 second, the arrival of a flexural wave is indicated. It appears that the combined effect of the "P" wave and the flexural wave were sufficient to exceed the limit of the instrument and the seismic mass came in contact with the arresting device after being displaced +6 inches. If one extrapolates the curve, it would appear that the total displacement was in the order of eight inches. At approximately 0.440 second, a negative displacement begins which is probably accompanied by some tilt as the instrument went off scale on the recorder at approximately 0.698 second and remained off scale until about 3.65 seconds, at which time it shows a positive displacement of about one inch. This event observed at 3+ seconds is believed to be a separate event due possibly to reflection from an impedance discontinuity located at the 5 to 10 thousand foot depth or possibly due to fallback of soil in the crater lip formation.

The velocity of the several waves is uncertain, since there is only one observation point and since the origin is broad compared to the distance to the observation point.

When the instrument was recovered and returned for post test inspection, damage apparently due to several causes, was observed. The suspension wire for the tilt indicating pendulum was broken. This is believed to be the only damage due to the blast. Other damage to the cable and connectors is believed to have been done in recovering the instrument from the test site.

Energy from the detonation can be transmitted to the instrument through several paths:

a. An airblast wave will propagate over the surface,

generating a seismic wave which will cause ground displacement at the instrument test station:

- b. Compressional "P" waves and transverse "S" waves will be transmitted through each of the near surface layers (see Figure 28 for approximate geologic formation);
- c. A Rayleigh wave will be transmitted through the top two or three layers and;
- d. A flexural wave will very probably be transmitted through the surface layer.

In addition, as the crater is formed, the surface layer will be displaced away from the explosion as the soil is pushed out of the crater to form the lip.

These events will probably overlap on a time scale close to the explosion. Farther out, since the velocity varies for the different types of waves, there will be a greater time separation for the different events.

The horizontal seismometer is designed to be insensitive to transverse displacements, but should record the horizontal component of the various waves. The instrument will also respond to tilt of its longitudinal axis. It is therefore difficult in the present instrument to separate those phenomena due to tilt and those due to displacement except by assuming that response to tilt alone will be at the natural frequency of the instrument. Displacements faster than the natural response time of the instrument must therefore be forced displacements, but may be accompanied by some tilt.

A synopsis of the recorded data taken from this instrument during the Suffield Experimental Station shot in July 1964 is shown in Figure 29.

### 5.4 Recommendations

Recommendations for further development of this instrument are:

- a. A new transducing element should be used which is less subject to degradation by currents used in ordinary test equipment and which is not so sensitive to contamination.
  - b. A level indicating device should be incorporated which

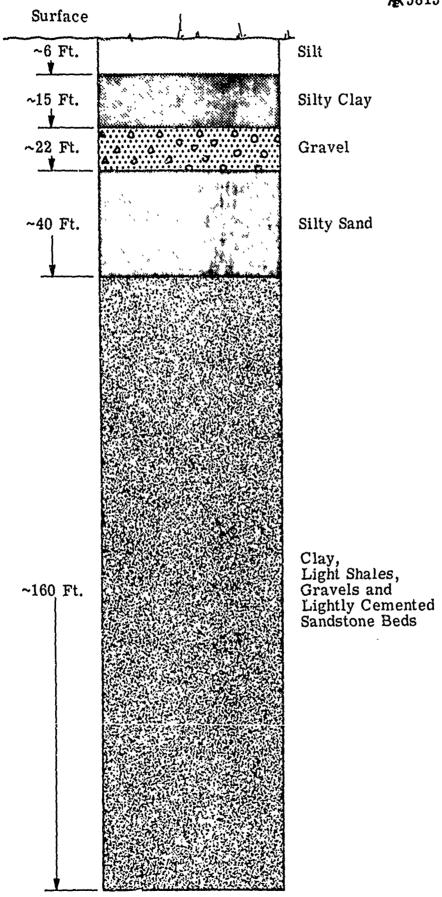
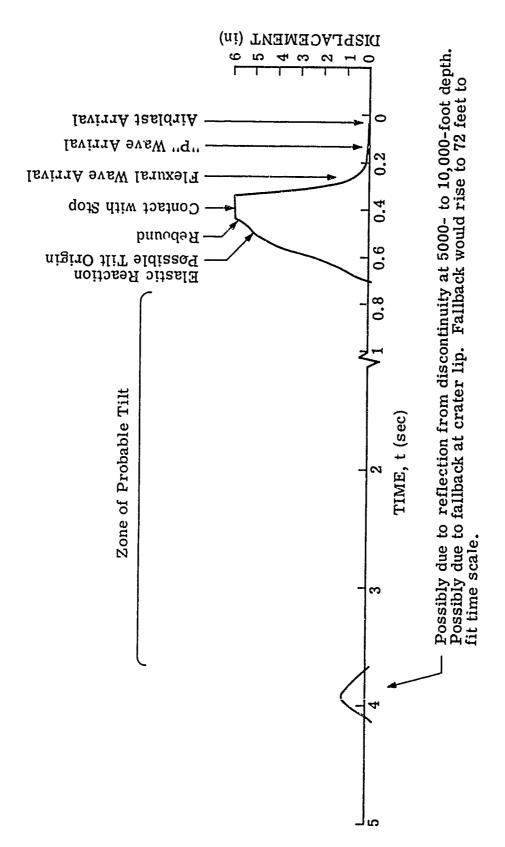


Figure 28. Approximate Thickness of the Various Beds to be Expected Between the Instrument Location and Impact Zone.



Figure 29. Synopsis of Recorded Data.



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will provide for continuous recording of the angle of the instrument both in the longitudinal and transverse directions while the displacement measurements are being taken.

- c. For measurements in soil a more flexible cable can be used, thereby reducing the chances of the cable influencing the performance of the instrument.
- d. A transducing element which does not contribute friction to the seismic mass will make possible a longer time constant for the instrument.

Future field tests of this instrument should have supporting instrumentation at the same station in order to fully
evaluate its performance. Some possibilities in supporting
instrumentation would be both horizontal and vertical velocity
meters, photographic coverage of the test site with a Fastex
camera and vertical posts to show gross shifts in the soil.

A contained explosion will provide a simpler phenomenon to observe since both airblast and crater formation will be avoided but displacement will be present.

# 6.0 DYNAMIC TRIAXIAL APPARATUS

In addition to the dynamic oedometer, the Ballistic Research Laboratories felt that another laboratory device for the measurement of Poisson's ratio should be designed and fabricated. Therefore a dynamic triaxial apparatus was designed by Atlantic Research Corporation under the direction of Dr. Werner Heierli and Dr. Alva Matthews of the Paul Weidlinger Consulting Engineering firm. A cross section of the laboratory device is shown in Figure 30.

The unit is somewhat similar to the conventional triaxial apparatus. However, it is designed for a maximum confining pressure of 1000 psi and has provisions for a rapid moving loading piston.

The unique feature of the proposed design was to determine Poisson's ratio by measuring the overall expansion of the soil specimen (1). This could be accomplished only if the specimen deformed uniformly over the entire length. In conventional triaxial testing, nonuniform deformation can be attributed to the complex stress condition existing at both ends of the specimen. It was felt that this condition existed because of the friction imposed on the soil specimen by the confining end plates; hence, if this friction could be reduced to a negligible amount, it seemed possible that uniform deformation of the specimen could be obtained.

Two techniques were going to be tried in an attempt to reduce the end plate friction to a negligible amount. The first technique involved the use of the Teflon powder system that was tested in the earth shock tube. The surfaces of both end plates coming in contact with the specimen would be highly polished and a thin layer of Teflon powder would be placed between the polished surface and the specimen. The second technique would use the method developed by Dr. Rowe, University of Manchester, England. A thin flexible membrane is placed over the ends of the specimen coming into contact with the end plates. Between the membrane and the end plates a thin lubricating film of silicone oil is placed thus reducing the friction in this area to a minimum. In addition, the membrane is so flexible that it is easily stretched radially by

<sup>(1)</sup> Heierli, Weirner and Matthews, Alva T., Informal Report to Aberdeen Proving Ground, Paul Weidlinger, Consulting Engineer, Contract R6219 DA-30-069-AMC-8(R) May, 1963

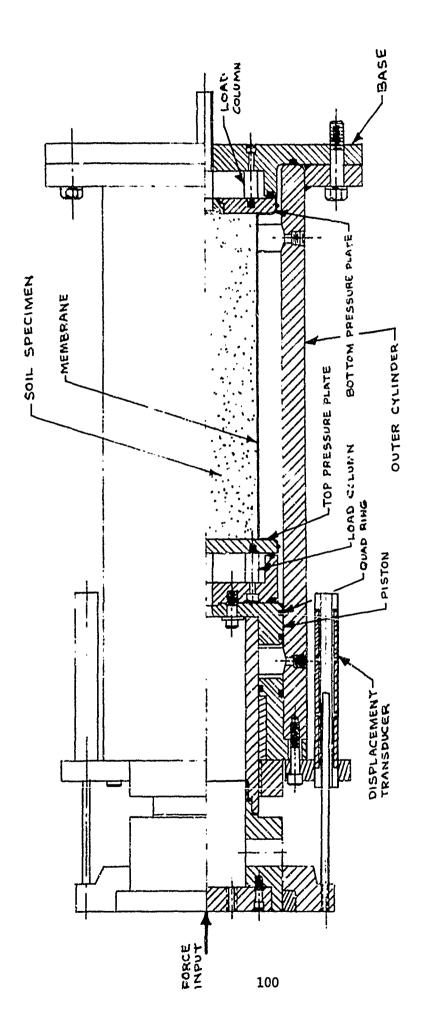
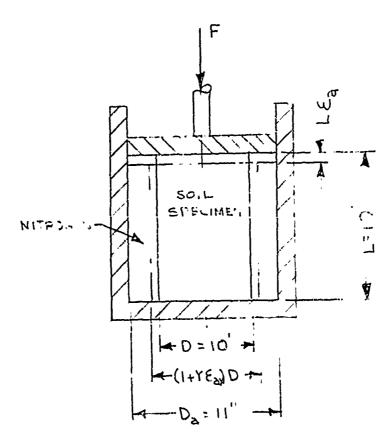


Figure 30. Dynamic Triaxial Apparatus.

the expanding specimen thus it acts as a part of the soil and does not add any appreciable strength to it. Therefore, it was felt that one of these two techniques would allow uniform specimen deformation, from which one could determine a representative value of Poisson's ratio of the specimen.

If the friction at the end plates could be overcome and a uniform deformation of the specimen could be realized, a method had to be devised to measure the lateral expansion of the soil. When the top plate moves downward, the sample expands in cross section and the volume of gas contained in the cavity surrounding the specimen decreases. Thus, the pressure increase in the gas can be attributed to both the downward motion of the piston and the radial expansion of the soil specimen. Since the downward motion of the piston is measured by the Schaevitz LVDT gages, the pressure increase due to this motion could be readily calculated and deducted from the total pressure increase to yield that pressure increase directly attributable to the radial expansion of the soil specimen. An example as worked out by Dr. Heierli is shown below.



Soil

E = 5000 psi (Young's Modulus)  

$$\gamma = 1/3$$
  
 $\Delta \sigma_{ax} = 100$  psi (axial stress increment)

$$\Delta \varepsilon_{ax} = \frac{100}{5000} = 0.02$$

Nitrogen

 $\Delta p = Unknown (psi)$ 

p = 15 psi (initial confining pressure  $V_0 = 165 \text{ in}^3$  (initial volume of qas)

 $\Delta V_a = (-165)(0.02) = -3.30 \text{ in}^3$  (change of volume of gas due to motion of plunger)

 $\Delta V_{e}$  = Change of volume of gas due to soil expansion

Now

$$\Delta V_{e} = -\left[\frac{\pi (1 + \gamma \epsilon_{a})^{2} D^{2}}{4} - \frac{\pi D^{2}}{4}\right] \left[L - L(\epsilon_{a})\right]$$

$$\Delta V_{e} = -\left(\frac{\pi D^{2} L}{4}\right) (2\gamma \epsilon_{a}) (1 - \epsilon_{a})$$

Substituting, we have

$$\Delta V_{p} \stackrel{\text{a}}{=} -10.25 \text{ in}^{3}$$

The volume of gas decreases by  $\Delta V_z + \Delta V_c = 16.80 \text{ in}^3$ , of which  $10.25 \text{ in}^3$  are due to the expansion of the soil. The corresponding pressure increase in gas pressure  $\Delta p$  is, with  $\chi = c_p/c_v = 1.40$  (the ratio of the specific heats at constant pressure and constant volume, respectively):

$$\Delta p - (-p)(k) \cdot \frac{\Delta V}{V}$$

(from  $p \cdot V^k = (const. for adiabatic compression)$ 

$$\Delta p = (-15)(1.4)\left(\frac{-10.25 - 3.30}{165}\right) = 1.73 \text{ psi}$$

The pressure increase can be measured with a precision differential pressure transducer. The pressure increase can be calibrated in terms of volume decrease of the gas in dynamic tests.

One problem area using this approach must be investigated; that is, the increase in gas pressure adds to the initial confining pressure. Therefore, the gas pressure increase must be small enough not to influence the test, but it must be large enough to be measured with relative ease. An attempt was made to assure that this situation would exist, but as yet it has not been proven.

# 6.1 Design

#### 6.1.1 General

Several designs of the dynamic oedometer were incorporated into the dynamic triaxial apparatus; namely, the pressure plate design and the use of strain gage columns to measure applied and transferred loads. It was intended to use the "Hyge" pneumatic device described in paragraphs 3.2 and 3.3, as the force applicator.

The triaxial apparatus consists of three main members, the outer cylinder, and the top and botton pressure plates. The soil specimen was to be prepared in a 4 inch diameter flexible membrane which was confined by the top and bottom pressure plates. A pressure pulse was to be introduced into the specimen through the guided top pressure plate. The displacement of this plate was to be measured by two diametrically opposed Schaevitz LVDT gages. The input pulse was to be measured both by the load column arrangement and a Baldwin-Lima-Hamilton Load cell placed between the applicator and the top pressure plate assembly. As the pressure pulse passed through the specimen it would be measured by the load columns in the bottom pressure plate assembly which is identical to the top plate assembly.

The movement of the top pressure plate is quided by a piston assembly placed in an oilite bushing. The piston itself has a double sealing arrangement which should keep leakage down to an extreme minimum. Since the piston moves quite rapidly, it was felt that quad sealing rings should be used to prevent twisting of the rings.

## 6.1.2 Pressure Plates

The average pressure at the top and bottom of the soil specimen is measured by identical arrangements of a pressure plate supported by four equally spaced load columns. The plates which form a part of the pressure plate assembly are designed to be as light as possible to minimize inertia effects, and still be strong enough to exhibit sufficiently small deflections. The mass above the load columns does not influence the inertial pressure loss of the plate.

When tests of gages imbedded in the soil specimen are desired, a terminal arrangement consisting of a Teflon plug and wire pins is inserted into the bottom pressure plate to facilitate the electrical connections. To minimize interference between the movement of the soil particles and the Teflon plug, the plug is centrally located and mounted flush to the pressure plate surface.

## 6.1.3 Load Columns

The average pressure at the top and bottom of the soil specimen is measured by load columns. These load columns are to be instrumented with standard resistive foil elements connected in a Wheatstone bridge arrangement. Piezoresistive strain elements were considered to instrument the columns because of their high gage factor; however, it was decided that their high temperature coefficients would produce many drift problems.

Foreign matter is kept out of the load column cavity by sealing off the volume with "O" rings. The "O" ring in contact with the pressure plate has minimum squeeze on it, so that it would not influence the pressure plate loading.

# 6.1.4 Cylinder

The outer steel cylinder was designed for 1000 psi internal pressure, with a substantial safety factor. Therefore, at all confining pressures, the lateral deformation of the outer cylinder will be negligible. Thus, the increase of gas pressure will be due entirely to the downward motion of the piston and the deformation of the soil specimen.

The inner surface of the cylinder is highly polished in order that a minimum amount of friction would be realized from the movement of the quad sealing rings.

# 6.1.5 Force Applicator

Pressure input into the triaxial apparatus was to be accomprished by a pneumatic loading device, the "Hyge" unit which was used on the earth shock tube. The assembly was to be mounted vertically on a concrete foundation with the "Hyge" unit anchored on a suitable frame assembly and the driving ram attached directly to the top of the triaxial piston assembly.

# 6.2 Testing

Because of limited funding, the dynamic triaxial apparatus was never assembled; therefore, no testing could be done.

#### 7.0 SUMMARY

In the course of the contract, effort was expended in several areas including

- 1. strain gage development
- 2. earth shock tube experimentation
- 3. development of laboratory equipment
- 4. development of a horizontal displacement meter

The contract funding was such that only a low level of effort could be expended by Atlantic Research Corporation.

As problems occurred in one area, effort was normally transferred to other areas of interest before solutions to the problems could be found.

The results of the program may be tabulated as follows:

# Earth Strain Gage

After modifications to original design, satisfactory results were obtained from these gages in some of the tests run in the earth shock tube.

## Earth Shock Tube

This equipment was designed and installed, and five series of tests were performed during the contract. An evaluation of the Teflon powder friction reduction system was not completed. In general, the use of the shock tube for evaluation of gages was not successful because of the anomalously high gage output which was recorded.

# Dynamic Oedometer

The unit was designed and instrumented under this contract, but tests and evaluations were conducted under a subsequent contract and are reported in Part II of this report.

# Dynamic Triaxial Apparatus

The triaxial apparatus was designed and fabricated but funding was never allocated for assembly of the system.

## Horizontal Displacement Meter

This seismometer was designed and field tested. The test results were evaluated and recommendations were made for improvements but no further work was carried out on this unit.

# 8.0 CONCLUSIONS AND RECOMMENDATIONS

Due to limited funds some of the studies were stopped before conclusive data were obtained. These investigations are noted below and the indications of the limited study are given.

#### a. Earth Shock Tube

The tests conducted with the earth shock tube showed that soil samples confined in tubes having large langth to diameter ratios cannot be used for one dimensional wave propagation studies. This confirms the conclusions of others (1). However, observations from the earth shock tube tests indicated that uniform density throughout the length of the sand sample was not achieved, and the Teflon powder layer between the sand and the confining tube was not sufficiently compacted. The tests conducted with the Teflon powder friction reduction system are inconclusive because of anomalous pressure measurements of the wave propagating through the soil sample. In an attempt to evaluate the behavior of gages embedded in the earth shock tube, it was found that the costs were excessive. it was decided that the gage behavior should be studied in the laboratory with the aid of laboratory devices such as the dynamic triaxial apparatus.

# b. Earth Strain Gage

During the limited testing of the earth strain gage, it apparently operated satisfactorily. However, because of the anomalous pressure measurements, no meaning could be attached to the values obtained with this device. Because of its simplicity and low cost, it is strongly recommended that further testing should be done with this gage. Its precision should be verified.

<sup>(1)</sup> R.V. Whitman, et al, "The Behavior of Soil Under Dynamic Loadings," Volume 3, Final Report on laboratory studies; Massachusetts Institute of Technology; Contract Number DA-49-129-Eng-227, August 1954.

# c. Dynamic Oedometer

The conclusions and recommendations for this phase of the project have been given in Part II of this report. As stated previously, it is highly recommended that effort be devoted to further evaluation of the dynamic oedometer.

## d. Horizontal Displacement Meter

The conclusions and recommendations for this phase of the project have been given in Part I, Section 5.0 of this report.

# e. Dynamic Triaxial Apparatus

The components of this device were fabricated, but due to lack of funds it was never assembled. Therefore, no conclusions could be drawn. However, the following recommendations are made:

- that the device be assembled and that initial work be directed to the study of the behavior of imbedded gages;
- 2) the results of the initial study should then be directed towards tests of the earth shock tube using the Teflon powder friction reduction system. The conclusions of these tests may evolve an economical laboratory device to be used for one dimensional wave propagation or yield information which will lead closer to the realization of such device.

# PART II DYNAMIC OEDOMETER

Contract DA-18-001-AMC-877(X)

## 1.0 INTRODUCTION

This section of the report is concerned only with the tests on soils performed under a later contract, DA 18-001-AMC-877(X), with the Ballistics Research Laboratories of the U.S. Army.

These tests made use of the dynamic oedometer and some other Part I of this report. items which have been described in The oedometer was designed to investigate soil compaction and imbedded stress gage behavior before further earth shock tube tests were carried out (Part I of this report). In addition to the investigation of stress gage behavior under conditions of one-dimensional stress, it was hoped that experiments could be conducted to determine accurate values of constrained modulus, Poisson's ratio, and other soil constants useful for propagation predictions. The test series was intended to measure, for certain chosen soils, the calibration factors for buried gages and values for dynamic behavior parameters in support of planned tests in the earth shock tube. However, the sets of measured data and other observations during the tests indicated unexpected results, specifically that the loading piston developed a tilt during tests, and that a uniform state of stress could not be easily achieved if lateral displacement of the sample was not permitted.

This report describes these tests and discusses some alternate mechanisms which may possibly have combined to cause these apparent findings in the tests performed.

## 2.0 DESCRIPTION OF TESTS

## 2.1 Apparatus

The dynamic oedometer is a laboratory device to be used to investigate soil compaction and mbedded stress gage behavior. It was also expected that the device would be useful in one-dimensional wave propagation studies. Figure 1 is a sectional sketch of the oedometer, and Figure 2 is a photograph of it assembled for use within a frame provided for the support of the loading device. The loading device shown is an ordinary hydraulic ram for static tests. For dynamic tests, the oedometer equipped with the Hyge Shock Tester, Figure 3, is capable of developing a maximum loading

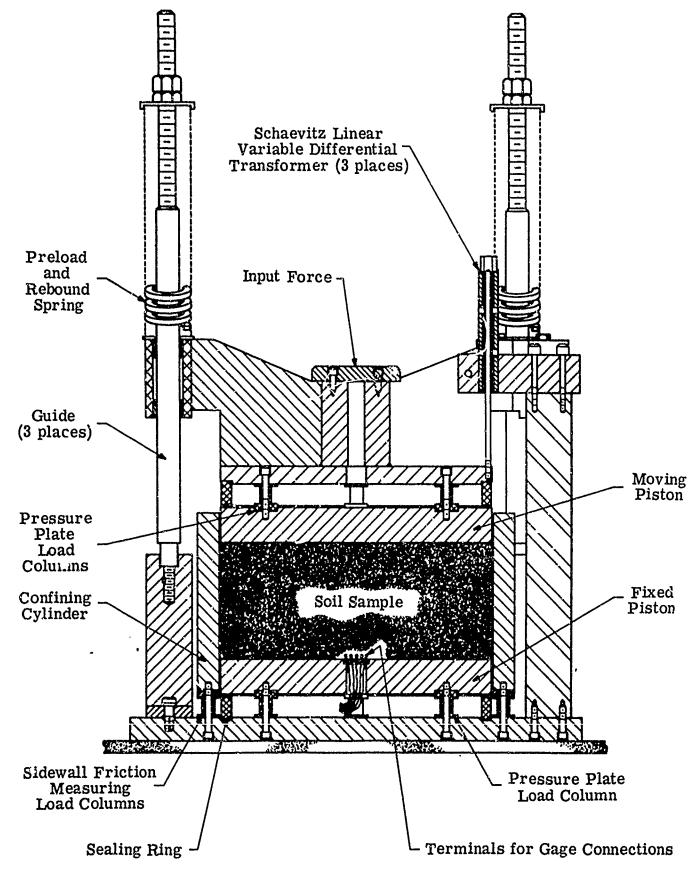


Figure 1. Cross Section of Dynamic Oedometer.

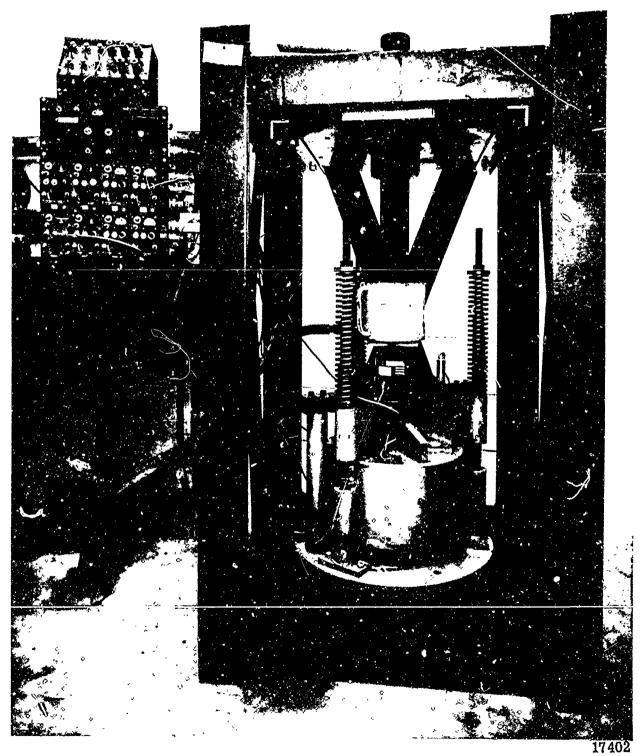


Figure 2. Oedometer with Hydraulic Ram Installed for Static Test.



Figure 3. Hyge Shock Tester Mounted on Oedometer.

of 40,000 pounds, with rise time of a few milliseconds when the motion of its piston is resisted by a test system of sufficient rigidity. It operates by the sudden delivery of high-pressure nitrogen gas to a cylinder and piston through a fast-acting valve. In static tests, to provide smooth and precise increases in force up to the peak value which was preselected for each test, several hydraulic rams were used of which the largest has a 60,000 pound rating. A 50,000 pound dynamic load had been used in the design of the oedometer, the loading frame, and other apparatus to which it might apply.

The details of the oedometer are given in Part I of this report and need not be repeated here. The oedometer has been designed to accept a cylindrical soil sample, nominally 12 inches in diameter and 6 inches high. It has a heavy steel base plate, to which a disc, serving as the bottom piston, and an outer wall, consisting of a machined heavy pipe section, are firmly attached by means of four spool-shaped forced links each. Also attached to the base plate are 3 heavy rods to guide the upper piston, and the supporting posts for gages to measure that piston's motion. The upper piston is a thick aluminum disc attached by four more force links into a heavy assembly of steel plates and discs to form a rigid unit. This unit distributes the concentrated loads delivered to its top and connects the piston with the bearings which slide freely up and down on the guide rods. To control rebound and to permit preloading a sample when desired, a compression spring extends up from each of the bearings to a . nut and washer on the threaded upper end of each guide rod. Actually, the upper and lower pistons themselves are identical. Each is a 1 1/2 inch thick aluminum disc, fitted to a 0.010 inch clearance with the oedometer wall. This opening was designed to be small enough to prevent intrusion of Ottawa Sand of the size retained on a No. 30 ASTM wire mesh, but large enough to be maintained open by the rigidity of the quide rods.

Gages attached to the oedometer parts permitted the following measurements:

The set of force links assembled to the upper piston measured the force delivered by the loading device to the upper surface of the soil sample.

The links under the lower piston provided a measurement of the force delivered through the sample and piston to the base.

The difference between these two quantities was considered to have been transferred to the walls by friction, and was measured by the set of force links which support the wall.

Strain gages were applied to the wall to detect the strain in a circumferential direction at two distances from the base to measure circumferential tension in the oedometer wall. Obviously, this strain results from the force exerted by the soil in its tendency to spread laterally under a compressive vertical load. To minimize this motion and so relate the measurement to Poisson's ratio, the wall had been designed far thicker than required for the design loads but was provided with broad vertical notches for the installation of the strain gage. The remaining measurement made on the oedc meter was of the motion of the upper piston relative to the base plate. Originally, the two linear-variable differential transformers, were mounted with their cores attached to the upper piston unit at opposite ends of a single diameter. During the test program, it was found necessary to increase the number of transformers to three, equally spaced around the edge of the head unit at a radius of 5 5/8 inches from its center.

## 2.2 Instrumentation and Calibration

Each of the four force links in each of the three sets was fitted with resistance strain gages. The bodies of these links were spool-shaped with a central hole for a mounting bolt. Because the four in each set connected two rigid elements, and because their cross section was held to a minimum to increase sensitivity, their installation and calibration presented a problem. In actual soil tests, all in

each set were to be electronically connected to provide a single output signal. With a set of four between two rigid plates, however, the output would not be linearly related to the load through the full range unless each in the set was equally preloaded. Otherwise, one or two might not start carrying load at the beginning of a c/cle, while another might be above its yield stress when the load approached the design value for the set. Furthermore an adequate preload was needed to provide proper operation during rebound in dynamic tests. These problems were solved by connecting each force link to give a separate electronic output before assembly of each set to the mating parts. By monitoring of these outputs, the lengths of the separate links could be adjusted by shimming and light filing to give simultaneous contact under no load, and the mounting bolts tightened to adjust each to the chosen preload. After assembly in this manner, each set was checked and found to have an output curve of adequate linearity and precision under the proper range of calibrating loads.

The strain gages reading tension in the oedometer wall were of the semiconductor type. These offered better temperature compensation at the expected low stresses than strain gages of the metallic type on the 1/4 inch thickness of steel left under the vertical notches in the wall where the gages were to be applied. Better compensation was achieved by building bridges with pairs of elements oriented at right angles to each other, having a positive and negative gage factor, and aligning only the proper one of each pair in the direction of the strain to be measured. Naturally, the element at right angles made the pair also sensitive to strain at right angles to that desired to be measured, which in this case would be expected to result from soil friction. However, the strain from such vertical forces was negligible, in comparison, when estimated values for both forces were each divided by the appropriate cross-sectional area for the Since the wall was 1 inch thick except at the notches, its horizontal cross section was nearly 60 times the area of the metal left at the notch. This conclusion was verified,

and the bridge calibrated, by subjecting the wall ring to several combinations of vertical load and internal air pressure. With the ring resting on a rubber sheet on the metal floor of the loading frame, the vertical load was imposed on its top by a hydraulic ram through a stack of steel plates and another rubber sheet. The vertical load was chosen to be more than needed to keep a seal between the ring and the rubber sheets and also to hold the plates down against the air pressure. Output of the ring tension bridge was monitored while the vertical load was imposed as well as while the air pressure increments were being added through a valve and gage system. The actual gages used were Kulite-Bytrex Corporation Types DB-102 and DBN-102.

Motion of the upper piston was measured by Schaevitz Engineering Model 1000 S-L linear variable differential transformers (LVDT). As mentioned, the number used was increased from two to three during the tests, after tilting of the upper piston had been observed. The cores for these gages are mounted vertically in threaded holes near the edge of the upper piston assembly, and the armatures were held in adjustable clamps supported by the oedometer base plate. The cores are adjusted to an appropriate zero position after the soil sample and the rest of the apparatus are in place.

In addition to the measurements of forces on, and motions of, the soil samples, provision was made for the operation of gages within the soil samples. Leads for such gages were connected through a fitting installed in the bottom piston. The number of internal gages varied from none in some tests to three in others. All gages used were Kulite-Bytrex Model HFA-1000 pressure cells. These were disc-shaped gages approximately 2 inches in diameter, having a rubber diaphragm which was stiffened by a thin metal disc and backed with oil which transmits pressure changes to the semiconductor-type transducer.

For both static and dynamic tests, data were recorded through the 3KC carrier amplifiers and other parts of a CEC System D conditioning unit, with other electronic elements as required, on a CEC Model 5-124 oscillograph. At most, 12 channels of data were recorded. Of these, mentioned above are the upper piston force, the lower piston force, the force delivered vertically by the sidewall, the tension in the sidewall, 3 channels of data on the upper piston motion, and as many as 3 records from the gages buried in the sample. In addition, in static tests a load cell was installed under the loading ram to check on the piston load cell outputs. This load cell was a Baldwin-Lima-Hamilton SR-4 Type C rated to 50,000 lb.

# 2.3 Auxiliary Equipment

Some auxiliary apparatus was needed to make the performance of soil tests possible with an oedometer of the size and type described. In addition to standard laboratory equipment for measurement and control of sample properties, the principal items were a jig for raising and lowering the movable piston, and a sleeve for placing a Teflon liner at the periphery of certain samples of soils. The jig shown installed in Figure 4, was designed to permit precise and deliberate handling of the assembly, and to eliminate shocks or other disturbances to the sample. With it, the oedometer can be in place in the loading frame during sample placement, removal, and all other parts of a test. The sample is thus provided with some isolation from environmental vibration and disturbance by the rigidity and weight of over a ton represented by the loading frame itself, as well as being protected from the obvious changes which would result from moving the assembled oedometer or man handling the loading piston. Figure 5 shows the sleeve for placing the Teflon liner which was simply a thin cylinder of sheet metal, which can be set in the oedometer and has a diameter 1/2 inch smaller. To provide a controlled sample with a thin lubricating layer of Teflon separating it from the oedometer wall, the sample is placed within the sleeve and the Teflon powder in the space between the sleeve and oedometer wall. The sleeve is



Figure 4. Jig for Handling Movable Piston Installed on Oedometer.

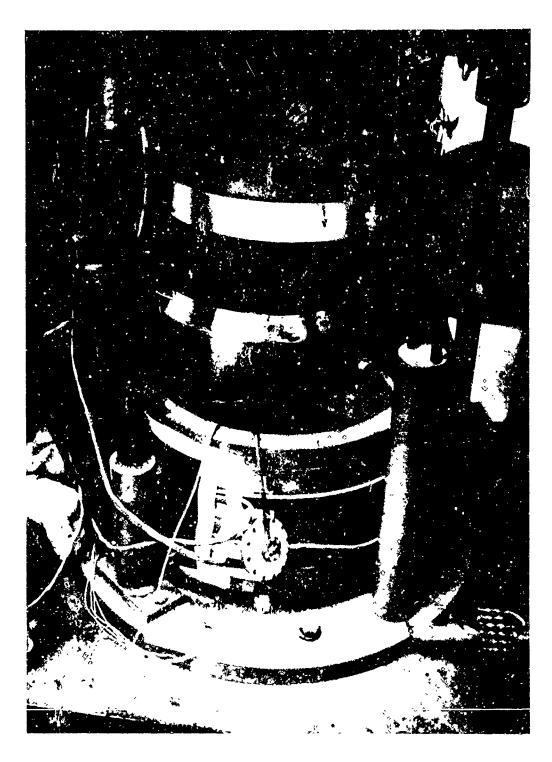


Figure 5. Sleeve for Placement of Teflon Powder Lubricant.

withdrawn slowly as the height of the sample is increased. This caused a minimum disturbance of the sample and Teflon lubricant.

# 3.0 TEST PROGRAM

The actual program consisted of a series of static and dynamic tests on a set of soil samples providing some variety in their properties. The planning of such a program was aided by the fact that all gages and instrumentation were useable in both static and dynamic tests without change. However, the difficulty in changing the loading device from the static rams to the Hyge Shock Tester, and the demand for use of the latter on other programs, prevented static and dynamic tests from being alternated or interspersed randomly. Therefore, it was originally planned to complete all apparatus debugging and calibration in static tests, before dynamic tests were tried. Actually, almost the entire effort was devoted to static tests, and only enough dynamic tests were done to confirm that similar behavior resulted. A data summary of the static tests is listed in Table I, and the summary of the dynamic tests is presented in Table II.

The emphasis and direction of this stage of the program changed as a logical response to certain observations in the earlier tests. From the start, great care was taken to control the uniformity and properties of the soil samples and to operate and align the apparatus to prevent disturbance and misfit. Each sample was to be placed in the apparatus very carefully and tested only once, in its virgin condition. The change in the nature of the planned program consisted in an increase in the intensity with which all such precautions were pursued, to insure that carelessness was not responsible for some of the unexpected observations. of instrumentation was also increased, and some tests were repeated, to insure that the observations were not a result of chance geometry. When the behavior had been demonstrated to be reproducible in static tests, only enough dynamic tests were performed to determine whether it also occurred under these conditions.

TABLE I. Data Summary of Static Tests Measurements at Point of Maximum Applied Load.

**)**;

		Density	È	Sample	Sample Height	Maximum	dein	Dispunction (mil	1							pedoca	Imbedded Guage Loads	T	Teflon
No.	Soil Type	Befor? (gm/cn <sup>3</sup> )	After (gm/cm³)	Before (in.)	After (in.)	Applied Load (K lb)	Left	Rear R	Right	Moving Piston (K lb)	Fixed Piston (K lb)	Cylinder (K lb)	Top (psi)	Bottom (psi)	15 E	Right (ps1)	Center (ps1)	Rear (psi)	Lubricant Used
				Tabi	Table I. Data	Summary of	of Static Tests Measurements	its Measu	rements	at Point	( Maximu	of Maximum Applied I	Load.						
2	Ottawa Sand	N.R.	ΝC.	3-38	N.R.	16	0.044	0.036	0.017	19.5	12.3	2.5							
91	Ottawa Sand	N.R.	N.C.	3-3 8	N.R.	7.	0.046	0.013	0.048	14.0	8.3	2,2							
17-1	Ottawa Sand	1.82		3-1	E.	2	0.055	0.047	970.0	0.0		<b>9</b>							
2:	Ottawa Sand		o z	3-9, 16	Z.	2 :	0.059	0.055	0.034	10.1	E.F.	1.7							
18-1	Oftawa Sand	1.20	, , , ,	3-9/16	z.	01	0.036	0.018	0.032	0.11		1.7							
18-2	O'tawa Sand		S.C.		N.R.	2	0.037	0.019	0.034	0.11	8.6	2.1							
19-1	Ottawa Sand	1.76	Z.C.	3-7 /16	N.R.	ę,	0.021	0.037	0.022	20	8,5	3.8			_				
19-2	Ottawa Sand		, N		N.R.	9	0.026	0.03	0.026	9.5	0.6	1.8				, -			
20	No Data																		
21	Ottawa Sand	1.70	O.N.	3-12	N.R.	2.9	0.00	0.015	0.038	5.6	2.3	0.3							
24	Ottawa Sand	1.60	1.66	4-1 32	3-7 8	2.5	0.038	0.014	0.024	2.4	1.9	0.5	13	5.8					
r.	Oftawa Sand	1.74	1.75	4-5 16	4-9.32	2	0.030	0.022	0.026	9.6	7.5	2.5	28	ä			45		
ю	Ottawa Sand	1.72	1.73	4-15, 32	4-7/16	3.75	0.039	9000	0.019	3.6	5.6	1.0	ü	82		_	9		
-	Ottawa Sand	1.74	1.75	4-5/16	4-9/32	01	0.036	0.019	0.027	6.6	6.7	5.9	*	22			80		
60	Ottawa Sand	1.63	1.65	4-3 /32	4-1, 32	2	0.051	0.037	0.037	10.0	7.4	2.4	7	£			20		
	Ottawa Sand	99.1	1.68	-	3-15/16	2.5	0.037	0.011	0.010	2,3	1.7	90	12	15			50		
30	Ottawa Sand	1.60	1.63	4-1 32	3-31 32	12.8	0.061	0.034	0.051	12.6	9,3	3.4	\$	59			7		
	Ottawa Sand	1.74	1.74	•	3-63.64	01	0.032	0.019	0.020	10.1	7.8	2.1	9	<b>\$</b>	24	24			
8	Fine Soil	1.30	N.C.	4-1 3	N.R	01	0.193	0.128	0.170	0,01	7.2	2.7	စ္တ	E.F.	49				
	Fine Soll	1.35	O.Z.	3-7-8	3-11.16	01	0.155	0.102	0.132	6.6	7:4	2.5	33	33	61	42			
34	Fine Soil	1.38	N.C.	3-15 16	3-25, 32	2	0.147	901.0	0.124	5.7	7.1	5,3	22	21	37	55			
'n	Oftawa Sand	02.1	N.C.	4-1 16	3-15/16	01	0.095	0.073	0.079	6.6	9.6	0.5	9	۲-	20	84			Yes
ట	Ottawa Sand	1.70	N.C.	3-15 16	3-13 16	2	0.1.3	0.068	0.077	10.2	9.7	9"0	7	vo	20	78			Yes
-	Ottawa Sand	1.78	18.1	3-31 32	3-15 16	2	0.054	0.026	0.033	10.0	2.0	2.9	42	‡	16	91			
38	Ottawa Sand	1.70	N.C.	4-31 32	Z.R.	01	0.104	0.072	0.085	6.6	9.6	0.5	91	19	8	96			Yes
•	Ottawa Sand	1.58	N.C.	4-3 32	3-7 8	9	0.149	0.094	0.121	9.7	6.3	0.7	s	91	23	88			Yes
ç	Oftawa Sand	1.56	N.C.	4-1 8	3-29 32	2	0.159	0.114	0.131	10.2	7.	6.0	2	19	67				Yes
<del>=</del>	Ottawa Sand	1.79	1.79	4-5/64	4-1 16	2	0.038	0.014	0.032	10.0	7.1	5.9	35	S	12	23			
42	Ottawa Sand	1.79	1.80	4-3/16	4-5 32	2	0.054	0.040	2. 3.	10.1	7.	2.8	8	29	Ξ	12		E	
<del>1</del> 3	Aberdeen Soil	0.85	N.C.	4-1/16	2-11 16	2	0.990	0.972	0.787	6.6	7.7	2.3	2	92	115	86	-	7	
‡	Aberdeen Soil	98.0	Z.C.	4-1/16	2-3/4	2	o.s.	0.901	0.751	6,0	7.2	2.6	E	32	126	76	_	7	
45	Ottawa Sand	59.1	N.C.	4-1, 16	v.R.	2	0.109	0.074	0.810	6.6	9.5	6.5	n	<b>-</b>	7.	101		63	Yes
46	Ottawa Sand	1.66	ָרָי הייני	4-5/15	4-5/32	0	0.123	0.920	0.870	6.6	9.5	0.5	s.	œ	75	83		26	Yes
#	Ottawa Sand	1.76	1.79	-	3-15 16	02	0.540	0.300	0.280	16.1	8.9	3.1	7	ş	ĸ	22	_	-	
<del>4</del> 8	Ottawa Sand	1.75	6 4	5 32	4-3/32	2	0.940	0.710	0.360	6.6	9.9	3.1	ж. Ж.	E.F.	-	45		=	
49	Ottawa Sand	1.82	N.C.	5-23 32	5-17, 32	10	0.151	0.117	0.121	6.8	9.2	6,0	53	ñ	86	69		63	Yes
20	Ottawa Sand	1.68	N.C.		N.R.	2	0.149	0.132	0.123	10.1	9.3	8.0	=	7.8	58	86		28	res
51	Ottawa Sand	1.69	N.C.	6-3/32	5-29/32	10	0.152	0.133	0.141	10.0	9.9	1.0	21	ĭ	66	క్ర	_	28	Yes
52	Ottawa Sand	1.69	N.C.	6-3/32	5-7/8	2	0.149	0.128	0.131	- 1	9.0	=	ž	2:	8	وَرْ		25	Yes
					Table II. Data	ta Summary	of Dynamic Tests	ic Tests	Measurements	٦Į	則	enic Load.			İ				
23	Ottawa Sand	1.73	1.79	4-14	4-7/32	N.R.				32	7.7	<u></u>	2	91	2	*		2	
ž	No Data																		
22	No Data		:			:	- 5	-		,	į		_;	1		i			
92	Ottawa Sand	1.69	o c	6-7/32	N.R.	<b>j</b>	0.109	0.135	0.121	n ;		7	77	15	7	13		ž	Yes
<u> </u>	Ottawa Sand	1.08	) (	8/1-8	or 'C1-6	3 5	6,144	6.13	551.0			÷,	S S	C.21	771	165		115	Yes
															,		_		

Abbreviations:

N.R. - Not Recorded

N.C. - Not Calculable fron Data Recorded

O.S. - Off Scale Reading

One of the more important changes in the program was a reduction in the number of types of soils used in tests. It was felt necessary to limit samples to only those which could be most uniformly and reproducibly placed in the apparatus. However, more than one type of sample was used, to eliminate the chance that the tilting was a phenomenon peculiar to a certain sample only. The samples used are indicated in the second column of Table I, and their properties and method of placement are described below.

The most frequently used type of soil was Ottawa sand, passing the No. 20 but retained on the No.30 sieve. This soil was most often placed in the oedometer in the dense state, by means of the "raining" technique. In other tests, it was placed more loosely, by careful pouring from a spoon held close to the surface. densities achieved by these methods are listed in the table, for both the before-and-after test condition. The densities were determined by weighing the amount of soil placed in the cedometer, and by measuring the depth of the sample at several points. measurement was made by placing a steel straight edge across the top of the cylinder at precise locations marked with a scribe, and measuring down from it to the surface with a ruler. To facilitate the measurement, a plate of thin aluminum sheet was set carefully on top of the sample. From the known weight of soil, and the measured volume it occupied in the oedometer, the density of the entire quantity of soil or of a particular layer or increment could be calculated.

Another type of soil used was the finer fraction of the soil available in sand bars of a small stream draining part of the company's property in Alexandria, Virginia. The sample was prepared by drying followed by passage through a No. 40 sieve. It can be best described as a silty, dirty, fine sand, containing a small proportion of material passing the No. 200 sieve, normally less than 3%, while less than 10% passed the No. 100 sieve. It was always used in the dry condition and placed loosely. The grains of this soil were extremely angular and sharp-cornered, having been produced by erosion very close to the point found.

A further type of soil was the material found at the site of the Ballistic Research Laboratories installation where field tests were performed on soils by BRL personnel. The location is close to the Chesapeake Bay waterfront of the installation. The soil is a clayey, silty, fine sand, typical of such a beach deposit. It is quite similar to the local Virginia material described above, except for a larger propertion of fine material and less angularity of the grains. In the present case, it was used at the field moisture content, found to be 13% to 14%, but in a disturbed state and at a very low density, produced by forcing it while moist through a No. 4 sieve. This resulted in a flocculated, loose, and open structure.

When such fine soils were used, some technique to keep them from the clearance space between the pistons and the oedometer wall was, of course, necessary. The method used was to provide a layer of Ottawa sand, intended to be one-half inch thick, above and below the layer of finer soil. Consideration of the relative grain sizes indicates that these materials are in proper relation to serve as a filter course for the 0.010 inch clearance. The weight and height of the sand layers and of the other layer were, of course, measured separately, to permit density calculations. The Ottawa sand was "rained" into place in the bottom layer, but the top layer had to be placed in the loosest and gentlest condition because of the possible disturbance and compaction of the very weak and fragile fine soil structure.

In a number of samples, a layer of Teflon powder was placed between the sample and the cylinder wall, to investigate the effect of reducing the side friction. To produce this layer, a jig was used consisting of a sleeve of thin copper sheet 1/2 inch less in diameter than the oedometer's internal diameter. With this sleeve centered in the oedometer, soil was placed within the sleeve and Teflon powder in the annular space in shallow layers, as low as 1/4 inch in the case of denser soil samples, and the sleeve was withdrawn with care for that height before the next layer was added. In Figure 6 the typical state of the Teflon powder after a test is shown. After compression the Teflon was firmly compacted and adhered to the oedometer wall, as shown.



Figure 6. Molding of Teflon Powder Lubricant After Loading.

Except for two cases where failures of electronic channels were being checked, each sample was subjected to load only once, and was then removed from the oedometer. Care was taken, of course, to record the before and after density, height, and visual evidences of tilt where such were observed. As noted, however, visual evidence of tilt was found very difficult to photograph.

During the program, 58 sets of data were recorded. Of these, 15 were basic calibrations of the oedometer and its associated instrumentation. Of the 43 actual tests, 37 were performed under static loading conditions and the remaining six were run under dynamic loading conditions, using the Hyge Shock Tester.

Twenty-nine of the static tests were completely successful and 7 were partially successful when considered from the stand-point of data output. One of the six dynamic tests appears to have been completely successful and three appear to have been partially successful. In the remaining tests, no useable data were recorded.

Under all test conditions, the data from the LVDT's indicate a tilting of the movable piston. In all but three of the static tests, the data indicated that the same point on the movable piston underwent the greatest displacement. These three tests were among the first six performed.

Under the dynamic loading conditions, the movable piston was again tilted, but the point of greatest displacement was different than in the static tests.

Table I is a brief summary of the static tests. The data listed in the table are for the condition of maximum applied load. The data in Table II are the recorded values of the various parameters at peak dynamic load. It will be noted that several numbers are missing from the series of test numbers. These numbers were used for instrumentation calibration.

Complete data for all the tests are listed in the appendix.

## 4.0 TYPICAL TEST DATA

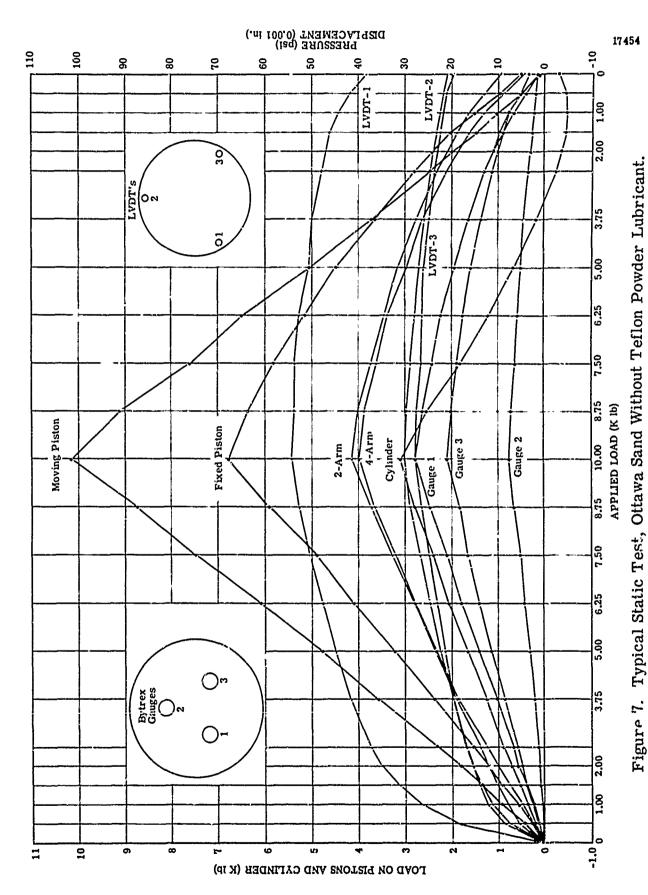
# 4.1 Static Tests

Two basic sets of static tests were conducted during the program. The only difference between the test procedures was the use of a Teflon powder lubricating layer in one set of tests. The soil samples used in these tests were both Ottawa sand, but as noted on the curves, there were small differences in the densities and original heights of the samples. Because of this, an absolute correlation of the two tests cannot be achieved; however, since these tests are very similar to the other tests in the series, generalized trends can be studied.

Figure 7 presents curves of 11 parameters plotted against the applied static load. This particular sample had an original density of 1.76 gm/cm<sup>3</sup> and an original height of 4 inches. The final density was about 1.79 gm/cm<sup>3</sup>, and the total permanent compression approximately 1/16 of an inch.

The relative positions of the LVDT's and the imbedded gages are shown in the sketch at the right side of the figure. The curve marked "upper strain gage" is a measure of the radial stress in the soil sample near the top of the cylinder, while the "lower strain gage" curve is a similar measurement made near the base of the cylinder.

From the first application of load, a relatively large difference is noticed in the loads as measured by the moving and fixed pistons. This difference is approximately equal to the value of the axial load in the cylinder. It is also noted that during the initial loading period, the force measured by the moving piston is less than the applied load. At the initial point of loading, an extremely large discrepancy is observed in the displacements of the three LVDT's. This difference persists throughout the tests. The three imbedded gages also show large differences in output, as well as anomalously low stress levels. In this test, the upper and lower train gages measuring the radial stress in the soil are in reasonably good agreement. (This is not completely typical of the test series.)



As the load is reduced on the sample, it is noted that the compression of the soil is decreased as is expected, and that the amount of elastic recovery is somewhat proportional to the maximum displacement. The most interesting phenomenon observed during this relaxation period is that a point is reached at which the load on the fixed piston exceeds the load on the moving piston, which is approximately equal to the applied load. At approximately this same point the axial load in the cylinder becomes negative, indicating that the load columns supporting the cylinder have gone into tension.

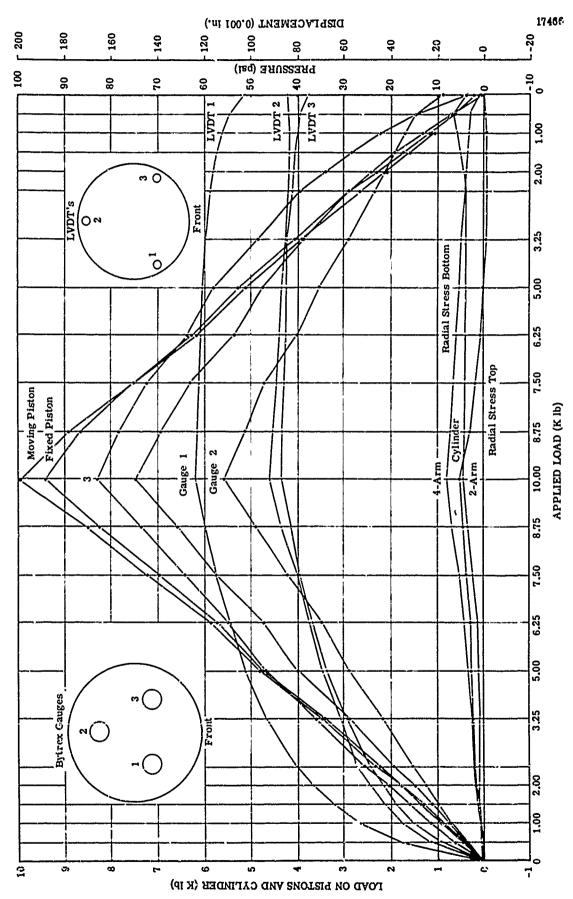
Figure 8 presents data from Test 46 which is quite similar to that discussed above. The original soil density was 1.66 gm/cm<sup>3</sup> and the original sample height was 4 5/16 inches. The total permanent compression was approximately 5/32 of an inch. The major difference between this and the preceding test was the use of a Teflon powder lubricating layer between the sample and the cylinder wall.

The discrepancy in the displacement measurements was similar to that noted in previous tests. Several very significant changes were noted in this test. There was much closer agreement between the loads measured on the fixed and moving pistons accompanied by a corresponding decrease in the cylinder axial load. The stresses measured by the imbedded gages, while not at all in agreement were much higher than in the previous test, and the radial stresses in the soil as measured by the upper and lower strain gages were greatly decreased and not nearly as well in agreement.

The pheromenon of the pressures on the fixed and moving pistons observed in the previous test during the relaxation of the load recurred but on a much smaller scale.

## 4.2 Dynamic Tests

Only six dynamic tests were attempted during the program. Of these, two were complete failures in that no data was obtained. Three of the remaining tests had a failure in one or more data channels.



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The data of these tests are listed in Table II. Two successful tests were performed without the use of the Teflon powder lubricating layer. In the last three tests, the height of the soil sample was greater than that used in all but the two final static tests.

Test 56 compares reasonably well with Tests 51 and 52. The only major change is the greater frictional force measured in the cylinder.

Tests 52, 57 and 58 cannot be easily compared with any of the static tests because of the magnitude of the load applied to the sample.

In these tests, the exact values of loads reported are somewhat questionable. The rise times of the pulses were very short and there was a tendency for some of the traces to overlap, thereby making it extremely difficult to determine precisely the peak values.

#### 5.0 OBSERVATIONS

# 5.1 Piston and Cylinder Loads

In general, the applied load as measured by the load cell and the load measured by the strain gage bridge attached to the upper piston do not agree too well. Normally the load measured on the piston is lower than the applied load by up to 5% as the load is increased from zero. In some of the tests, good agreement is achieved when the applied load is in the 5000 to 7500 pound range. In approximately 25% of the tests, the peak load measured by the moving piston was in excess of the applied load, but in most of these cases it is felt that this difference is within the measurement accuracy of the system.

In the tests run with no lubricating layer to reduce the side wall friction in the oedometer, it appears that a large percentage of the applied force is transferred to the cylinder. In all but a few cases, the magnitude of this friction force was 20% to 31% of the applied force.

When a Teflon lubricating layer was utilized, the frictional force was reduced to a value of between 5% and 10% of the applied load.

In tests 15 through 20 (see complete data in Appendix), the values of the friction forces reported are generally lower than those measured in the remainder of the tests. On review of the original data sheets, some question is raised as to the validity of the initial calibration check run before each test.

In all the tests, it was observed that as the load was relaxed, a point was reached when the force on the fixed piston exceeded the force on the movable piston. At approximately the same point, the axial cylinder load became negative. In most tosts, the difference between the force on the movable and fixed piston is about the same as the negative load on the cylinder. It was also noted that the load on the fixed piston was greater than the applied load.

A possible explanation exists for this phenomenon. As the axial load was increased to a maximum, the radial stress also increased to a maximum, but as the axial load was reduced to zero, the radial load decreased, but approached some final values. This residual radial stress creates a friction force tending to raise the cylinder, thus transferring the cylinder weight to the fixed piston along with part of the bolt tension in the cylinder load columns.

# 5.2 Cylinder Radial Loading

The radial loading in the cylinder was greatly decreased by the use of Teflon powder. This powder was loosely placed between the cylinder and the sample. It is believed that the powder was so loose that it compacted due to radial displacement in the sample. This allowed the sample to strain appreciably in the lateral direction without creating a true force indication in the cylinder wall, thus destroying the one-dimensional character of the tests. This is similar to what one would expect if the sample were confined in a flexible cylinder, since the load carrying (or transmitting) ability of the Teflon is very small compared to the cylinder.

It is believed that the radial force measured when Teflon was used is not an accurate measure of the strain and therefore is not useful for computing Poisson's ratio.

In the tests performed without the use of Teflon, the radial loading measured was considerably higher, but because of the frictional forces a state of one-dimensional stress suitable for computing Poisson's ratio was not achieved. (1) It is believed that this effect may be considerably reduced by decreasing the volume of Teflon.

# 5.3 Piston Displacement

When the compression of the soil sample measured by the LVDT was compared with the depths recorded before and after a test, Tables I and II, it was noted that the LVDT's generally indicated a lower compression than was measured manually. It is felt that this discrepancy is in part due to some initial unrecorded compression during placement of the piston and the loading apparatus. In Test No. 43, when a very low density soil was used, a compression of approximately 3/8 of an inch was recorded during placement of the piston and ram. During Test No. 44, a compression of 1/2 inch was recorded during this initial period.

With the higher density soils, it is believed that this initial compression during placement of the loading appartus was so small that it was not readily observed by the personnel running the tests, and therefore, was not recorded. (1)

The displacements recorded in Tables I and II are the maximum displacements under load. As the load was decreased on the sample, the displacement also dropped to a lower value in all tests except No. 43 and No. 44. In view of this fact, it is felt that some other factor(s) must have an effect on the displacement discrepancy. One factor could be the method of measuring the soil depth before and after the test. The tolerance involved in this measurement technique was about  $\pm 1/64$  inch.

(1) "Two dimensional wave propagation experiments in soils" - prepared under Aberdeen Contract R6219.

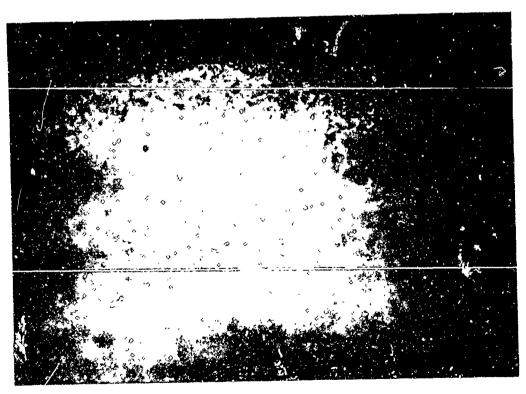
During the testing, a tilt of the movable piston It was indicated assembly under load was quite pronounced. in two ways. The most certain indication was a difference in the output signal of the LVDT's measuring the vertical travel of different points on the assembly. The other was a pattern which was found on the surface of a number of samples when the upper piston was removed after a test. This pattern was a fine ridge on the surface of the sample near its edge and extending for some distance around its circumference. The pattern was difficult to photograph, The photographer's difficulty as can be seen from Figure 9. arose principally from the extent to which the apparatus limited the space and geometry within which the lights and camera could be arranged, although the pattern was quite visible to the naked eye. The sight of this obvious pattern first led to the speculation that tilting of the piston might be occurring, and thus caused the test procedure and apparatus to be modified and extended to permit checking of this tilting.

## 5.4 Imbedded Gages

The original dimensions of the oedometer were chosen so that the height/diameter (H/D) ratio of the soil specimen could be made quite small. By so doing, the expected high friction forces at the soil-cylinder boundary could not have a major effect on the central area of the soil where the gages are imbedded. The stress state on the gages should thus be nearly one-dimensional However, the measurements made by the imbedded gages in the test without Teflon show values consilerably lower than the average pressure applied to the sample. This might result from certain arching effects across the gages, or it might reflect an effect of side friction because the three gages were actually placed outward from the center area. However, the same trend was observed in Tests 25 through 30 when one gage was placed at the center of the sample, and in Tests 31 through 40 when two gages were placed on a single diameter about 3 inches on each side of the axis of the oedometer. To determine if this were so, further tests - with varying H/D ratios - should be made for comparisons.



Figure 9a. View of Sand Ridge Formed on Upper Surface of Soil Sample.



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Figure 9b. View of Sand Ridge Formed on Upper Surface of Scil Sample.

Figure 9a and 9b. Two Views of Sand Ridge Formed on Upper Surface of Soil Sample.

No conclusions can be drawn from the fact that gage measurements made with the Teflon present were considerably closer to the applied load. Too much radial strain occurred, and the one-dimensional state of stress was no longer present.

The highest readings recorded by the imbelied gages occurred during Tests No. 43 and 44 when Aberdeen soil was used. As was stated previously, this soil was very fine and was less angular than the other soils tested. It is felt that these properties, coupled with the fact that soil was placed in the oedometer in a very low density state (0.8 gm/cm<sup>3</sup>), allowed the soil to strain appreciably in the lateral direction, thus eliminating the tendency of the soil to bridge over the gages. The angularity of the soil particles may be the most important factor in determining the ability of soils to arch or bridge.

## 6.0 DISCUSSION OF RESULTS

In the course of the test program, it became apparent that a true one-dimensional state of stress could not be achieved with the present oedometer design.

The use of Teflon powder as a lubricant did reduce the frictional forces in the oedometer, but introduced lateral strains too large for one-dimensional conditions to be maintained.

Tilting of the upper piston with load application was a second factor contributing to the lack of one-dimensionality in the oedometer. The causes of the tilting could have been (a) binding of bearings on the guides, (b) inaccurate alignment of the guide assembly and/or loading apparatus, (c) binding of the piston, and (d) variation in height across the surface of the sample. The consistency of direction of tilt in various tests of one type of loading leads one to discount the last two possibilities (which would have caused random tilting). Future testing with the oedometer must include instrumentation to determine the cause of the tilt, and modifications to eliminate it.

#### 7.0 SUMMARY

A series of tests using the dynamic oedometer filled with a number of different sand types has been performed. During these tests, difficulty was encountered in maintaining a one-dimensional state of stress in the region of the sand mass where the gages were inbedded. The causes of the difficulty were (a) tilting of the piston head, (b) side friction effects and (c) lateral expansion into the Teflon powder when this was used to reduce the friction effects. Until these problems can be solved, the oedometer will not be suitable for the stated purposes of gage calibration and measurement of the soil constants necessary for propagation prediction.

#### 8.0 RECOMMENDATIONS

Although the test series discussed in this section of the report did not achieve the desired results, it did point out the difficulties involved in attempting to simulate a condition of one-dimensional stress in the laboratory.

Before the dynamic oedometer is "written off" as a useful laboratory instrument, further effort should be expended to determine the cause of the tilting of the upper piston. Once this problem has been solved, a series of tests should be performed with varying height-to-diameter ratios to determine if the friction effect can be sufficiently reduced without the use of a lubricant. In the performance of these tests, gage locations should be selected so that a profile of readings across the oedometer could be obtained.

If the friction effects can not be sufficiently immunized by the proper choice of H/D ratio, a test series should be performed to determine if varying the thickness and/or density of the Teflon powder lubricant would result in a one-dimensional state of stress.

# APPENDIX DYNAMIC OEDOMETER TEST DATA

#### APPENDIX

#### GENERAL:

The following pages are a complete tabulation of all data recorded during the test program.

At the top right hand corner of each data sheet, a pair of sketches, showing the approximate angular position of the imbedded gages and the linear variable differential transformers, is provided so that gage readings and displacements can be correlated.

In the tests using one imbedded gage, the gage was positioned at the approximate center of the sample. When more than one gage was used, the gages were positioned angularly as shown at a radius of approximately 3 inches (1/2 the radius of the oedometer).

Tests 15 through 52 are static tests. Tests 53 through 58 are tests under dynamic loading. In the dynamic tests, the first line of data is under peak loading conditions and the second line is a steady state condition.

EF - EQUIPMENT FAILURE
NC - NOT CALCULABLE
FROM DATA RECORDED IMBEDDED GAGE LOADS NR - NOT RECORDED NO. 2 Š. NO. 1 psi SYME JLS: BOTTOM NR NR RADIAL LOADS psi GAGE LOCATIONS LVDT LOCATIONS TOP NR psi 00 LVDT NO. 3 in/0.001 14 14 17 DISPLACEMENTS LVDT NO. 2 in/0.001 Test stopped at 16000-lb load because of tilting of 36 28 34 31 29 31 HEIGHT OF IMBEDDED GAGES ABOVE BUTTOM OF SAMPLE BEFORE LOADING LVDT NO. 1 in/0.001 38 38 44 33 41 DATE \_ 3-3/8 inches Ungraded Ottawa Sand CYLINDER 500 2000 2500 1500 1500 0 5 MEASURED AXIAL LOADS NR KK BEFORE LOADING \_\_ SAMPLE DENSITY: BEFORE LOADING \_\_\_ FIXED PISTON Ib 12000 10500 500 2500 5500 9000 7000 5000 0 AFTER LOADING\_ AFTER LOADING\_ moving piston MOVING 14000 4500 9000 5000 0006 13500 19000 4 SAMPLE DESCRIPTION SAMPLE HEIGHT: TEST NUMBER APPLIED LOAD REMARKS:\_  $\infty$ 16 12 ω 4 K-lb

NO. 3 psi

EF - EQUIPMENT FAILURE NC - NOT CALCULABLE FROM DATA RECORDED IMBEDDED GAGE LOADS NR - NOT RECORDED NO. 2 psi NO. psi SYMBOLS: BOTTOM NR NR RADIAL LOADS psi L VDT LOCATIONS GAGE LOCATIONS TOP psi NR °0⊲ In/0.001 L.VDT NO. 3 37 49 43 46 43 DISPLACEMENTS L VDT NO. 2 in/0.001 10 12 10 13 HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 1/11/67in/0.001 LVDT NO. 1 34 හ 47 43 41 DATE \_ 3-3/8 inches Ungraded Ottawa Sand CYLINDER 1000 1500 2250 1250 0 MEASURED AXIAL LOADS NR N. NR SAMPLE DENSITY: BEFORE LOADING \_\_\_\_ BEFORE LOADING AFTER LOADING.... AFTER LOADING \_\_\_\_ FIXED PISTON 6500 6500 3500 8250 4250 250 16 MOVING PISTON IE 14000 9250 5750 9250 5250 SAMPLE DESCRIPTION SAMPLE HEIGHT: TEST NUMBER APPLIED LOAD KEMARKS. ¥-18  $\simeq$ 4  $\infty$  $\infty$ 

NO. 3

psi

SYMBOLS:  NR - NOT RECORDED  EF - EQUIPMENT FAILURE  NC - NOT CALCULABLE  FROM DATA RECORDED	IMBEDDED GAGE LOADS	NO, 1 NO, 2 NO, 3																	
l	RADIAL LOADS	ВОТТОЖ			·														
LVDT LOCATIONS GAGE LOCATIONS	RADI	TOP																	
C A C C C C C C C C C C C C C C C C C C	175	L VDT NO. 3	0	19	22	24	26	25	25	24	25	30	31	33	34	34	33	32	31
9 01NG	DISPLACEMENTS	L VDT NO. 2 in/3.001	0	30	41	47	47	47	48	43	27	38	44	55	47	45	43	44	29
2/16/66 h 3 S BEFORE LOADIN		LVDT NO. 1	0	32	44	20	52	54	50	47	31	46	30	54	29	57	54	20	38
SAMPLE DESCRIPTION Ungraded Ottawa Sand  Grain Size > 0.0165 inch  SAMPLE DENSITY: BEFORE LOADING 1.82 gm/cm  AFTER LOADING NC  AFTER LOADING 3-1/4 inches  AFTER LOADING NR  HEIGHT OF IMBEDDED GAGES ABOVE EDITTOM OF SAMPLE BEFORE LOADING	T LOADS	CYLINDER	0	550	740	1170	1600	740	620	185	09	370	820	1230	1670	740	0	-370	09-
ON Ungraded  BEFORE LOADING  BEFORE LOADING  BEFORE LOADING  AFTER LOADING  AFTER LOADING  BETORE LOADING	MEASURED AXIAL LOADS	FIXED PISTON	0	2100	4270	6110	EF												छंड
PTION Ungrad  Trion Grain S  Trion Grain S  Trion Grain S  AFTER LOADING  AFTER LOADING  AFTER LOADING  EDDED GAGES ABOVE	MEA	MOVING PISTON	0	2320	4640	7360	10000	7620	4990	2500	0	2230	4640	7220	10180	7500	4820	23 20	0
SAMPLE DESCRIPTION SAMPLE DENSITY: BE SAMPLE HEIGHT: BE AF HEIGHT OF IMBEDDED REMARKS:		APPLIED LOAD K-1b	0	2.5	ည	7.5	10	7.5	2	2.5	0	2.5	വ	7.5	10	7.5	വ	2.5	0

			SYMBOLS:	NR - NOT RECORDED	EF - EQUIPMENT FAILURE	FROM DATA RECORDED			
	ļ	Oc	$\begin{pmatrix} 1 & 2 \\ 2 & 1 \end{pmatrix}$	ر در	L VDT LOCATIONS	(			GAGE LOCATIONS
TEST NUMBER $\frac{18}{18}$ DATE $\frac{2/17/67}{1}$	SAMPLE DESCRIPTION Ungraded Ottawa Sand	Grain Size > 0.0165 inch	SAMPLE DENSITY: BEFORE LOADING $-1.70~\mathrm{gm/cm}^3$	AFTER LOADING NC	SAMPLE HEIGHT: BEFORE LOADING_3-9/16 inches	AFTER LOADINGNR	HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING	REMARKS. Calibration of gauges measuring radial load is	questionable,

APPLIED	MEA	MEASURED AXIAL LOADS	AL LOADS		DISPLACEMENTS	NTS	RADI	RADIAL LCADS	IMBED	IMBEDDED GAGE LOADS	LOADS
	MOVING PISTON	FIXED PISTON	CYLINDER	LVDT NO. 1	LVDT NO. 2	LVDT NO. 3	TOP	воттом	NO. 1	NO. 2	NO. 3
_	-    -  -	2	2	00.00	in/0.001	in/0.001	psi	psi	psi	psi	psi
	*	ءَ -	<b>&gt;</b>	<b>&gt;</b>	ء 	0	0	0			
	2800	2200	260	21	<u>ص</u>	12	27	12			
	2300	4400	1000	28	12	24	45	23			
	8000	7000	1400	33	16	28	29	32			
	11000	9200	1750	36	18	32	72	40			
	8200	0003	810	36	17	30	70	36			
	5500	2900	0	33	16	28	64	31			4 to-
	290€	3300	-440	30	14	25	20	24			
	330	250	-80	20	11	15	10	4			
	3200	2500	200	28	13	24	15	œ			
	2600	4800	1100	31	15	27	27	17			<del></del>
	8200	7300	1500	34	17	30	39	22			
	11000	0086	2100	37	19	34	20	31			-,
	8500	8000	1000	36	17	32	48	26		<del>,</del>	
	2800	6100	190	35	17	30	45	24			
	\$200	3800	-250	32	15	28	36	17			
	1300	420	09	22	12	17	4				
ļ	**************************************		1				_			-	

NC - NOT CALCULABLE FROM DATA RECORDED EF - EQUIPMENT FAILURE NR - NOT RECORDED SYMBOLS: GAGE LOCATIONS LVDT LOCATIONS Calibration of gauges measuring radial load is HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE EEFORE LOADING 0.0787 > Grain Size > 0.0165 inch 3-7/16 inches SAMPLE DENSITY: BEFORE LOADING  $1.76~\mathrm{gm/cm}^3$ DATE Graded Ottawa Sand NC NR BEFORE LOADING \_\_ AFTER LOADING \_\_\_ AFTER LOADING\_ questionable SAMPLE DESCRIPTION SAMPLE HEIGHT: TEST NUMBER .\_ REMARKS:\_

L.OADS	HO. 3	psį																	
IMBEDDED GAGE LOADS	NO. 2	isd																	
IMBED	NO. 1	psi																	
RADIAL LOADS	воттом	psi	0	7	13	19	25	22	18	11	ଷ	œ	12	20	26	24	20	14	3
RADIA	TOP	psi	0	***	4	က	7	4	4	4	က	က	<del></del>	-1	က	-3	F-	-	3
TS	LVDT NO. 3	in/0.001	0	12	<u></u>	20	22	21	20	18	14	20	22	24	26	25	24	23	17
DISPLACEMENTS	LVDT NO. 2	in/0.001	0	18	56	31	37	35	31	28	18	28	32	37	39	37	36	33	21
	LVDT NO. 1	in/0.001	0	7	14	18	21	20	18	14	10	18	20	23	56	25	23	20	14
L LOADS	CYLINDER	4	0	490	925	1300	1790	926	309	247	0	555	686	1480	1910	686	247	-247	0
MEASURED AXIAL LOADS	FIXED	9	0	2080	4310	6340	8460	692:0	5080	2540	154	2080	4150	6210	0006	6920	5150	3000	231
MEA	MOVING	9	0	2210	4590	7000	93 50	7030	4670	2130	-82	2130	4510	7030	9510	7120	4670	2290	0
	APPLIED LOAD	K-Ib	0	2.5	വ	7.5	10	7.5	വ	2.5	0	2.5	വ	7.5	10	7.5	ນ	2.5	0

F390-1-68

NC - NOT CALCULABLE FROM DATA RECORDED IMBEDDED GAGE LOADS EF - EQUIPMENT FAILURE NR - NOT RECORDED NO. 2 ps. МО. П psi SYMBOLS: BOTTOM RADIAL LOADS 9.4 14.4 7.5 LVDT LOCATIONS GAGE LOCATIONS 9.5 14.7 6.0 TOP n/0.001 LVDT NO. 3 8.3 16.7 23.7 34.4 32.4 31.3 28.3 DISPLACEMENTS questionable. Measurements at 1.02 K-lb load also LVDT NO. 2 in/0.001 0.0787 inch > Grain Size > 0.0165 inch 3.≿ 9.6 13.5 15.1 Calibration of gauge, measuring radial load is 13 HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 2/23/67 in/0.001 LVDT NO. 1 2.9 5.6 9.2 8.3 DATE 3-1/2 inches SAMPLE DENSITY: BEFORE LOADING  $1.69~\mathrm{gm/cm}$ CYLINDER Graded Ott. va Sand MEASURED AXIAL LOADS 185 247 309 62 -62 -62 0 BEFORE LOADING FIXED PISTON Ib 385 308 1230 2310 1390 462 AFTER LOADING\_ 1080 AFTER LOADING, questionable 21 MOVING PISTON 328 574 410 1480 2630 1480 1070 0 SAMPLE DESCRIPTION SAMPLE HEIGHT. TEST NUMBER APPLIED LOAD REMARKS:\_ ₹ 5 1.02 0.46 1.63 2.90 0.46 1.51 1.11

NO. 3

psi

TER LOADING 1.66 gm/cm 3  FER LOADING 4-1/32 inches  TER LOADING 3-7/8 inches  FROM DATA RECORDED  FROM DATA RECORDED  GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING	SAMPLE DESCRIPTION Graded Ottawa Sand  0.0787 inch > Grain Size < 0.0165 inch  1.00 gm/cm	SYMBOLS:
inches sample before Loading	Si	
	AFTER LOADING 3-7/8 inches HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING	FROM DATA R

*ED* AILURE ABLE ECORDED

GAGE LOCATIONS

REMARKS. Moving piston appeared tilted before load was applied

		_												 	 	
LOAUS	NO. 3	psi														
IMBEDDED GAGE LOAUS	NO. 2	psi														
IMBED	NO. 1	psi														
RADIAL LOADS	воттом	pe:	0	0.9	2.0	3.0	4.4	5.8	5.2	4.6	3.1	1.8	1.0			
RADIA	TOP	şsd	0	2.9	5.7	8.4	10.5	13.1	12.0	11.3	8.9	6.3	3.4			
175	LVDT NO. 3	in/0.601	0	15	18	21	22	24	23	23	22	22	18			-
DISPLACEMENTS	LVDT NO. 2	in/0.001	0	2	6	12	14	15	14	14	13	13	14			
	LVDT NO. 1	in/0.001	0	18	31	33	35	38	38	37	35	33	27			
L LOADS	CYLINDER	1b	0	116	232	291	407	523	291	116	0	-58	- 58			
MEASURED AXIAL LOADS	FIXED	15	0	312	704	1094	1485	1875	1640	1407	1094	625	78			
MEA	MOVING	16	0	345	922	1293	1897	2415	1982	1638	1034	604	0			
	APPLIED LOAD	K-lb	0	0.5	1.0	1.5	2.0	2.5	2.0	1.5	1.0	0.5	0			

EF - EQUIPMENT FAILURE
NC - NOT CALCULABLE
FROM DATA RECORDED NR - NOT RECORDED SYMBOLS: LVDT LOCATIONS GAGE LOCATIONS °0∾ 0.0787 inch > Grain Size > 0.0165 inch HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 3/1/66 1.74- gm/cm 1.74+ gni/cm<sup>3</sup> BEFORE LOADING 4-5/16 inches 4-9/32 inches \_ DATE Graded Ottawa Sand 25 (Sheet 1 of 2) SAMPLE DENSITY: BEFORE LOADING \_\_ AFTER LOADING... AFTER LOADING \_\_ 2 inches SAMPLE DESCRIPTION SAMPLE HEIGHT: TEST NUMBER \_ REMARKS:

	K. AS	KT ASURED AXIAL LOADS	LLOADS		DISPLACEMENTS	TS	RADIA	RADIAL LOADS	IMBEDE	IMBEDDED GAGE LOADS	LOLOS
APPLIED LOAD	MOVING	FIXED	CYLINDER	LVDT NO. 1	LVDT NO. 2	LVDT NO. 3	TOP	воттом	NO. 1	NO. 2	NO. 3
	41	<u>_</u>	-4	in/0.001	in/0.001	in/0.001	psi	psi	psi	psi	psi
T	0	0	0	0	ο	0	0	0	0		
	446	336	117	8.5	8.5	8.8	2.9	3.3	2.1		
	892	756	292	12.9	11.4	12.1	6.8	6.3	5.3	-	
	1428	1176	409	15.3	13.5	14.1	10.8	9.1	8.1		
	1874	1512	526	17.0	14.2	15.4	14.4	11.4	11.1		
	2231	1764	643	17.9	15.0	16.3	16.8	12.9	13.3		
3.75	3570	2772	936	20.8	17.3	18.9	25.2	18.1	19.2		
	5087	3948	1345	23.7	18.6	20.8	34.1	23.2	26.2		
6.25	6909	4704	1579	25.2	19.5	22.3	38.9	26.3	29.8		
	7229	5544	1930	26.8	20.3	23.5	44.9	29.8	31.3		
8.75	8434	6552	2223	28.2	21.2	24.6	51.2	33.5	39.9		
	9639	7560	2574	29.7	22.1	25.8	57.8	37.4	45.2		
8,75	8434	6930	1872	29.2	20.8	25,4	53.6	35.4	42.1		
	7363	6425	1287	28.7	19.9	25.4	51.2	34.1	39.9		
6.25	6909	5880	702	27.8	19.5	24.6	48.0	31.5	36.9		
	5087	5208	351	27.3	18.6	24.2	44.1	29.3	33.4		
3.75	3481	4032	0	25.8	16.9	23.1	36.2	25.4	27.5		

EF - EQUIPMENT FAILURE
NC - NOT CALCULABLE
FROM DATA ROSORDED IMBEDDED GAGE LOADS NR - NOT RECORDED NO. 2 psi 17.5 22.2 20.8 8.0 NO. 1 13.3 psi SYMBOLS: BOTTOM 20.5 17.8 14.4 9.8 RADIAL LOADS 28.4 26.8 20.9 9.23.9 15.8 LVDT LOCATIONS GAGE LOCATIONS 40L psi 0,1 21.6 20.8 18.9 20.1 in/0.001 LVDT NO. 3 DISPLACEMENTS 13.8 11.9 LVDT MO. 2 in/0.001 15.8 15.1 HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING in/0.001 24.4 24.4 23.4 22.3 20.4 LVDT NO. 1 CYLINDER -409 -175 -292 ~468 -468 -292 MEASURED AXIAL LOADS 3108 2772 2268 1512 1008 BEFORE LOADING\_ FIXED 336 SAMPLE DENSITY: BEFORE LOADING AFTER LOADING AFTER LOADING\_ MOVING 2321 1963 1428 892 446 SAMPLE DESCRIPTION SAMPLE HEIGHT APPLIED LOAD REMARKS: K-15 2.5 2.0 1.5 0.5 1.0

3/1/66

DATE

TEST NUMBER 25(Sheet 2 of 2)

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psı

NC - NOT CALCULABLE FROM DATA RECORDED EF - EQUIPMENT FAILURE NR - NOT RECORDED SYMBOL S: LVDT LOCATIONS GAGE LOCATIONS SAMPLE DENSITY: BEFORE LOADING \_\_\_\_1.72 gm/cm $\frac{3}{1.72 \text{ gm/cm}^3}$ HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 4-15/32 inches 4-7/16 inches  $1.73~\mathrm{gm/cm}^3$ DATE Graded Ottawa Sand BEFORE LOADING\_ AFTER LOADING .\_\_ AFTER LOADING\_\_ 2-1/4 inches 26 SAMPLE DESCRIPTION TEST NUMBER \_\_ SAMPLE HEIGHT: REMARKS:\_\_

AGE LOADS		_	psi	-		<u> </u>												
IMBEDDED GAGE LOADS			_	+-	6													
LOADS	воттом	psi		0	0	0 5 10	0 5 10 17	0 5 10 17 20	0 5 10 17 20 28	0 5 10 17 20 28 24	0 5 10 17 20 28 24	0 5 10 17 20 28 24 22	0 5 10 17 20 28 24 22 20	0 5 10 17 20 28 24 22 20 17	0 5 10 17 20 28 24 22 17 12	0 5 10 17 20 28 22 20 17 12	0 10 10 20 28 22 20 17 4	0 10 17 20 22 22 20 17 4
RADIAL LOADS	TOP	psi		0	0 &	0 8 16	0 8 16 27	0 8 16 33	0 8 16 27 33 43	0 16 8 33 43 38	0 16 33 33 35 35 37	0 16 27 33 43 38 35 31	0 116 33 33 33 33 35 37 37	16 33 33 33 33 35 37 18 18	0 16 33 33 33 31 18 18	0 11 33 33 44 33 37 7 7	16 33 33 43 33 77 7 7	16 33 33 43 33 43 43 43 43 43 43 43 43 43
NTS	LVDT NO. 3	ın/0.001		0	0 &	0 8 11	0 8 11 15	0 8 11 15 16	0 8 11 15 16 19	0 11 15 16 19	0 11 15 16 19 18	0 11 15 16 19 19 17	0 11 15 16 19 17 16	0 11 15 16 19 17 15	0 11 15 16 19 17 15 15	0 11 15 19 17 15 15	0 11 15 16 17 17 15 15	0 11 15 16 17 17 15 15
DISPLACEMENTS	LVDT NO. 2	in/0.001		0	2	0 2 9	0 2 2 0	0 6 9 9 9	0 0 0 0 0	2 2 2 2 2 2	0 6 9 9 5 4	0 6 9 9 9 6 4 4	0 5 9 9 9 5 4 4 8	0 6 9 9 9 5 4 4 8 8	0 6 9 9 9 6 4 4 8 8 0	0 5 9 9 9 5 4 4 8 8 0	0 6 9 9 6 4 4 8 6 0	0 6 9 9 6 4 4 8 8 0
	LVDT NO. 1	in/0.031		0	20	0 20 27	20 27 34	20 27 34 35	20 27 34 35	20 27 34 35 38	0 20 34 35 39 37	20 27 34 35 39 37 37	20 27 34 35 37 37 37	20 24 35 35 37 37 37 37	20 20 34 33 35 37 37 37 37	20 34 35 37 37 37 37 37 37	20 24 34 35 37 37 37 37 37 37	20 20 34 33 34 34 35 36 37 37
IL LOADS	CYLINDER	16		0	0 122	0 122 245	0 122 245 245	0 122 245 245 674	0 122 245 245 674 980	0 122 245 245 674 980 368	0 122 245 245 674 980 368 184	0 122 245 245 674 980 368 184	0 122 245 245 674 980 368 184 0	0 122 245 245 674 980 368 184 0 -184	0 122 245 245 674 980 368 184 -184	122 245 245 245 674 980 368 184 -245 -184	122 245 245 245 674 980 368 184 -184	122 245 245 245 674 980 368 184 -245 -184
MEASURED AXIAL LOADS	FIXED	<u>a</u>		0	331	331 662	331 662 1241	331 662 1241 1655	331 662 1241 1655 2565	331 662 1241 1655 2565	331 662 1241 1655 2565 2069 1820	331 662 1241 1655 2565 2069 1820 1490	331 662 1241 1655 2565 2069 1820 1490	331 662 1241 1655 2565 2069 1820 1490 1158	331 662 1241 1655 2565 2069 1820 1490 1490 1158 662	331 662 1241 1655 2565 2069 1820 1490 1158 662	331 662 1241 1655 2565 2069 1820 1490 1158 662 166	331 662 1241 1655 2565 2069 1820 1490 1158 662 166
MEA	MOVING	15		0	452	452 996	0 452 996 1538	452 452 996 1538 2353	452 452 996 1538 2353 3620	452 452 996 1538 2353 3620 2353	452 452 996 1538 2353 3620 2353 1900	452 996 1538 2353 3620 2353 1900	452 996 1538 2353 3620 2353 1900 1358	452 996 1538 2353 3620 2353 1900 1358 905	452 996 1538 2353 3620 2353 1900 1358 905 362	452 996 1538 2353 3620 2353 1900 1358 905 362	452 452 996 1538 2353 3620 2353 1900 1358 905 362	452 452 996 1538 2353 3620 2353 1900 1358 905 362
	APPLIED LOAD	K-lb		0	<b>0</b> 0.5	0 0.5 1.0	0 0.5 1.0 1.5	0 0.5 1.0 1.5 2.5	0 0.5 1.0 1.5 2.5 3.75	0 0.5 1.0 1.5 2.5 3.75	0 0.5 1.0 2.5 3.75 2.0	0 0.5 1.0 2.5 3.75 2.0	0 0.5 1.0 2.5 2.5 2.0	0 0.5 1.0 2.5 2.5 2.0 1.6	0 0.5 1.0 2.5 2.5 2.0 1.6 0.5	0 0.5 1.0 2.5 2.5 2.0 0.5	0 0.5 1.5 2.5 2.5 0.5 0.5	0 0.5 1.5 2.5 2.5 0.5 0.5

Sand Ridge 0.0787 inch > Grain Size > 0.0165 inch DATE \_\_ Graded Ottawa Sand 27 (Sheet 1 of 2) SAMPLE DESCRIPTION rest NUMBER .\_\_

SAMPLE DENSITY: BEFORE LOADING ...  $1.74~\mathrm{gm/cm}^3$ 

ENSITY: BEFORE LOADING  $\frac{1.13 \, \mathrm{gm/cm}^3}{1.75 \, \mathrm{gm/cm}^3}$ 

SAMPLE HEIGHT: BEFORE LOADING 4-5/16 inches
AFTER LOADING 4-9/32 inches

 REMARKS. Ridge of sand noted after test on surface in position shown by Atlantic Research LVDT Location Sketch.

L'VDT LOCATIONS

GAGE LOCATIONS

SYMBOLS:

NR - NOT RECORDED

EF - EQUIPMENT FAILURE

NC - NOT CALCULABLE

FROM DATA RECORDED

		MEA	MEASURED AXIAL LOADS	IL LOADS		DISPLACEMENTS	<b>1TS</b>	RADIA	RADIAL LOADS	IMBEDI	IMBEDDED GAGE LOADS	LOADS
	APPLIED LOAD	MOVING	FIXED	CYLINDER	LVDT NO. 1	LVDT NO. 2	L.VDT NO. 3	TOP	воттом	NO. 1	NO. 2	NO. 3
	K-15	16	9	91	in/0.001	in/0.001	in/0,001	psi	psį	psi	isd	psi
<u> </u>	0	0	0	0	0	0	0	0	0	0		
	0.5	402	179	226	8	1.7	4	0	-0.1	0.2		
	1.0	305	478	406	13.8	3.8	8.1	2.4	8.0	9.0		
	1.5	1368	807	565	17.0	5.3	10.2	4.2	1.8	8.0		
	2.0	1868	1196	678	19.7	6.7	12.1	6.3	2.8	1.2		
	2.5	2334	1435	846	21.5	7.8	13.8	8.4	3.9	1.5		
	3.75	3542	2317	1188	25.1	10.3	16.9	14.2	7.0	2.4		
	5.0	4830	3199	1581	27.9	12.5	20.0	20.1	10.2	3,5		
	6.25	6182	4036	1920	30.6	14.3	21.7	25.6	13.1	4.6		
	7.5	7342	4963	2260	32.3	15.8	23.9	31.5	16.2	5.8	-	
<del></del>	8.75	8694	5830	2600	34.1	17.4	25.5	37.8	19.0	7.0		
	10.0	9918	6728	2940	35.6	18.7	27.0	44.1	22.0	8 8 8		
	8.75	8694	6167	2315	35,1	18.2	26.5	38.4	19.9	7.5		
	7.5	7583	5741	1718	34.8	18.2	26.0	35.2	18.4	8.9		
	6.25	6229	5247	1018	34.1	17.5	25.4	31.5	16.3	0.9		
	5.0	5055	4560	452	33.6	16.3	24.6	27.8	14.2	5.4		
	3.75	3735	3797	0	32.6	15.6	23.2	. 2.4	11.7	4.4		

NO. 3 IMBEDDED GAGE LOADS NC - NOT CALCULABLE FROM DATA RECORDED EF - EQUIPMENT FAILURE NR - NOT RECORDED NO. 2 ρš NO. 1 2.7 2.1 1.5 0.5 psi 3.1 SYMBOLS: BOTTOM RADIAL LOADS Ď\$ 14.9 12.9 16.8 11.0 9.4 LVDT LOCATIONS GAGE LOCATIONS TOP psi °0√ 20.9 19.7 18.4 16.9 12,6 L.VDT NO. 3 in/0.001 DISPLACEMENTS 13.5 LVDT NO. 2 in/0.001 14.5 12.8 11.9 11.1 HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING DATE 3/2/66 17.9 31.0 30.4 1.75T NO. 1 29.2 27.8 25,3 CYLINDER -452 -282 -406 -452 -361 -113 MEASURED AXIAL LOADS TEST NUMBER 27 (Sheet 2 of 2) 1644 2990 2332 1046 224 2093 FIXED PISTON 16 BEFORE LOADING\_ SAMPLE DENSITY: BEFORE LOADING AFTER LOADING\_ AFTER LOADING\_ 2415 644 80 2093 1610 1046 MOVING SAMPLE DESCRIPTION SAMPLE HEIGHT: APPLIED LOAD REMARKS: **K**-15 2.5 2.0 5. 1.0 0.5

LVDT LOCATIONS 0 HEIGHT OF IMBEDDED GAGES ABOVE NOTTOM OF SAMPLE BEFORE LOADING ... DATE 3/3/66 AFTER LOADING 4-1/32 inches  $1.65 \text{ gm/cm}^3$ BEFORE LOADING 4-3/32 inches SAMPLE DENSITY: BEFORE LOADING.  $1.63~\mathrm{gm/cm}^3$ 0.0787" > Grain Size > 0.0165" SAMPLE DESCRIPTION Graded Offawa Sand AFTER LOADING \_\_\_ TEST NUMBER 28 (Sheet 1 of 2) 2 inches SAMPLE HEIGHT

SYMBOLS:

NR - NOT RECORDED

EF - EQUIPMENT FAILURE

NC - NOT CALCULABLE

FROM DATA RECORDED

GAGE LOCATIONS

REMARKS:

		_					-								*****				
LOADS	NO. 3	psi																	
IMBEDDED GAGE LOADS	NO. 2	psi													77.0				
IMBED	NO. 1	psi	0	9.0	1.2	2.1	3.0	4.1	9.9	9.3	12.2	14.9	17.8	20.5	19.0	17.5	15.6	13.5	
RADIAL LOADS	ваттом	psí	0	3.5	6.2	8.8	11.2	13.1	18.4	23.6	28.4	32.2	36.8	41.3	38.1	35.4	32.2	29.3	
RADI	TOP	psi	0	7.1	12.1	17.1	21.3	24.8	33.6	41.7	48.8	56.4	63.0	68.2	64.3	63.0	56.7	51.2	
175	LVDT NO. 3	in/0.001	0	14.1	17.4	20.5	22.3	23.9	26.9	30.3	32,3	33.8	35.4	36.9	36.4	36.4	35.9	34.8	
DISPLACEMENTS	L VDT NO. 2	in/0.001	0	14.0	17.6	19.9	22.1	23.8	27.2	7.6%	32.1	33.9	35.7	36.9	36.3	35.7	35.7	35.1	
	LVDT NO. 1	in/0.001	0	19.0	25.2	29.2	32.1	34.3	38.7	42.4	44.6	47.5	48.7	49.9	51.1	50.4	49.7	48.9	
L LOADS	("YLINDER	2	0	120	240	360	480	600	006	1200	1500	1740	2100	2400	1860	1140	099	240	
MEASURED AXIAL LOADS	FIXED	2	0	325	650	875	1381	1788	2681	3656	4631	5525	6581	7434	6825	6216	5769	4794	
MEA	MOVING	91	0	510	935	1360	1870	2380	3570	4845	6120	7395	8670	10030	84	7522	6120	4930	
	APPLIED LOAD	K-15	0	0.5	1.0	1.5	2.0	2.5	3.75	5.0	6.25	7.5	8.75	10.0	8.75	7.5	6.25	5.0	

NC - NOT CALCULABLE FROM DATA RECORDED EF - EQUIPMENT FAILURE NR - NOT RECORDED SYMBOLS: L VDT LOCATIONS GAGE LOCATIONS HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING \_ DATE 3/3/66TEST NUMBER 28 (Sheet 2 of 2) SAMPLE HEIGHT: BEFORE LOADING ... SAMPLE DENSITY: BEFORE LOADING ... AFTER LOADING\_ AFTER LOADING\_ SAMPLE DESCRIPTION REMARKS:\_\_

	7		7							 -0-00-100-0	 	 		 
LOADS	NO. 3	psi								-				
IMBEDDED GAGE LOADS	NO. 2	psi												
IMBED	NO. 1	) S.C.	11.7	9.0	7.7	6.5	4.8	3.2	1.2					
RADIAL LOADS	воттом	psi	24.9	19.3	17.2	14.2	10.9	7.1	2.3		_			
RADI	TOP	psi	44.9	36.2	31.5	26.4	23.6	14.7	5.5					
17.5	L VDT NO. 3	in/0.001	34.3	32.8	32.3	31.8	30.8	29.8	28.3					
DISPLACEMENTS	LVDT NO. 2	in/0.001	33.3	32.7	31.5	30.3	30.0	29.4	26.4		Topographic			
	LVDT NO. 1	in/0.001	47.5	45.3	45.3	43.8	42.4	40.9	37.3					
L LOADS	CYLINDER	16	-120	-360	-420	-420	-420	-360	-120					
MEASURED AXIAL LOADS	FIXED PISTON	91	3981	2925	2438	1869	1381	812	244				<del></del>	
MEA	MOVING	16	3655	2380	1870	1445	935	510	0					
0004	LOAD	K-16	3.75	2.5	2.0	1.5	1.0	0.5	0	<b>→ → → → → → → → → →</b>		•		
			1.5	51				*********			 	 		 

EF - EQUIPMENT FALLURE
NC - NOT CALCULABLE
FROM DATA RECORDED NR - NOT RECORDED SYMBOLS: LVDT LOCATIONS HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 3/8/66 3-15/16 inches 0.0787" > Grain Size > 0.0165" 1.68 gm/cm<sup>3</sup> 1.66 gm/cm<sup>3</sup> 4 inches Graded Ottawa Sand SAMPLE DENSITY: BEFORE LOADING \_\_ AFTER LOADING \_\_ SAMPLE HEIGHT: BEFORE LOADING AFTER LOADING\_ 29 2 inches SAMPL E DESCRIPTION TEST NUMBER .\_ REMARKS:\_

GAGE LOCATIONS

		T				·····									 8
LOADS	NO. 3	psi													 F390-1-68
IMBEDDED GAGE LOADS	NO. 2	psi													
IMBEDI	NO. 1	psi	0	1.0	1.9	2.9	3.9	4.9	4.4	3.9	3.2	2.3	ວ.ຄ		
RADIAL LOADS	воттом	psi	0	3.2	9.9	9.6	13.0	15.1	14.3	12.9	10.2	9.9	3.7		
RADIA	TOP	psi	0	3.7	6.1	8.1	9.7	11.6	10.7	10.2	9.4	7.9	1.6		
(TS	LVDT NO. 3	in/0.001	0	5.4	7.2	8.3	9.3	10.2	10.1	8.6	9.3	9.1	8.8		
DISPLACEMENTS	LVDT NO. 2	in/0.001	0	5.9	7.8	9.0	10.2	11.1	10.5	10.1	9.7	9.0	9.0		
	LVDT NO. 1	in/0.001	0	19.6	26.6	31.0	34.3	36.8	35.9	35.4	34.3	32.1	26.3		
L LOADS	CYLINDER	ιδ	0	133	228	342	475	570	418	190	38	-48	-38		
MEASURED AXIAL LOADS	FIXED	lb	0	300	099	066	1410	1725	1575	1350	066	00.9	30		
MEA	MOVING	16	0	416	968	1360	1888	2320	2048	1568	1056	576	64		
	APPLIED LOAD	<b>K</b> -15	0	0.5	1.0	1.5	2.0	2.5	2.0	1.5	1.0	0.5	0		

3/10/66\_ DATE \_ 4-1/32 inches SAMPLE DENSITY: BEFORE LOADING  $1.60~\mathrm{gm/cm}^3$ 1.63 gm/cm<sup>3</sup> 0.0787" > Grain Size > 0.0165" Graded Ottawa Sand 30 (Sheet 1 of 2) AFTER LOADING \_\_\_ SAMPLE DESCRIPTION TEST NUMBER

LVDT LOCATIONS

EF - EQUIPMENT FAILURE
NC - NOT CALCULABLE
FROM DATA RECORDED NR - NOT RECORDED SYMBOLS:

SAMPLE HEIGHT: BEFORE LOADING 4-1/32 inches	LVDT LOCATIONS	O Z
	(	
HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING		
2 inches	70	
Accidental application of 12.8 K-lb load at end of test.		
NEW XING.	GAGE LOCATIONS	

											_							
1_OADS	NO. 3	psi																
IMBEDDED GAGE 1.0ADS	NO. 2	psi																
IMBED	NO.	psi	0	1.2	2.7	4.3	5.9	7.3	11.6	15.6	19.3	23.5	27.1	31.4	33.0	31.1	28.4	25.1
RADIAL LOADS	воттом	psi	0	3.1	9.9	10.1	12.9	15.8	22.4	28.5	34.6	41.2	45.2	50.9	52.5	49.7	44.6	39.9
RADI	T0.	psi	0	3.0	5.2	7.4	9.1	10.6	14.8	18.5	22.4	27.1	30	34.2	34.6	32.1	29.8	25.6
ıTS	LVDT NO. 3	in/0,001	0	14.3	21.2	27.0	27.6	29.4	34.7	37.6	39.8	43.6	45.4	47.6	49.4	48.8	48.5	47.7
DISPL ACEMENTS	LVDT NO. 2	in/0.001	0	0.9	8.0	9.5	11.6	13.0	16.6	19.7	22.5	25.4	27.2	29.2	31.5	31.0	29.4	28.9
	LVDT NO. 1	in/0.001	0	16.8	27.8	31.2	34.5	36.8	43.1	46.2	49.1	51.9	53.3	55.5	59.4	58.4	58.4	55.5
LOADS	CYLINDER	16	0	59	238	404	524	999	1000	1357	1666	1975	2321	2618	1368	1142	476	1.19
MEASURED AXIAL LOADS	FIXED	1p	0	248	620	930	1302	1643	2558	3472	4340	5332	6138	7091	7285	6626	5812	4960
MEA	MOVING	91	0	340	748	1275	1700	2142	3400	4675	5780	7140	8262	9588	8432	7293	6120	4760
	APPLIED	K-16	0	0.5	1.0	1.5	2.0	2.5	3.75	5.0	6.25	7.5	8.75	10.0	8.75	7.5	6.25	5.0

NC - NOT CALCULABLE FROM DATA RECORDED IMBEDDED GAGE LOADS ž EF - EQUIPMENT FAILURE NR - NOT RECORDED NO. 2 Š NO. 15.6 12.6 7.0 psi 15.1 10.1 SYMBOLS: SOTTOM RADIAL LOADS psi 23.6 19.6 14.4 8.4 0.9 63.7 34.1 27,1 LVDT LOCATIONS GAGE LOCATIONS TOP psi 17.7 16.8 14.9 12.9 11.3 6.3 <u>о</u>гч LV5T NO. 3 in/0.031 44.7 43.2 42.2 40.6 38.2 50.9 45.1 DISPLACEMENTS LVDT NO. 2 in/0.001 26.0 25.5 24.6 26.3 24.7 33.9 25.3 HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 3/10/66 LVDT NO. 1 in/0.001 49.7 54.8 52.9 52.3 51,4 47.6 43.8 O DATE CYLINDER -333 -416 -428 -416 -309 -190 3380 MEASURED AXIAL LOADS TEST NUMBER 30 (Sheet 2 of 2) FIXED PISTON Ib SAMPLE DENSITY: BEFORE LOADING. BEFORE LOADING 2480 3782 2883 1953 1395 852 155 9300 AFTER LOADING. AFTER LOADING. MOVING PISTON 1b 3570 2295 1785 1360 765 340 2580 -68 SAMPLE DESCRIPTION SAMPLE HEIGHT: APPLIED Load 3.75 REMARKS: ×. 2.5 2.0 1.5 1.0 0.5

psi

NR - NOT RECORDED

EF - EQUIPMENT FAILURE

NC - NOT CALCULABLE

FROM DATA RECORDED SYMBOLS: LVDT LOCATIONS REMARKS. As load was being relaxed, load was decreased from HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING \_ DATE \_\_3/11/66 SAMPLE DENSITY: BEFORE LOADING  $1.74-\ \mathrm{gm/cm}^3$  after Loading  $1.74+\ \mathrm{gm/cm}^3$ 3-63/64 inches 0.0787" > Grain Size > 0.0165" SAMPLE DESCRIPTION Graded Ottawa Sand BEFORE LOADING 4 inches TEST NUMBER 31 (Sheet 1 of 2) AFTER LOADING. 2 inches 8.75 to SAMPLE HEIGHT:

	E LOADS	NO. 3	psč					-											
	IMBEDDED GAGE LOADS	NO. 2	psi	0	1.8	3.6	5.4	8.9	8.4	11.0	13.4	16.5	18.1	20.4	23.8	21.8	18.1	18.4	المستحدد والم
	IMBED	NO. 1	psi	0	1.1	2.5	3.8	4.9	6.2	9.3	12.2	14.9	18.2	8.02	24.1	22.1	18.2	18.7	
	RADIAL LOADS	воттом	psi	0	3.3	7.2	10.8	13.3	16.0	22.3	28.0	33.6	37.5	42.0	48.1	44.6	37.5	38.8	
GAGE LOCATIONS	RADI	TOP	psi	0	0.9	11.6	16.8	19.7	23.5	31.5	37.8	44.1	50.4	55.9	60.4	53.0	50.4	50.7	
CAGE L	175	L VDT NO. 3	in/0.001	0	3.0	5.1	6.9	8.1	9.1	11.4	13.4	15.0	16.3	18.0	19.7	18.9	17.4	18	
	DISPLACEMENTS	LVDT NO. 2	in/0.001	0	3.6	5.8	7.7	9.2	9.7	11.5	13.4	14.8	15,6	16.9	18.6	17.5	16.8	17.4	
K-lb.		LVDT NO. 1	in/0.001	0	9.9	11.1	14.4	16.5	18.3	21.8	24.2	26.3	27.8	29.4	31.8	31.4	29.2	29.4	
to 5.49 K-lb instead of 7.5 K-	r Lokns	CYLINDER	16	0	164	234	10	468	585	889	1170	1427	1696	2048	2153	1755	410	644	-
K-lb inst	MEASURED AXIAL LOARS	FIXED	lЬ	0	316	632	1027	1422	1833	2844	3792	4708	5688	9899	7821	6920	5372	5625	
75 to 5.49 F	MEAS	MOVING	1.6	0	422	930	1426	1927	2366	3634	4816	6135	7504	8670	10140	8488	2999	6084	
-1				i															

2.0 2.5 3.75 5.0 6.25 7.5 8.75

1.5

APPLIED LOAD

Х-15

LVDT LOCATIONS GAGE LOCATIONS HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING \_ DATE 3/11/66 SAMPLE DENSITY: BEFORE LOADING\_ TEST NUMBER 31 (Sheet 2 of 2) SAMPLE HEIGHT: REFORE LOADING\_ AFTER LOADING\_ AFTER LOADING\_ SAMPLE DESCRIPTION \_\_ REMARKS.\_

SYMBOLS:

\$ 25

EF - EQUIPMENT FAILURE
NC - NOT CALCULABLE
FROM DATA RECORDED NR - NOT RECORDED

L		MEA	MEASURED AXIAL LOADS	T LOADS		DISPLACEMEN 7S	.75	RADI	RADIAL LOADS	IMBED	IMBEDDED GAGE LOADS	LOADS
	APPLIED LOAD K-16	MOVING PISTON 16	FIXED PISTCN 16	CY1.1NDER 1b	LVDT NO. 1 in/0.001	LVDT NO. 2 in/0.051	L.VDT AO. 3 in/6.001	TOP	BOTTOM psi	NO. 3	NO. 2 psi	NO.
J	5.0	5002	5024	176	29.0	16.5	17.4	49.1	36.8	17.0	17.0	
156	3.75	3718	4108	-234	27.7	15.6	16.2	45.0	32.0	14.7	16.0	
	20	2451	3128	-458	25.8	14.3	14.1	39.1	27.6	11.8	12.9	

LOADS	NO. 3	psi							·		
IMBEDDED GAGE LOADS	NO. 2	psi	17.0	16.0	12.9	12.4	10.7	9.2	9.9	3.0	
IMBED	NO	psi	17.0	14.7	11.8	10.6	9.1	7.4	5,3	2.6	
RADIAL LOADS	воттом	psi	36.8	32.0	27.6	25.2	22.1	18,4	13.1	0.0	
RADI	TOP	psi	49.1	45.0	39.1	36.1	32.0	27.1	20.5	10.5	
.75	LVDT MO. 3	in/0.001	17.4	16.2	14.1	14.4	13.9	13.1	12.1	10.8	
DISPLACEMEN 7S	LVDT NO. 2	170.071	16.5	15.6	14.3	13.6	12.8	11.9	11.1	9.5	
	LVDT NO. 1	in/0.001	29.0	27.7	25.8	25.3	23.7	20.8	20.3	16.3	
L LOADS	CYI, INDER	lb	176	-234	-458	-491	-515	-445	-445	-257	
MEASURED AXIAL LOADS	FIXED	16	5024	4108	3128	2623	2180	1580	1011	316	
MEA	MOVING	16	5002	3718	2451	1960	1521	980	422	0	
i i	LOAD	K-1b	5.0	3.75	2.5	2.0	1.5	1.0	0.5	0	

DATE 3/15/66 SAMPLE DESCRIPTION Graded Ottawa Sand See Remarks TEST NUMBER 32 (Sheet 1 of 2)

SAMPLE DENSITY: BEFORE LOADING 1.65; 1.30; 1.67 gm/cm

BEFORE LOADING 1/2"; 2-49/64"; 55/64; by layers APTER LOADING NC SAMPLE HEIGHT:

AFTER LOADING NR

HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING \_

Ottawa < 0.0165"; 55/64" loose Ottawa > 0.0165" on top.

LVDT LOCATIONS 0 REMARKS. 1/2" dense Ottawa > 0.0165" on bottom; 2-49/64" fine

GAGE LOCATIONS

SYMBOLS:

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NR - NOT RECORDED EF - EQUIPMENT FAILURE

NC - NOT CALCULABLE FROM DATA RECORDED

LOADS	NO. 3	psi														-		
IMBEDDED GAGE LOADS	NO. 2	psi	सञ														표.	
IMBED	۲0. ٦	psi	0	2.5	5.4	8.1	10.8	13.3	19.4	25.3	32.1	38.4	44.2	48.9	46.4	42.9	39.1	
RADIAL LOADS	воттом	psi	स				· · · · · · · · · · · · · · · · · · ·					<u> </u>	·				सस	
RADIA	TOP	psi	0	3.1	5.0	7.4	8.6	10.2	14.7	17.8	23.2	25.6	28.2	30.1	28.2	25.4	23.5	
ITS	LVDT NO. 3	in/9.601	0	43.6	66,4	83.4	93.7	103.2	121.2	133.8	149.5	154,5	166.6	170.4	169.7	169.7	169.7	
DISPLACEMENTS	LVDT NO. 2	in/0.001	0	23.7	40.7	51.0	59.7	67.5	84.3	95.5	107.3	114.2	122.5	127.7	127.5	126.6	125.6	
	LVDT NO. 1	in/0.561	0	46.8	73.1	92.0	103.6	114.2	136.4	148,5	164.6	175.3	187.7	192.8	1.92.1	191.4	191.1	
LOADS	CYLINDER	16	0	178	238	476	595	714	1011	1428	1921	2118	2380	2737	2190	1630	1071	
MEASURED AXIAL LOADS	FIXED	91	0	250	593	936	1248	1810	2527	3432	4446	5304	6318	7160	6786	6131	5569	
MEAS	MOVING	<b>-P</b>	0	406	879	1352	1774	2298	3617	4732	6084	7402	8619	10072	1668	49.09	6591	¥
	APPLIED LOAD	K-lb	0	0.5	1.0	<b>i.5</b>	2.0	2.5	3,75	5.0	6.25	7.5	8.75	10.0	8.75	7.5	6.25	

IMBEDDED GAGE LOADS HOT CALCULABLE FROM DATA RECORDED EF . EQUIPMENT FAILURE NR - NOT RECORDED NO. 2 psi 田田 EF , , , psi 34.6 28.9 22.6 15.8 19.3 11.4 SYMBOLS: BOTTOM RADIAL LOADS ps. EF 五五 LVDT LOCATIONS GAGE LOCATIONS ďOL psi ကင္င 20.5 17.3 16.1 15.8 13.4 10.8 003 LVDT NO. 3 in, 0.001 161.6 168.2 166.6 146.4 163.6 169.2 164.1 159.1 DISPLACEMENTS LVDT NO. 2 in/0.001 124.6 124.6 123.5 122.5 122.5 122.5 121.4 122.5 HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING in/0.001 LVDT NO. 1 189.9 188.5 187.0 182.6 186.3 175.3 163.1 DATE CYLINDER 130 595 262 214 143 ₽ MEASURED AXIAL LOADS TEST NUMBER 32 (Sheet 2 of 2) FIXED PISTON SAMPLE DENSITY: BEFORE LOADING SEFORE LOADING 4680 3744 2730 1872 1342 842 156 2231 AFTER LOADING. AFTER LOADING 2 MOVING 84 919 3988 2704 2197 1724 1268 SAMPLE DESCRIPTION SAMPLE HEIGHT: APPLIED Load REMARKS: **₹**-16 5.0 3.75 2 :53 2.0 1.5 1.0 0.5

3/12/66

NO. 3

psi

TEST NUMBER 33 (Sheet 1 of 2)

SAMPLE DESCRIPTION Ottawa Sand, see remarks

SAMPLE HEIGHT BEFORE LOADING 3-7/8 inches total

AFTER LOADING 3-11/16 inches total
HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING
2 inches

REMARKS: 21/32" layer at 1.7 density on bottom; 2-5/8" of 1.35 density in middle; 19/32" of 1.7 density on top.

GAGE LOCATIONS

 $\begin{array}{c} 1 \\ 2 \\ 0 \\ 0 \\ 0 \\ 1 \end{array}$ 

SYMBOL 5:

NR - NOT RECORDED

EF - EQUIPMENT FAILURE

NC - NOT CALCULABLE FROM DATA RECORDED

LOADS	NO. 3	psi																
IMBEDDED GAGE LOADS	NO. 2	psi	0	2.4	4.6	6.8	9.0	11.0	16.5	21.7	27.3	32.5	37.6	42.4	39.9	37.0	33.4	29.3
IMBED	NO. 1	psi	0	3.0	6.1	9.5	12.7	15.9	23.9	31.6	39.4	47.5	54.5	61.1	58.0	53.4	48.2	41.8
RADIAL LOADS	воттом	psi	0	2.6	4.7	6.9	Ω	10.5	14.2	18.1	21.8	25.4	28.5	31.5	29.8	27.6	24.8	21.7
RADIA	T0F	psi	0	4.2	7.1	9.4	11.6	13.6	17.8	22.2	26.5	30.4	34.0	37.3	35.2	32.1	29.4	25.6
т.	LVDT NO. 3	in/0.001	0	34.3	51.8	64.4	71.7	79.2	93.0	102.5	112.1	119.7	126.2	132,3	132.3	131,3	131,3	130.3
DISPLACEMENTS	LVDT NO. 2	in/0.001	0	18.0	27.7	36.1	42.2	48.4	62.3	72.1	82.0	88.9	2.96	102.4	9.66	99.0	98.3	97.6
	LVDT NO. 1	in/0.001	0	38.5	59.2	73.5	83.3	93.4	109.1	121.8	133.4	139.3	148.1	154.9	154.9	154.4	153.9	152.9
L LOADS	CYLINDER	lb	0	176	235	446	517	682	975	1292	1622	1880	2232	2526	2056	1434	940	517
MEASURED AXIAL LOADS	FIXED	15	0	316	632	980	1343	1738	2623	3508	4487	5435	6541	7394	6992	6304	5467	4561
MEA:	MOVING P:STON	lb	0	414	897	1380	1811	2329	3554	4796	6141	7348	8720	3867	6806	7762	6452	5141
	LOAD	K-1b	0	0.5	1.0	1.5	2.0	2.5	3.75	5.0	6.25	7.5	8.75	10.0	8.75	7.5	6.25	5.0
				15														

F390-1-68 <mark>ზ</mark>. ა psi IMBEDDED GAGE LOADS HC - NOT CALCULABLE FROM DATA RECORDED EF - EQUIPMENT FAILURE HR . NOT RECORDED NO. 2 2.4 19.9 17.8 15.5 12.9 9.7 24.8 NO. 1 .sd 19.0 13,9 27.2 23.7 SYMBOLS: BOTTOM 13.9 7.6 18.6 15.2 12.2 10.1 RADIAL LOADS , se GAGE LOCATIONS L VDT LOCATIONS 13.9 18.9 17.8 16.6 15.6 psi 001 125.8 122.3 120.2 126.2 125.1 LVDT NO. 3 in/0.001 129.1 113.1 DISPLACEMENTS 95.5 95.2 92.7 94.1 LVDT NO. 2 in/0.001 94.1 HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 147.6 149.0 141.2 LVDT NO. 1 in/0.001 148.1 146.1 145.1 CYLINDER -24 -212 -235 -212 -118 -141 MEASURED AXIAL LOADS FIXED PISTON Ib BEFORE LOADING AFTER LOADING\_ SAMPLE DENSITY: BEFORE LOADING AFTER LOADING\_ 2449 1895 158 1343 853 3729 2781 HOVING 1035 3795 2674 2070 1622 604 SAMPLE DESCRIPTION SAMPLE HEIGHT: APPLIED LOAD 3.75 ₹ 5 REMARKS: 2.5 2.0 1.5 1.0 0.5

ſ,

DATE

33 (Sheet 2 of 2)

TEST NUMBER

SAMPLE DESCRIPTION Ottawa Sand, see remarks DATE 34 (Sheet 1 of 2) TEST NUMBER

SAMPLE DENSITY: BEFORE LOADING GE remarks
AFTER LOADING

SAMPLE HEIGHT: BEFORE LOADING 3-15/16 inches total

AFTER LOADING 3-25/32 inches total

REMARKS. 9/16" layer at 1.61 density on bottom; 2-17/32" layer of fine < 0.0165" grain at 1.38 density in middle; 27/32" layer at 1.64 density on top.

GAGE LOCATIONS

 $\begin{array}{c} 1 & 2 \\ 0 & 3 \\ 0 & 0 \\ 0 & 2 \\ 0 & 1 \\ \end{array}$ 

SYMBOLS:

MR - NOT RECORDED

EF - EQUIPMENT FAILURE

NC - NOT CALCULABLE FROM DATA RECORDED

_	<del></del>		_		_	-													
LOADS	NO. 3	) s O																	
IMBEDDED GAGE LOADS	NO. 2	psi	0	2.2	4.7	7.3	10.0	12.6	20.0	27.0	34.2	40.2	48.2	55.2	51.9	48.2	43.1	38.1	
IMBE	NO. 1	psi	0	1.4	3.1	5.1	6.8	6°8	13.2	18.1	22.6	27.1	32.3	36.7	34.3	32.1	28.7	25.1	
RADIAL LOADS	воттом	psi	0	6.0	1.9	3.0	3.9	4.9	7.4	6.7	12,5	15.0	18.1	21.0	19.3	17.2	14.8	13.0	
RADI	TOP	psi	0	1.3	2.1	3.0	3.7	4.4	8.9	9.4	12.6	15.8	18.6	22.1	19.4	17.0	14.7	12.3	
NTS	LVDT NO. 3	in/0.001	0	34.1	50.0	9.09	67.7	74.7	87.0	97.6	104.0	112.0	119.7	124.2	123.2	123.2	122.7	122.2	
DISPLACEMENTS	LVDT NO. 2	in/0.001	0	22.1	33.5	42.4	49.1	55.4	0.89	77.8	86.9	94.1	101.0	105.9	104.5	103.8	103.8	103.8	
	LVDT NO. 1	in/0.001	0	40.e	59.7	81.1	81.8	89.3	103.6	115.9	126.6	135.4	142.2	147.1	147.1	146.1	146.1	145.1	
L LOADS	CYLINDER	16	0	180	288	480	009	744	1128	1440	1752	2136	2436	2820	2160	1500	984	528	
MEASURED AXIAL LOADS	FIXED PISTON	16	0	278	587	927	1236	1576	2503	3399	4326	5222	6141	7092	6628	6026	5253	4542	
MEA	MOVING	1 <sub>b</sub>	0	406	879	1352	1774	2332	3515	4732	6033	7149	8518	2996	8720	7436	6185	4969	
APPLIED	LOAD	<b>K</b> -16	0	0.5	1.0	1.5	2.0	2.5	3.75	5.0	6.25	7.5	8.75	10.0	8.75	7.5	6.25	5.0	

EF - EQUIPMENT FAILURE
NC - NOT CALCULABLE
FROM DATA RECORDED MR - NOT RECORDED SYMBOLS: LVDT LOCATIONS DATE  $\frac{3}{21/67}$ TEST NUMBER 34 (Sheet 2 of 2) SAMPLE DENSITY: BEFORE LOADING \_\_\_ SAMPLE HEIGHT: BEFORE LOADING \_\_ AFTER LOADING\_ AFTER LOADING\_ SAMPLE DESCRIPTION \_\_

			-GADS	NO. 3	psí								
			IMBEDDED GAGE LGADS	NO. 2	psi	33.7	26.5	23.7	20.7	16.7	12.5	5.4	
			IMBED	NO. 1	psi	21.8	17.0	15.1	12.8	10.1	6.3	1.2	
			RADIAL LOADS	воттом	psi	11.1	8.9	7.9	6.9	5.8	4.6	2.7	
		GAGE LOCATIONS	RADIA	TOP	psi	10.5	8.4	7.7	6.9	6.3	6.4	5.2	
<u> </u>	ノ ー	C76E L	175	LVDT HO. 3	in/0.001	121.2	120.2	119.5	118.4	115.2	114.5	109.8	
DH.C			DISPLACEMENTS	LVDT NO. 2	in/0.001	102.1	101.0	99.0	98.6	98.3	98.3	97.6	
BEFORE LOADING				LVDT NO. 1	in/0.001	145.1	143.7	141.2	139.3	137.3	136.4	126.6	
HEIGHT OF IMBEDDED GAGES ABOVI; BOTTOM OF SAMPLE			L LOADS	CYLINGER	lb	216	09-	-144	-240	-264	-240	-180	
S ABOVE BO			MEASURED AXIAL LOADS	FIXED	16	3677	2750	2240	1854	1313	865	247	
EDDED GAGE			MEA	MOVING	16	3786	2535	2028	1555	1014	575	30	
HEIGHT OF IMB	REMARKS:			APPLIED	K-15	3.75	2.5	2.0	1.5	1.0	0.5	0	

SYMBOLS: LVDT LOCATIONS GAGE LOCATIONS 01 HEIGHT OF IMBEDDED GAGES ABOVE NOTTOM OF SAMPLE BEFORE LOADING 3/24/66gm/cm<sup>3</sup> AFTER LOADING 3-15/16 inches SAMPLE HEIGHT: BEFORE LOADING 4-1/16 inches 0.0787" > Grain Size > 0.0165" SAMPLE DESCRIPTION Graded Ottawa Sand Tetlon powder lubricant used. SAMPLE DENSITY: BEFORE LOADING 1.71 AFTER LOADING NC TEST NUMBER 35 (Sheet 1 of 2) 2 inches REMARKS:\_

NR - NOT RECORDED

EF - EQUIPMENT FAILURE

NC - NOT CALCULABLE

FROM DATA RECORDED

	MEA	MEASURED AXIAL LOADS	L LOADS		DISPLACEMENTS	475	RADIA	RADIAL LOADS	IMBED	IMBEDDED GAGE LCADS	LCADS
LOAD	MOVING PISTON	FIXED	CYLINDER	LVDT NO. 1	L.VDT NO. 2	LVDT NO. 3	T0.P	воттом	NO. 1	NO. 2	NO. 3
K-1b	lb	qI	16	in/0.001	in/0.001	in/0.001	psi	18 C	psi	psi	psi
0	0	0	0	0	0	0	0	0	O	0	
0.5	424	325	0	29.8	20.6	24.5	0.2	0.7	5.2	4.9	
1.0	953	812	49	41.5	29.3	34.1	0.1	6.0	10.2	10.1	
1.5	1377	1300	61	49.7	36.3	41.2	0	1.1	14.4	15.0	
2.0	1800	1706	122	56.0	40.1	45.8	0.2	1.4	18.2	20.0	
2,5	2330	2210	182	59.7	44.3	49.2	0.3	1.7	21.7	24.3	
3.75	3530	3331	219	69.2	52.6	57.1	1.2	2.4	30.7	36.0	
5.0	4501	4306	243	74.3	55.0	61.9	1.6	2.9	36.4	44.2	
6.25	5913	5752	304	81.6	61.2	67.2	2.6	4.0	46.7	56.0	
7.5	7307	6947	413	87.3	66.2	72.1	3.7	4.9	54.2	67.8	****
8.75	8525	8288	462	0.06	9.69	76.4	5.1	5.8	62.4	9.77	
10.0	9884	9620	510	94.8	73.2	79.2	6.3	6.9	70.4	86.8	
8.75	8684	8580	243	94.1	72.1	79.2	6.2	6,2	64.1	80.2	
7.5	7466	7434	182	92.5	71.6	77.8	6.0	5.7	59.0	73.6	
6.25	6072	6208	49	92.0	70.1	8.97	6.1	5.3	52.7	65.8	
5.0	4977	5168	0	9.06	68.5	76.3	5.5	5.2	45.8	57.4	· · · · · ·

EF - EQUIPMENT FAILURE
NC - NOT CALCULABLE
FROM DATA RICORDED SYMBOLS: LVDT LOCATIONS HEIGHT OF IMBEDDED CAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING ... 3/24/86 TEST NUMBER 35 (Sheet 2 of 2) SAMPLE DENSITY: BEFORE LOADING ... SAMPLE HEIGHT: BEFORE LOADING \_\_ AFTER L'SADING AFTER LOADING\_ SAMPLE DESCRIPTION \_\_ REMARKS:\_

NR - NOT RECORDED

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GAGE LOCATIONS

		_									 	 	 
LOADS	NO. 3	isd											
IMBEDDED GAGE LOADS	NO. 2	į	47.9	37.1	32.9	27.1	21.3	14.3	8.0				
IMBED	NO. 1	psi	38.8	30.2	25.8	21.9	16.9	11.4	4.7				
RADIAL LOADS	воттом	pai	4.6	3.9	3.5	3.1	2.8	2.5	1.3				
RADIA	TOP	psi	5.2	4.2	3.7	3.2	2.8	2.3	1.7				
TS	LVDT NO. 3	in/0.001	75.8	75.2	74.2	72.2	71.7	71.2	69.4				
DISPLACEMENTS	LVDT NO. 2	in/0.001	67.5	67.0	65.9	63.8	63.1	61.8	61.2	<del></del>			
	LVDT NO. 1	in/0.001	88.9	87.7	87.2	86.7	85.2	83.3	80.4				
LOADS	CYLINDER	15	-24	-36	-49	-36	-36	-36	-24				
MEASURED AXIAL LOADS	FIXED	91	3900	2632	2178	1625	1138	618	86				
MEAS	MOVING	92	3706	2436	1942	1412	953	424	0				
	APPLIED LOAD	K-1b	3.75	2.5	2.0	1.5	1.0	0,5	0				
L			<del></del>	164							 	 	 _

3/29/67 DATE \_\_\_ SAMPLE HEIGHT: BEFORE LOADING -3-15/16 inches SAMPLE DENSITY: BEFORE LOADING  $1.70~\mathrm{gm/cm}^3$ 0.0787" > Grain Size > 0.0165" Graded Ottawa Sand NC TEST NUMBER 36 (Sheet 1 of 2) AFTER LOADING \_\_\_ SAMPLE DESCRIPTION

SYMBOLS:

NC - NOT CALCULABLE FROM DATA RECORDED EF - EQUIPMENT FAILURE NR - NOT RECORDED

LVDT LOCATIONS

HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING

REMARKS. Teflon powder lubricant used

2 inches

AFTER LOADING 3-13/16 inches

GAGE LOCATIONS

OADS	NO. 3	psi			,	., .		<u> </u>	<del></del>					<del></del>			
IMBEDDED GAGE LOADS	NG. 2	psí	0	2.5	5.8	6.6	13.4	17.8	28.2	38.8	50.6	58.5	6.89	78.9	73.1	64.4	_
IMBEDI	NO.	psí	0	4.4	9.0	13.5	17.4	21.4	30.5	39.1	47.4	55.2	63.2	70.4	66.2	61.5	0
RADIAL LOADS	воттом	psi	0	0.3	0.4	9.0	0.7	0.0	1.4	2.1	رن ت	3.3	4.0	4.6	4.0	3.6	33
RADIA	TGP	psi	0	0	0	0	0	0	0	0.3	0.7	1.1	1.8	2.1	2.1	2.2	2.5
115	LVDT NO. 3	in/0.001	0	18.4	28.8	36.4	42.0	45.4	54.3	9.09	2′99	71.2	75.0	77.3	77.3	8.5%	76.3
DISPLACEMENTS	LVDT NC. 2	in/0.001	0	14.6	22.9	29.3	34.5	39.0	46.0	52.2	57.1	61.8	64.9	68.0	0.79	65.9	64.9
	LVDT HO. 1	in/0.001	0	26.9	40.3	48.7	54.1	58.9	67.7	73.1	79.1	82.8	9.88	92.0	90.3	89.3	89.3
LOADS	CYLINDER	lb	0	118	176	212	235	235	294	400	435	470	494	588	423	235	164
MEASURED AXIAL LOADS	FIXED PISTON	<u>a</u>	0	342	808	1244	1788	2239	3421	4665	5956	7184	8444	9703	8770	7588	6483
MEA	MOVING PISTON	91	0	424	932	1356	1864	2373	3644	2000	6272	7695	8797	10170	9017	7661	6483
721 120	LOAD	K-1b	0	0.5	1.0	1.5	2.0	2.5	3.75	5.0	6.25	7.5	8.75	10.0	8.75	2.5	6.25

SYMBOLS: LVDT LOCATIONS GAGE LOCATIONS HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING ... DATE  $\frac{3/29/67}{}$ TEST NUMBER 36 (Sheet 2 of 2) SAMPLE DENSITY: BEFORE LOADING \_\_ SAMPLE HEIGHT: BEFORE LOADING \_\_ AFTER LOADING\_ AFTER LOADING\_ SAMPLE DESCRIPTION \_\_ REMARKS:\_\_

NR - NOT RECORDED

EF - EQUIPMENT FAILURE

NC - NOT CALCULABLE

FROM DATA RECORDED

		7					<del></del>				4,345	]8
LOADS	NO. 3	psi	******	·						·		F390-1-68
IMBEDDED GAGE LOADS	NO. 2	psi	49.7	39.4	28.6	24.8	19.7	15.7	9.9	6.4		
IMBED	NO. 1	isd	50.0	43.0	35.2	31.8	27.6	22.9	14.3	3.7		
RADIAL LOAES	воттом	psi	2.8	2.3	1.6	1.3	1.0	0.7	0.4	0,1		
RADIA	ТОР	isd	2.2	2.1	2.1	2.0	2.0	1.6	1.6	1.6		
(TS	LVDT NO. 3	in/0.001	75.2	74.5	72.7	71.7	71.2	70.2	68.2	66.2		
DISPLACEMENTS	LVDT NO. 2	in/0.001	63.8	62.3	61.0	60.7	59.0	57.6	57.1	57.1		
	LVDT NO. 1	in/0.001	88.6	88.0	9.98	83.8	83.3	82.8	79.9	75.5		
L LOADS	CYLINDER	શુ	59	0	0	-24	-24	-47	-59	-59		
MEASURED AXIAL LOADS	FIXED	9	5256	3965	3942	2208	1773	1275	653	249		
MEA	MOVING PISTON	15	51.70	3814	2627	2034	1559	1085	508	34		
6	LOAD	K-lb	5.0	3.75	2.5	2,0	1,5	1.0	0.5	0		

LVDT LOCATIONS 0 3/31/66 Load applied smoothly rather than in steps. DATE \_\_\_ SAMPLE HEIGHT: BEFORE LOADING 3-31/32 inches AFTER LOADING 3-15/16 inches 1.81 gm/cm<sup>3</sup> SAMPLE DENSITY: BEFORE LOADING  $1.78~\mathrm{gm/cm}^3$ 0.0787" > Grain Size > 0.0165" SAMPLE DESCRIPTION Graded Ottawa Sand AFTER LOADING .\_\_ 37 TEST NUMBER .. REMARKS:\_

SYMBOLS:

EF - EQUIPMENT FAILURE NR - NOT RECORDED

NC - NOT CALCULABLE FROM DATA RECORDED

GAGE LOCATIONS

LOADS	NO. 3	psi										
IMBEDDED GAGE LOADS	NO. 2	isc	0	3.3	7.5	12.2	16.2	14.0	11.1	7.7	2.8	
IMBED	NO. 1	psi	0	3.4	7.2	11.4	15.7	13.2	10.5	7.0	0.4	
RADIAL LOADS	Воттом	psi	0	10.5	21.9	33,4	43.8	38.8	31.8	22.8	1.4	
RADIA	401	psi	0	16.3	24.9	33.6	42.0	40.4	36.2	29.4	12.1	
TS	L.VDT NO. 3	in/0.001	0	22.0	27.3	30.6	33.3	33.0	31.2	29.5	21.2	
DISPLACEMENTS	LVDT NO. 2	in/0.001	0	15.6	20.2	23.8	8.92	26.5	25.5	23.5	20.1	
	LVDT NO. 1	in/0.001	0	38.9	48.8	50.0	54.1	52.9	51.7	49.1	38.3	
L LOADS	CYLINDER	16	0	774	1547	2285	2927	1571	357	-405	-238	
MEASURED AXIAL LOADS	FIXED	9	C	1576	3244	5145	6952	5701	4635	3013	464	
WEA:	MOVING	15	C	2373	4746	7458	2966	8678	4949	2542	170	
	APPLIED	X-1b	c	20 50	5.0	7.5	10.0	7.5	5.0	2.5	0	
				67								

SYMBOLS: inch > Grain Size > 0.0165 inch  $1.70 \text{ gm/cm}^3$ Graded Ottawa Sand

0.0787 inch > Grain ~. SAMPLE DENSITY: BEFORE LOADING. SAMPLE DESCRIPTION TEST NUMBER \_\_\_ SAME HEIG REX

שר בני עבוזטו ו	AFER DENSITE DELONE ECADINO -			\0	CHCACCHA FCZ 4 GZ
	AFTER LOADING	NC			EF - EQUIPMENT FAILURG
WPLE HEIGHT:	BEFORE LOADING	4 inches		LVDT LOCATIONS	NC - NOT CALCULABLE
	AFTER LOADING	NR		6	CACA ALAC MORT
EIGHT OF IMB	EIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING	TTOM OF SAMPLE	BEFORE LOADING	0,	
	2 inches				
EM &RKS:	Load applied smoothly rather t	othly rather	than in step. Teflon		
	powder lubricant used	used.		GAGE LOCATIONS	
	MEASURED AXIAL LOADS	L LOADS	DISPLACEMENTS	RADIAL LOADS	IMBEDDED GAGE LOADS

												<b>~</b>
LOADS	N.). 3	psi										F390-1-38
IMBEDDED GAGE LOADS	NO. 2	psí	0	24.7	46.3	68.4	89.4	77.3	61.0	39.4	10.5	
IMBED	NO. 1	psi	0	32.4	56.1	压死	6.3	83.7	67.7	45.8	9.0~	
RADIAL LOADS	воттом	psi	0	5.0	EF	स्र	18.4	15.4	12.5	8.2	2.1	
RADIA	TOP	psi	0	4.7	7.4	11.2	15.2	13.4	11.3	8.0	3.4	
T.S	L.VDT NO. 3	in/0.001	0	46.0	62.9	73.5	84.8	83.4	9.08	77.8	69.3	
DISPLACEMENTS	LVDT NO. 2	in/0.001	0	37.6	52.4	62.3	72.1	71.6	69.0	67.0	64.4	
	L.VDT NO. 1	in/0.001	0	66.1	83.2		104.0	102.3	101.6	98.2	88.0	
L LOADS	CYLINDER	91	0	209	290	418	464	209	23	-46	-58	
MEASURED AXIAL LOADS	FIXED		0	2384	4647	7152	9596	7569	5245	2742	293	
MEAS	MOVING	9	0	2369	4771	7304	9836	7567	2067	2468	99	
	APPLIED LOAD	K-1b	0	2.5	5.0	7.5	10.0	7.5	5.0	2.5	0	

LVDT LOCATIONS 20/2 01 0.0787 inch > Grain Size > 0.0165 inch HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 4/5/66 4-3/32 inches 39(Sheet 1 of 2) DATE 3-7/8 inches 1.58 gm/cm<sup>3</sup> Graded Ottawa Sand Teflon powder lubricant used. NC SAMPLE DENSITY: BEFORE LOADING \_\_\_ SAMPLE HEIGHT BEFORE LOADING \_\_ AFTER LOADING. AFTER LOADING \_\_ 2 inches SAMPLE DESCRIPTION TEST NUMBER REMARKS:\_

GAGE LOCATIONS

SYMBOLS:

NR - NOT RECORDED

EF - EQUIPMENT FAILURE

NC - NOT CALCULABLE

FROM DATA RECORDED

	MEA	MEASURED AXIAL LOADS	L LOADS		DISPLACEMENTS	175	RADIA	RADIAL LOADS	IMBED	IMBEDDED GAGE LOADS	LOADS
APPLIED LOAD	MOVING	FIXED	CYLINDER	LVDT NO. 1	LVDT NO. 2	L.VDT NG. 3	TOP	воттом	NO. 1	NO. 2	NO. 3
χ-lb	16	91	9	in/0.001	in/0.001	in/0.001	psį	psı	, sd	psi	psi
0	0	0	0	0	0	0	0	0	0	<u>۔۔۔</u>	
0	425	373	47	42.7	20.3	33.9	0.7	1.2	2.4	5.1	
1.0	918	809	116	64.5	30.5	51.5	1.8	2.7	4.8	9.5	
1.5	1360	1244	163	81.3	40.8	64.6	1.7	3.5	7.4	14.7	
2.0	1802	1711	210	90.1	48.4	71.7	1.3	4.2	8.6	18.9	
2.5	2312	2177	233	99.5	57.1	77.8	1.3	4.8	12.6	23.7	
3.75	3485	3323	256	114.2	64.9	91.9	1.8	6.5	19.3	35.0	
5.0	4760	4416	408	125.6	67.5	0.66	1.9	8.1	25.7	44.7	
6.25	6052	5600	466	134.4	78.2	106.1	2.4	10.2	32.0	56.2	
7.5	7395	8669	524	139.3	84.4	111.1	3.2	12.0	39.1	68.4	
8.75	8568	7977	676	146.1	0.06	118.2	3.7	13.5	44.5	77.8	
10.0	9724	9330	722	149.0	94.1	121.2	4.5	15.7	51.0	87.6	
8.75	8840	8584	485	149.0	91.7	121.2	4.3	13.8	48.0	84.4	
7.5	7395	7114	396	148.5	92.0	120.2	4.5	12.9	42.1	74.2	
6.25	6273	6204	210	148.1	9.06	120.2	5.2	12.2	39.1	69.4	
5.0	4964	4976	70	147.1	90.0	119.2	5.5	11.2	32.9	61.9	

F390-1-68

54.5

27.4

0 0

118.1

88.6

146.1

0

3810

NO. 3 IMBEDDED GAGE LOADS NC - NOT CALCULABLE FROM DATA RECORDED psi EF - EQUIPMENT FAILURE NR - NOT RECORDED NO. 2 39.7 34.7 28.2 21,8 p & . ж . 13.8 17.1 20.3 9.6 4.8 SYMBOLS: 80TT0₩ RADIAL LOADS 7.4 ດ ນີ້. 3.9 LVDT LOCATIONS GAGE LOCATIONS 0.0 4.0 9.0 ت ئ 5.1 TOP .isa 115.2 114.5 113.8 112.4 113.2 in/0.001 DISPLACEMENTS LVDT NO. 2 In/0.001 86,5 86.5 84.8 84.4 83.7 HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 143.2 LVDT NO. 1 in/0.001 132.5 144.2 139.3 137.3 145.1 CYLINDER -47 -47 -58 -58 -23 -23 MEASURED AXIAL LOADS FIXED PISTON 16 BEFORE LOADING SAMPLE DENSITY: BEFORE LOADING 622 2644 2146 1213 187 AFTER LOADING\_ 1711 AFTER LOADING. MOVING PISTON 16 2465 1530 2006 1020 578 102 SAMPLE DESCRIPTION SAMPLE HEIGHT: APPLIED LOAD REMARKS: **₹**-15 2.0 2.5 1.0

DATE 4/5/36

TEST NUMBER 39(Sheet 2 of 2)

SYMBOLS: LVDT LOCATIONS GAGE LOCATIONS Graded Ottawa Sand 0.0787 inch > Grain Size > 0.0165 inch HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING \_ 4/1/66 3-29/32 inches DATE \_ SAMPLE DENSITY: BEFORE LOADING  $1.56~\mathrm{gm/cm}^3$ 4-1/8 inches Teflon powder lubricant used NC 40(Sheet 1 of 2) SAMPLE HEIGHT: BEFORE LOADING \_\_\_ AFTER LOADING\_\_\_ AFTER LOADING\_ 2 inches SAMPLE DESCRIPTION TEST NUMBER REMARK S:\_

NR - NOT RECORDED EF - EQUIPMENT FAILURE

NC - NOT CALCULABLE FROM DATA RECORDED

APP1 1ED	HEA	MEASURED AXIAL LOADS	L LOADS		DISPLACEMENTS	4TS	RADI,	RADIAL LOADS	MBED	IMBEDDED GAGE LOADS	LOADS
LOAD	MOVING PISTON	FIXED PISTON	CYLINDER	LVDT NO. 1	LVDT NO. 2	· VDT	T0P	воттом	NO. 1	NO. 2	NO. 3
K-tb	16	2	lb	in/0.001	in/0.001	in/ 0.00 1	psi	n d.	, sd	psi	. <u>.</u>
0	0	0	0	0	0	0	0	0	0		
0.5	442	375	7.1	49.7	29.7	39.4	0.8	1.0	3.2	EF	
1.0	952	844	167	73.1	46.4	57.6	1.1	1.9	6.8	-	
1.5	1360	1250	214	89.3	57.6	7.07	٠ <u>.</u>	2.8	10.3		
2.0	1938	1750	238	100.2	66.4	77.8	1.8	3.7	13.8		7.24
2.5	2380	2188	262	107.7	73.2	85.6	2.1	4.5	17.2		
3.75	3655	3375	405	123.7	83.0	97.0	3.2	6.9	25.6		
5.0	4998	4609	476	134.4	92.7	107.1	5.0	9.7	33.9		
6.25	6120	5703	524	143.2	102.1	114.0	6.3	11.4	42.1		
7.5	7497	6938	686	147.1	103.8	121.2	7.8	13.8	50.3		-
8.75	8670	9808	714	155.8	111.1	128.3	9.4	16.3	58.6		•
10.0	10200	93.75	857	158.8	114.2	131.3	10.9	18.9	9.99		<del></del>
8.75	8840	8281	643	158,3	114.2	131,3	8.6	16.9	62.4		
7.5	7599	7312	452	157.8	114.2	131.3	9.3	15.4	57.7		
6.25	6324	6141.	286	158.8	114.2	131.3	8.4	13.3	52.4		
5.0	5100	5062	214	156.8	114.2	131.3	7.5	12.2	47.9		***************************************
3.75	3740	3828	95	155.8	112.1	129.3	6.3	10.3	41.4	- 타	
						A	7				٦

NC - NOT CALCULABLE FROM DATA RECORDED IMBEDDED GAGE LOADS EF . EQUIPMENT FAILURE NR - NOT RECORDED NO. 2 ps. 田田 EF NO. 1 ps. 32.9 24.4 29.1 19.3 12.5 SYMBOLS: BOTTOM RADIAL LOADS .<u>.</u> 8.0 7.0 5.9 LVDT LOCATIONS GAGE LCCATIONS 4.5 psi ON 124.2 122.2 128.3 126,2 123.2 120.2 in/0.001 DISPLACEMENTS in/0.001 114.4 112.8 112.4 110.7 110.7 110.7 LV9T NO. 2 HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING DATE  $\frac{4}{7}/66$ in/0.061 153.9 152.9 149.0 144.2 154.4 147.1 CYLINDER MEASURED AXIAL LOADS 24 -24 -24 -24 0 0 FIXED PISTON SAMPLE HEIGHT: BEFORE LOADING SAMPLE DENSITY: BEFORE LOADING. 1219 219 2750 AFTER LOADING\_ 2188 1656 6.56 AFTER LOADING\_ MOVING 2550 1462 1020 595 2040 SAMPLE DESCRIPTION TEST NUMBER APPLIED Load REMARKS:\_ K-16 2.5 2.0

NO. 3 ps

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40(Sheet 2 of 2)

SYMBOLS: LVDT LOCATIONS 0.078 inch > Grain Size >-0.0165 inch HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 4-1/16 inches 4-5/64 inches Teflon powder lubricant not used DATE \_\_  $1.79~\mathrm{gm/cm}^{3}$ Graded Ottawa Sand S 41 (Sheet 1 of 2) SAMPLE HEIGHT: BEFORE LOADING \_\_\_\_ SAMPLE DENSITY: BEFORE LOADING .... AFTER LOADING \_\_ AFTER LOADING\_ 2 inches SAMPLE DESCRIPTION TEST NUMBER .\_\_ REMARKS:\_

GAGE LOCATIONS

EF - EQUIPMENT FAILURE
NC - NOT CALCULABLE
FROM DATA RECORDED MR - NOT RECORDED

						-													
LOADS	NO. 3	psi	0	1.2	2.4	3.6	4.8	9	6	11.5	14.1	16.4	19	21.5	20.1	19.1	17.8	15.3	
IMBEDDED GAGE LOADS	NO. 2	psi	0	3.1	5.4	7.7	14.3	五五										सञ	
IMBED	NO. 1	psi	Ö	0.5	8.0	1,3	1.9	2.5	3.9	5.5	7.2	8.7	10.4	12.3	11.4	10.4	6.3	8.1	
RADIAL LOADS	воттом	psi	0	4.8	8.4	12.2	14.9	17.5	24.2	29.6	34.9	39.9	44.6	49.9	48,3	44,4	41.7	37.3	
RADIA	TOP	psi		6.4	10.4	13.5	15.9	18.9	24.6	30.2	36.3	42.5	47.9	53.6	50.4	47.2	44.1	39.7	
rs	L.VDT NO. 3	in/0.001	0	11	14.8	17	18.6	20.1	23.4	25.8	27.8	28.7	30.3	31.9	31.7	30.7	30.3	29.9	
DISPLACEMENTS	LVDT NO. 2	in/0.001	0	3.8	4.9	5.7	6.4	7.1	9.8	10.1	11.2	12.1	13,3	14	13.8	13.7	13.4	12.6	
	LVDT NO. 1	in/0.C01	0	12.9	17.5	20.6	22.5	24.4	27.8	30.6	32.7	34.8	35.9	38	37.7	37.1	35.9	35.4	
LOADS	CYLINDER	92	0	218	315	508	678	799	1210	1525	1936	2238	2602	2904	2226	1500	992	460	
MEASURED AXIAL LOADS	FIXED	<u>.a</u>	0	316	632	948	1296	1643	2550	3476	4424	5214	61.62	71.10	9639	61.15	5546	47.72	
MEA:	MOVING	4	0	448	996	1449	1932	2415	3726	4916	6244	7400	8746	10005	8970	7590	6348	5141	
	APPLIED	K-15	0	0.5	1.0	1.5	2.0	2.5	3.75	5.0	6.25	7.5	8.75	10.0	8.75	7.5	6.25	5.0	

\_ DATE \_\_ SAMPLE DESCRIPTION \_\_

T RECORDED JIPMENT FAILURE T CALCULABLE DM DATA RECORDED

		200	SYMBOLS
SAMPLE DENSITY:	CAMPLE DENSITY: BEFORE LOADING	3	a LOX - ax
	AFTER LOADING		EF - EQUIP
SAMPLE HEIGHT:	BEFORE LOADING	LVDT LOCATIONS	NC - NOT C
	AFTER LOADING		
HEIGHT OF IMBED	HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING		
		<u> </u>	
REWARKS:		GAGE LOCATIONS	

		-т								ω
LOADS	NO. 3	psi	13.7	11.4	10.1	8.6	<b>6.</b> 8	4.7	2.3	F390-1-68
IMBEDDED GAGE LOADS	NO. 2	psi	표.						म्	
IMBEDI	NO. 1	psi	6.8	5.4	4.5	3.7	2.9	1.9	0.7	
RADIAL LOADS	воттом	psi	32.4	27.1	24.5	21.4	18	13	8.9	
RADIA	T0P	psi	34.9	29.4	26.8	23.9	20.8	16.8	8.4	
ıTS	LVDT NO. 3	in/0.001	28.8	27.9	27.1	26.3	25.2	25.5	21.2	
DISPLACEMENTS	LVDT NO. 2	in/0.001	12.1	11.4	11.2	10.6	10.4	9.9	9,4	
	LVDT NO. 1	in/0.001	34.5	32.7	31.6	31	29.4	27.5	25.1	
L LOADS	CYLINDER	9	0	-315	-460	-484	-484	-460	-266	
MEASURED AXIAL LOADS	FIXED	9	3950	3065	2623	2212	1643	1027	909	
MEA:	MOVING	91	3795	2501	2036	1552	1035	586	104	
	APPLIED Lr n	dl-r	3.75	2.5	2.0	1.5	1.0	0.5	0	

Sand Ridge Size > 0.0165 inch 4/13/66 DATE \_ 0.0787 inch > Gra: Dense Ottawa Sand 42(Sheet 1 of 2) SAMPLE DESCRIPTION TEST NUMBER

SAMPLE DENSITY: BEFORE LOADING 1.79 gm/cm AFTER LOADING. 4-3/16 inches BEFORE LOADING SAMPLE HEIGHT:

HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 4-5/32 inches AFTER LOADING ...

Very slight ridge of sand (after load applied) around 2 inches REMARKS:\_

edge of sample between LVDT locations No. 1 and

GAGE LOCATIONS တ္ထ

SYMBOLS:

NC - NOT CALCULABLE FROM DATA RECORDED EF - EQUIPMENT FAILURE NR - NOT RECORDED

LVDT LOCATIONS

_	1		_																	_
LOADS	NO. 3		- C	, C	۲ - ۲	1 0	. c.	) e	י ני	7.6	α σ	2.0	14.2	17.2	ι α α	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	19.6	16.0	10.9 8.7	:
IMBEDDED GAGE LOADS	NO. 2		0	9.4	7 7	9 9	2 4	5	14.1	17.7	20.8	24.4	27.4	30.7	29.1	9.7.4	95 5		20.5	) 
IMBED	NO. 1	. <u>.</u>	0	C LC		1.7	2.5	3.1		8.9	8	10.3	12	13.9	12.9	12.3		0	8.6	
RADIAL LOADS	воттом	, a	0	6.2	9.11	17	21	24.7	33.6	41	47.6	53.9	59.1	66.7	63	60.9	57.2	2. 2.	47.2	
RADI/	TOP	psi	0	9,4	16	22.4	26.9	32.1	43.5	51	59.2	68.8	75.6	83	79.3	77.2	73	68.2	59.8	-
TS	LVDT NO. 3	in/0.001	0	17.9	22	24.7	26.5	28.3	31.5	33.6	35.9	37.9	39,4	41.2	40.3	40	39.4	38.8	37.3	
DISPLACEMENTS	LVDT NO. 2	in/0.001	0	15.8	19.9	22.5	24.4	26	29.1	31.7	33.9	36.3	38.3	40	40	39.4	38.8	38.1	36.6	_
	LVDT NO. 1	in/0.001	0	25.1	31.2	35.1	37.3	38.6	43	45.3	47.7	49.2	51.6	53.6	53.6	52.6	52.1	50.4	49.7	7
L LOADS	CYLINDER	1b	0	186	291	489	909	669	1118	1445	1817	2120	2446	2796	2697	1421	885	373	-23	
MEASURED AXIAL LOADS	FIXED	2	0	311	622	964	1322	1679	2566	3452	4354	5287	6158	7114	6648	6158	5520	4727	4012	-
MEA	MOVING	16	0	422	946	1420	1927	2366	3718	4986	6152	7605	8856	10140	8923	7909	6540	5138	3955	
APPLIED	LOAD	K-15	0	0.5	1.0	1.5	2.0	2.5	3.75	5.0	6.25	7.5	8.75	10.0	8.75	7.5	6.25	5.0	3.75	

NC - NOT CALCULABLE FROM DATA RECORDED IMBEDDED GAGE LOADS 3.1 EF - EQUIPMENT FAILURE NR - NOT RECORDED 15.6 11.6 NO. 2 17.1 14.1 psi 6 NO. J ž. 8.9 6.2 5.2 4.1 SYMBOLS: BOTTOM RADIAL LOADS 33.9 36.8 32.2 26.6 19.4 9.6 p si LVDT LOCATIONS GAGE LOCATIONS 14.2 51.3 47.6 43.3 35.3 TOP PS. 26 007 L.VDT NO. 3 in/0.001 33.9 32.5 30.9 28.8 34.5 35 DISPLACEMENTS L VDT NO. 2 in/0.001 35.4 34.4 33.7 32.7 31.3 HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LUADING 4/13/66in/0.001 46.5 44.7 LVDT NO. 1 47.6 43.2 47 DATE CYLINDER -443 -489 -466 -433 -233 -396 MEASURED AXIAL LOADS SAMPLE DEMSITY, BEFORE LOADING \_ FIXED PISTON BEFORE LOADING TEST NUMBER 42(Sheet 2 of 2) 3032 404 2550 1586 1011. 2177 AFTER LOADING\_ AFTER LOADING\_ MOVING 1590 1082 642 270 2636 2096 SAMPLE DESCRIPTION SAMPLE HEIGHT: APPLIED LOAD **χ**-15 2.0 1.5 1.0 0.5 REMARKS: 2.5

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1.7

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AMPLE DESCRIPTION 0.0787 inch >	\ \ \ \	Offawa Sand Grain Size >	Offawa Sand Grain Size > 0.0165	inch moist soil	t soil		00				
0.0853	0.0853		gm/cm <sup>3</sup>		orl	1	N	SYM	SYMBOL S:		
						ور ا	m)			NOT RECORDED	
BEFORE LOADING 4-1/16 inch		1-1/16	inch	٠		LVDT	LVDT LOCATIONS		EF - EQUIP	EQUIPMENT FAILURE NOT CAI CIII ABI F	LURE
AFTER LOADING $2-11/16$ inch	- 1	3-11/16	inch			-	(			FROM DATA RECORDED	ORDED
EIGHT OF IMBEDDED GAGES ABOYE BOTTOM OF SAMPLE	ABOYE BOTTOM OF S	TTOM OF S	AMPLE	BEFORE LOADING	DING		02/				
2 inches						( §	- 6				
7/16 inch Dense Ottawa sand	sand	sand	q uo	on bottom sample; then		3-1/8	o l				
inch moist soil; finally 1/2 inch loose of sample; moisture content of soil =	귀	귀	nch loo of soil	OE	ttawa sand on top percent		GAGE LOCATIONS				
KEASURED AXIAL LOADS	URED AXIAL LOADS	יר רסעםs			DISPLACEMENTS	NTS	RADIA	RADIAL LOADS	IMBED	IMBEDDED GAGE LOADS	LOADS
MOVING FIXED CYLINDER PISTON		CYLIND	ER	LVDT NO. 1	LVDT NO. 2	LVDT NO. 3	TOP	воттом	NO. 1	NO. 2	NO. 3
16 16 tb		و		in/0.001	in/0.001	in/0.001	281	psi	p S.	psi	, s d
0 0		0		0	0	0	0	0	0	0	0
385 325 119		119		305	353	302	F-4	1.3	4.1	2.5	2.5
805 618 238		238		455	200	420	3.2	2.8	7.9	5.1	5.4
1330   975   405		405		550	585	909	ည	4.2	12	8.2	G
1785   1300   500		200		284	651	563	6.3	5.6	16.6	11.4	12.6
2345 1658 654		654		662	694	580	7.1	6.8	20.9	14.9	16.9
3570 2698 952		952		775	186	999	8.6	9.9	33.9	24.1	28.4
4830   3656   1190	<del>-,, -,</del>	1190		813	852	720	9.8	13.1	47.9	34	40.6
6169 4648 1476		1476		860	968	747	11.3	15.9	61.8	43.3	54.8
7350 5850 1761		1761	~	927	923	268	13.5	19.4	80.2	54.2	70.4
8452 6728 1975		1975		947	949	770	15	22.8	94.7	63.9	84.3
9870 7751 2261	·	2261	_	066	972	797	16.8	25.9	114.4	74.3	98.1
8750   6971   1856	~	1856		066	972	797	15.6	23.6	103.4	68.1	90.9
7700   6459   1380	·	1380	-	066	972	197	14.7	21.5	97	61.7	84.3
6125   5525   785		785		966	372	797	12.9	19.2	88.9	52.9	73.4
4988   4794   476		476		066	696	788	11.8	16.9	78.9	45.5	63.6
3815 3868 190		190		896	965	776	10.4	14.6	67.3	35.3	51.8

4/19/66

DATE \_\_\_

TEST NUMBER 43 (Sheet 1 of 2)

A Commence of the second secon

EF - EQUIPMENT FAILURE
NC - NOT CALCULABLE
FROM DATA RECORDED NR - NOT RECORDED SYMBOLS: LVDT LOCATIONS GAGE LOCATIONS HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING \_ 4/19/66 DATE \_\_ 43 (Sheet 2 of 2) SAMPLE DENSITY: BEFORE LOADING\_ SAMPLE HEIGHT. BEFORE LOADING\_ AFTER LOADING\_ AFTER LOADING... SAMPLE DESCRIPTION TEST NUMBER \_\_ REMARKS:\_

EASU	ASURED AXIAL L	 	.0 ADS		DISPLACEMENTS	1 1	RADIA	RADIAL LOADS	IMBEDI	IMBEDDED GAGE LUADS	L'AADS
NG FISED CYLINDER	CYLINDER	<del></del>	LVDT NO. 1		LVDT NO. 2	L VDT MO. 3	TOP	воттом	- OX	NO. 2	NO. 3
16 lb lb ir/0.001	i lb		ir/0.001		in 0.001	ir 0.001	psì	psi	psı	psı	psi
2520 276248 968	48		896		964	27.6	8.5	11.4	52.5	25.8	38.8
2030 2275 -119 968	-119		896		964	776	7.6	8.6	46	22.3	32.5
1488 1885 -214 968	-214		896		964	776	7	8.6	39.4	18.6	26.5
1050 1365 -238 968	-238		896		964	776	9	6.8	31.3	íS	19.6
525 812 -262 968	-262		896		ъ́96	911	4.2	4.7	21.3	10.5	12
0 292 -262 968	-262		896		362	170	2.1	2.8	7.6	3.2	2.8
Spil compressed 3/8 inch during placement of piston before test began.	essed 3/8 inch during placemer	8 inch during placemer	placemer	#	of piston	before test	began.				

SAMPLE DESCRIPTION 1/2 inch Ott wa Sand; top & hottom 0.0787 > Grain 0,0165 inch moist Aberdeen soil (0,0787 > Grains) inch 4/21/66 DATE 44 (Sheet 1 of 2) TEST NUMBER

SAMPLE DENSITY: BFFORE LOADING 1.77;0.858;202 gm/cm<sup>3</sup>

SAMPLE HEIGHT. BEFORE LOADING 4-1/16 inches overall AFTER LOADING 2-3/4 inches overall

HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING

2 inches

REMARKS: 9/16 inch dense Ottawa sand on bottom; then 2-7/8 inch

woist soil; 5/8 inch Ottawa on top moisture content of soil = 13 percent

 $\begin{pmatrix} 1 & O_2 \\ O & 3O \\ O & SO \end{pmatrix}$ 

SYMBOLS:

NR - NOT RECORDED

EF - EQUIPMENT FAILURE NC - NOT CALCULABLE FROM DATA RECORDED

LVDT LOCATIONS

			Т	_															
LOADS	NO. 3	P S.	0	2.9	0.9	9.6	13.5	17.5	29	41.8	56.3	67.2	83.6	94.4	87.9	80.5	69.8	58.5	46.4
IMBEDDED GAGE LOADS	NO. 2	psi	0	1.8	3.6	5.6	7.5	9.2	14,4	19.5	25.5	30.4	36.7	42.4	39.4	36.1	33	28.4	23.8
IMBED	NG. 1	30	0	4.3	8.5	13.4	17.9	23	37.3	52.5	72.1	89.4	110.5	126.2	118.4	110.5	101.8	88.4	75.2
RADIAL LOADS	воттом	p si	0	1.4	3.1	5.2	7.7	8.8	13.2	17	20.6	24.5	27	31	28.6	27.3	24.3	20.6	18,3
RADIA	TOP	psi	0	1.3	4.4	10.8	121	17.6	22.5	25.2	28	29.4	31,3	31.7	30	28.8	27.3	25	23.6
17.5	LVDT NO. 3	in/0.001	0	260	408	460	530	567	640	665	710	727	746	156	756	756	756	756	756
DISPLACEMENTS	LVDT NO. 2	in/0.001	0	318	444	523	580	633	714	167	817	847	879	901	901	901	899	899	668
	LVDT NO. 1	in/0.001	0	358	202	605	089	793	937	1000	Offscale								Offscale
L LOADS	CYLINDER	16	0	186	302	464	638	742	1114	1392	1740	2042	2320	2575	2204	1624	1114	650	278
MEASURED AXIAL LOADS	FIXED PISTON	16	0	286	604	922	1240	1590	2512	3.403	4.452	5235	6320	7.203	6630	9019	5.406	4532	3752
MEASURE	NOT SI P	16	0	422	915	1408	1830	2394	3608	4893	6336	73 92	0088	9356	8976	7762	6442	5104	3942
ABD1 5ED	LOAD	K-10	0	0.5	1.0	1.5	2.0	2.5	3.75	5.0	6.25	7.5	8.75	10.0	8.75	7,5	6.25	5.0	3.75

SYMBOLS: GAGE LOCATIONS LYDT LOCATIONS HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING DATE \_\_ TEST NUMBER 44 (Sheet 2 of 2) SAMPLE HEIGHT: BEFORE LOADING ... SAMPLE DENSITY: BEFORE LOADING. AFTER LOADING\_ AFTER LOADING\_ SAMPLE DESCRIPTION REMARKS:\_

EF - EQUIPMENT FAILURE NR - NOT RECORDED

NC - NOT CALCULABLE FROM DATA RECORVED

LOADS	MO. 3	D.W.C	31.2	24.9	18.3	10.3	2.5	-3.7		0000
IMBEDDED GAGE LOADS	NO. 2	psi	19.3	17.2	15.5	13.4	10.8	ຜ		
IMBEDI	NO. 1	psi	58.5	50.8	42.6	33.4	21.5	9.9		
RADIAL LOADS	воттом	psi	13.8	13	11.2	O	7.4	3.0		
RADIA	TOP	psi	21.3	20.6	19.7	18.6	15.8	6.3	began	
175	LVDT NG. 3	in/0.001	756	756	746	746	746	746	placement of piston before test	
DISPLACEMENTS	LVDT NG. 2	ו 00.0/אין	268	897	897	897	897	895	of piston	
	LVDT NO. 1	in/0.001	Offscale					Offscale	t placemen	
L LOADS	CYLIMDER	Ф	23	-46	-70	-186	-232	-209	Soil compressed 1/2 inch during	
MEASURED AXIAL LOADS	FIXED	9	2639	2258	1781	1304	874	286	essed 1/	
MEA	MOVING	18	2640	2112	1690	1126	634	70	oil compr	1
	APPLIED 1040	K-ib	2.5	2.0	1.5	1.0	0.5	0	Note:	•
•				100						

SYMBOLS: HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 5/2/66 0.0787 inch > Grain Size > 0.0165 inch 4-1/16 inches DATE \_\_ 1.65 gm/cm Graded Ottawa Sand Teflon powder lubricant used NC NR 45 (Sheet 1 of 2) SAMPLE DENSITY: BEFORE LOADING SAMPLE HEIGHT: BEFORE LOADING AFTER LOADING \_ AFTER LOADING\_ SAMPLE DESCRIPTION \_\_ 2 inches

TEST NUMBER

$\begin{pmatrix} 1 & 2 \\ 0 & 3 \\ 0 & 0 \\ 1 & 0 \end{pmatrix}$	CAGE LOCATIONS
1 2 0 0 CATIONS	$\begin{pmatrix} 1 & 0 \\ 0 & 3 \\ 0 & 3 \end{pmatrix}$

EF - EQUIPMENT FAILURE NR - NOT RECORDED

NC - NOT CALCULABLE FROM DATA RECORDED

LOADS	NO. 3	psi	0	6.5	12.4	17.7	22.6	27.9	58.9		64.2	75.2	87.8	100.6	93.1	83.6	73.6	63.3	53
IMBEDDED GAGE LOADS	NO. 2	psi	0	4.8	6	11.6	14.4	17.9	38.2		42.1	50.3	58.6	9.99	61.9	55.2	48.8	41.8	34.9
IMBED	NO. 1	psi	0	3.7	9.7	12.1	16.2	20.2	43.3		47.4	56.3	65	74.3	9.69	63.8	55.1	48.6	40.7
RADIAL LOADS	ВОТТОМ	psi	0	0.5	1.1	1.6	2.1	2.4	44.		4.4	5.1	5.0	8.8	6.2	5.6	5.2	4.8	4.4
RADIA	TOP	psi	0	0.9	1.4	2,1	2.6	2.6	2.9		2.1	2.4	2.8	3.2	3.2	ლ ლ	ლ. ლ.	3.7	4
1TS	LVDT NO. 3	in/0.001	0	20	30.9	38.6	44.5	48.5	65.6		70.2	74.2	77.3	80.8	80.8	80.8	79.8	79.3	77
DISPLACEMENTS	LVDT NO. 2	in/0.001	0	14.5	24.4	29.1	33.4	38.3	58.1		61.8	66.4	71.1	74.2	72.7	72.1	71.6	70.1	68.2
	LVDT NO. 1	in/0.001	0	28.6	44.4	57.5	64.5	70.6	92		95.4	100.9	104	109.1	108.4	107.7	101	105	1.03.6
LOADS	CYLINDER	ą.	0	47	70	163	175	186	291		350	396	443	489	350	233	163	20	23
MEASURED AXIAL LOADS	FIXED	ą	0	364	848	1273	1727	2151	5151		5818	8069	8181	9575	8635	7499	627.2	5075	3939
MEA.	MOVING	9-	0	420	924	1344	1764	2285	5376		5998	7157	8518	9878	8803	7560	6199	5006	3763
	APPLIED LOAD	K-15	0	0.5	1.0	1.5	2.0	2.5	5.57	5.0	6.25	7,5	8.75	10.0	8.75	7.5	6.25	5.0	3.75
				-								_							

REMARKS:\_\_

NC - NOT CALCULABLE FROM DATA RECORDED EP . EQUIPMENT FAILURE NR - NOT RECORDED SYMBOLS: LVDT LOCATIONS GAGE LCCATIONS HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING \_ 5/2/66 DATE ⟨Sheet 2 of 2) SAMPLE DENSITY: BEFORE LOADING ... SAMPLE HEIGHT: BEFORE LOADING\_ AFTER LOADING ... AFTER LOADING... SAMPI E DESCRIPTION TEST NUMBER REMARKS:\_

T i		-1	·			******			] a
LOADS	NO. 3	psi	40.2	35	28.4	22.1	14	5.3	00 1 2000
IMBEDDED GAGE LOADS	NO. 2	psi	27.1	23.7	21.3	19.1	16.6	11.3	
IMBED	NO. 1	psi	31.6	27.9	22.3	16.4	7.1	8.0	
RADIAL LOADS	воттом	psi	3.7	3.4	2.8	2.4	1,9	1,3	
RADI/	TOP	psi	4.1	4.1	4.3	4.7	5.5	4.4	
TTS	LVDT NO. 3	in/0.061	75.3	75.8	74.5	73.2	711.7	69,7	
DISPLACEMENTS	LVDT NO. 2	in/0.001	67	29	62.9	62.9	67	67.5	
	LVDT NO. 1	in/0.001	102.3	102.3	100.9	99.5	95.4	90.3	
r LOADS	CYLINDER	16	0	0	-23	-23	0	0	
MEASURED AXIAL LOADS	FIXED	9	2727	2197	1742	1212	636	242	
MEA	MOVING	-	2587	2016	1428	1042	571	168	· · · · · · · · · · · · · · · · · · ·
	APPLIED LOAD	ж-1b	2.5	2.0	1.5	1.0	0.5	0	

NR - NOT RECORDED SYMBOLS: GAGE LOCATIONS LVDT LOCATIONS HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 5/3/66 0.0787 > Grains > 0.0165 inch 4-5/16 inches 4-5/32 inches SAMPLE DENSITY: BEFORE LOADING  $1.67~\mathrm{gm/cm}^3$ DATE Graded Ottawa Sand Teflon powder lubricant used NC 46 (Sheet 1 of 2) BEFORE LOADING\_ AFTER L'CADING ... AFTER LOADING ---2 inches SAMPLE DESCRIPTION SAMPLE HEIGHT TEST NUMBER REMARKS:\_

NC - NOT CALCULABLE FROM DATA RECORDED EF - EQUIPMEN'T FAILURE

LOADS	NO. 3	psi	0	e. 9	11.6	16.1	20.7	25.5	36	46	55.2	64.2	73.6	83.1	78.4	72.6	63.5	28	48.6	יי י טטטט
IMBEDDED GAGE LOADS	NO. 2	isd	0	4	7.2	9.8	12.5	15.4	21.7	28.9	34.9	42.1	48.9	26	51.4	47.3	40	35.5	28.9	
IMBED	NO. 1	psi	0	3.6	7.3	11.2	14,9	19.4	29.2	39.6	47.4	57.4	65.3	74.7	69.3	63.3	53.5	47.4	38.7	
RADIAL LOADS	BOTTOM	psi	0	0.3	8.0	1.2	1.5	1.8	2.4	ლ ლ	4.2	5.4	8.9	æ	7.1	6.3	5.6	5.2	4.5	
RADIA	401	psi	0	0.1	0.9	1.1	1.1	H.	1.1	1.1	1.6	2.4	3.4	4.5	4,2	4	ထ <sub>္</sub> လ	ල. ල.	3.9	
тѕ	LVDT NO. 3	in/0,001	0	22.9	34.7	43.2	49.1	54.5	62.1	69.7	74.7	79.5	83.3	86.9	86.6	82.8	82.8	85.1	84.3	
DISPLACEMENTS	L VDT NO. 2	in/0.001	0	19.4	29.9	37.8	43.4	48.4	58.8	67.5	74	80.4	7.98	91.9	6.68	83	88.2	87.2	84.6	
0	LVDT NO. 1	in/0.001	0	35.3	53.2	64.8	73.5	81.6	93.7	102.3	109.1	114.5	119.3	123.4	122.7	122.7	121.4	121	119,3	
LOADS	CYLINDER	91	0	24	09	119	167	190	238	262	309	405	452	200	286	167	09	0	-48	
MEASURED AXIAL LOADS	FIXED	 E _e	0	371	865	1282	1921	2240	3399	4635	5778	6952	8250	9455	8652	7532	6257	5238	4048	
MEAS	MOVING	4 8	0	420	907	1344	1764	2318	3444	4754	5914	7207	8268	9946	8870	7560	6174	5116	3864	
	APPLIED LOAD	<b>K</b> -15	0	0.5	1.0	1.5	2.0	2.5	3.75	5.0	6.25	7.5	8.75	10.0	8.75	7.5	6.25	5.0	3.75	

SAMPLE DESCRIPTION	TION	Oc	
SAMPLE DENSITY:	SAMPLE DENSITY: BEFORE LOADING	(1 4)	SYMBOLS:
	AFTER LOADING		RR - NOT RECORDED
SAMPLE HEIGHT:	BEFORE LOADING	LVDT LOCATIONS	NC - HOT CALCULABLE
	AFTER LOADING	(	FROM DATA RECORDED
HEIGHT OF IMBEDI	HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SALPLE BEFORE LOADING		
REMARKS:		GAGE I OCATIONS	

2/3/66

46 (Sheet 2 of 2)

TEST NUMBER

	_	_		_	_	<b>TOOL 3000</b>			 			 	
LOADS	NO. 3	psi	39.4	33.9	28.4	22.4	14.4	3.7					
IMBEDDED GAGE LOADS	NO. 2	psi	23.5	21.3	19.3	17.2	14.4	8.4					
INBED	NO. 1	psi	29.2	24.1	18.8	13.3	6.4	0.7				-	
RADIAL LOADS	воттом	psi	3.9	3.7	3.3	3.1	2.5	1.3					
RADI	TOP	p <b>s</b> i	ぜ	4.2	4.9	5.4	5.6	3.9					
175	LVDT NO. 3	in/0.001	82.3	81.8	80.8	79.8	78.3	75.8	***************************************				
DISPLACEMENTS	LVDŢ NO. 2	in/0.001	83.6	83	83	83	83.8	84.6					
	LVDT NO. 1	in/0.001	116.9	115.9	114.5	112.5	109.1	103					
L LOADS	CYLINDER	1b	09-	-71	-71	09-	-48	0	<b></b>				
MEASURED AXIAL LOADS	FIXED PISTON	9	2874	2287	1792	1236	695	232		1-1		 	
MEA	MOVING PISTON	r,	2654	2083	1613	1092	638	235					
	LOAD	K-ih	2.5	2.0	1.5	1.0	0.5	0					
							-	_			ø		

18.5

24.1 26

36.2

37.3 34.2

20.8

24.5 27.5

36.8 39.9 38.8

37.2 41,3

28.9

52.6

2773 3078

6772

10140

5908

8746

29.4

53.6 53.6

2538 1810 1198

6363 5318

9126

10.0 8.75

7605 6490 5036

4469

54.1

8.6 11.2 13.8 16.3

IMBEDDED GAGE LOADS NC ・ NOT CALCULABLE FROM DATA RECORDED EF - EQUIPMENT FAILURE NR - NOT RECORDED NO. 2 KO. 1 13.9 20.7 Sc 10.1 17.6 SYMBOLS: BOTTOM 18.4 32.2 D S. 10.3 12.5 RADIAL LOADS L VDT LOCATIONS GAGE LOCATIONS 18.9 11.8 13.8 22.6 28.2 32.6 TOP 30/ psi 02 061 L.7DT NO.3 in/0.001 24.7 19.4 21.5 23.2 15.2 16.7 12 14 DISPLACEMENTS REMARKS. Ridge of Sand (after load applied) around edge of sample LVDT HO. 2 im/0.001 19.5 13.6 15,3 16.6 22.3 24.6 26.6 between LVDT locations rumber 1 and number 2 HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING DATE 5/6/66 LVD7 NO. 1 in/0.031 31.2 34.8 37.3 41.5 44.7 47.5 50.2 3-5/16 inches 1.76 gm/cm<sup>3</sup> 1.79 gm/cm<sup>3</sup> 0.0787 > Grains > 0.0165 inch 4 inches CYLINDER 517 705 870 1645 1222 2044 2374 MEASURED AXIAL LOADS Ottawa Sand TEST NUMBER 47 (Sheet 1 of 2) SAMPLE DENSITY: BEFORE LOADING \_\_ FIXED PISTON 242 545 AFTER LOADING\_ BEFORE LOADING 606 1515 3212 1182 2363 4878 4091 AFTER LOADING\_ 2 inches MOVING 930 6084 1352 3549 4766 7478 2298 1791 SAMPLE DESCRIPTION SAMPLE HEIGHT. APPLIED LOAD 3.75 6.25 7.5 8.75 5.0 **K**-15

NO. 3

psi

psi

EF - EQUIPMENT FAILURE
NC - NOT CALCULABLE
FROM DATA RECORDED IMBEDDED GAGE LOADS NR - NOT RECORDED NO. 2 2.6 psi NO. 1 & 0. 16.3 12.8 10.7 6.7 3.8 ps. SYKBOLS: BOTTOM 26.2 21.5 16.5 19.3 8.6 RADIAL LOADS . . 13.1 LVDT LOCATIONS GAGE LOCATIONS 23.2 21.3 18.9 16.5 13.1 **TOP** 007 L.V.DT NO. 3 in/0.001 24.8 23.8 22.6 23.1 21.7 20.8 DISPLACEMENTS LVDT NO. 2 ia/0.001 24.4 23.9 25.8 22.5 21.8 23 HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 99/9/9 L.VDT NO. 1 in/0.001 45.9 47.6 46.8 49.7 41.6 44.1 DATE CYLINDER -235 -352 -446 -470 -446 -306 9 MEASURED AXIAL LOADS 47 (Sheet 2 of 2) FIXED PISTON 2424 1515 5.15 3636 2788 2030 939 SAMPLE DENSITY: BEFORE LOADING BEFORE LOADING AFTER LOADING\_ AFTER LOADING\_ MOVING 1994 3718 2434 1014 135 1453 507 9 SAMPLE DESCRIPTION SAMPLE HEIGHT: TEST NUMBER APPLIED LOAD 3.75 REMARKS. к.-Ib 2.5 2.0

ტ დ 10.0 11.1

F390-1-68

186

Sand Ridge LVDT LOCATIONS ON HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 2 inches (0.0787 inch > Grains > 0.0165 inch) DATE 5/9/66 4-5/32 inches 4-3/32 inches SAMPLE DENSITY: BEFORE LCADING  $1.76~\mathrm{gm/cm^3}$ 48 (Sheet 1 of 2) NC Ottawa Sand SAMPLE HEIGHT: BEFORE LOADING\_ AFTER LOADING \_\_ AFTER LOADING\_ SAMPLE DESCRIPTION TEST NUMBER \_

02 တ္ထ

GAGE LOCATIONS

REMARKS. Ridge of sand (after load applied) around edge of sample

between LVDT Locations No. 1 and No. 2

SYMBOLS:

NC - NOT CALCULABLE FROM DATA RECORDED EF - EQUIPMENT FAILURE NR - NOT RECORDED

NO. 3

LOAD	NO. 3	psi	0	4.2	7.4	10.3	13.2	15.8	21.8	26.8	31.6	36.8	42.3	45.3	43.4	40.9	39.4	36.3	
IMBEDDED GAGE LOADS	NO. 2	psi	0	0.2	0.5	1.0	1.5	2.2	3.6	5.3	6.7	8.3	6.6	11	10.6	8.0	8.9	7.8	
IMBED	NO. 1	psi	0	0	0	0.1	0.2	9.0	1.2	1.5	3.1	4.2	5.5	6.3	9	5.4	4.5	3.7	
RADIAL LOADS	воттом	psi	0	-2.1	-2.8	-3.1	-2.4	-1.9	-0.3	1,7	3.4	5.5	9.7	ت: 8	9.7	6.8	5.2	ლ ლ	
RADIA	TOP	psi	0	-4.5	-6.3	-6.8	-6.3	-5.8	-4.4	-2.8	-0.7	1.4	4.2	.9	4.7	.3	1.8	0	
TS	LVDT NÓ. 3	in/0.001	0	19.4	22.7	24.8	26.3	56.9	28.8	30.7	32.3	33.9	35.6	36.4	36.4	35.8	35.6	34.2	
DISPLACEMENTS	LVDT NO. 2	in/0.001	0	32.2	40.4	46	50.5	53.6	58.8	62.3	65	67.5	69.5	71.6	71.1	68.8	68.2	8.99	
	LVDT NO. 1	in/0.001	0	40.2	51.7	62.1	67.7	72.1	77.9	82.8	85.7	88.6	7.06	93.7	93.7	93.4	92	2.06	
L LOADS	CYLINDER	IS	0	165	271	472	590	732	1156	1581	1935	2360	2761	3068	2525	1723	1109	496	
MEASURED AXIAL LOADS	FIXED	1þ	0	276	553	860	1151	1504	2241	3101	3991	4835	5772	6562	6309	5680	5127	4390	
MEA!	MOVING	lb	0	422	913	1352	1791	2366	3617	4833	6084	7453	8720	9937	8923	7658	6439	5104	
	APPLIED LOAD	K-15	0	0.5	1.0	1.5	2.0	2.5	3.75	5.0	6.25	7.5	8.75	10.0	8.75	7.5	6.25	5.0	
L			<b></b>	10.			<del></del>												_

48 (Sheet 2 of 2)\_ DATE 5/9/66 SAMPLE DESCRIPTION TEST NUMBER

SAMPLE DENSITY: BEFORE LOADING\_ AFTER LOADING \_\_

HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING \_ BEFORE LOADING\_ AFTER LOADING\_ SAMPLE HEIGHT:

LVDT LOCATIONS

GAGE LOCATIONS

REMARKS:\_

SYMBOLS:

NR - NOT RECORDED

EF - EQUIPMENT FAILUSE NC - NOT CALCULABLE FROM DATA RECORDED

		~~~		_	-	~				
LCADS	NO. 3	psi	31.8	28	25.1	22.1	18.4	10.9	2.9	
IMBEDDED GAGE LCADS	NO. 2	isd	9.9	വ	4.4	3.7	2.6	1.1	0.2	
IMBED	NO. 1	psi	2.8	1.9	1.4	0.8	0.4	-0.1	-0.4	
RADIAL LOADS	воттом	psi	1.7	-0.3	-0.8	-2.3	-3.9	-5.6	-5.2	
RADI/	TOP	psi	-1.6	-3.2	-4	-5.1	6.9-	-9.1	-7.9	
175	L VDT NG. 3	in/0,001	33.6	32.6	32.1	31.1	30.6	30.3	30.3	
DISPLACEMENTS	LVDT NO. 2	in/0.001	65.4	63.3	çç	62.3	59.9	56.2	48.4	
	LVDT NO. 1	in/0.001	89.3	88.6	87.3	85.9	82.5	76.4	62.7	
L LOADS	CYLINDER	16	0	-425	-472	-531	-519	-425	-118	
MEASURED AXIAL LOADS	FIXED	16	3684	2763	2456	2057	1504	860	230	
MEA	MOVE-46 PISTON	19	3786	2467	2028	1589	1014	439	0	
	APPLIED LOAD	K-1b	3.75	2.5	2.0	1.5	1.0	0.5	0	

TEST NUMBER 49 (Sheet 1 of 2) DATE 5/13/66

(0.0787 inch > Grains > 0.0165 inch)

SAMPLE DENSITY: BEFORE LOADING  $1.824~\mathrm{gm/cm}^3$  AFYER LOADING  $\overline{\mathrm{MC}}$ 

SAMPLE HEIGHT: BEFORE LOADING 5-23/32 inches

AFTER LOADING N.R.
HEIGHT OF IMBEDDED GAGES ABOVE SCITTOM OF SAMPLE BEFORE LOADING 2 inches

03003

GAGE LOCATIONS

SYMBOLS:

NR - NOT RECORDED EF - EQUIPMENT FAILURE

NC - NOT CALCULABLE FROM DATA RECORDED

LVDT LOCATIONS

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n powder 1	
ks. Teflor	
REMAR	

EASURED AXIAL LOADS	a	a	LVDT	1	DISPLACEMENTS	l	RADI	RADIAL LOADS	IMBED	IMBEDDED GAGE LOADS	LOADS
K-1b	PISTON 16	PISTON Ib	CYLINDER 16	NO. 1 in/0.001	NO. 2 in/0.001	NO. 3 in/0.001	10P	BOTTOM	NO. 1	NO. 2	NO. 3
0	0	0	0	0	0	0	0	0	0	0	0
0.5	400	333	47	42.4	21.2	34.3	3.2	1.8	4.8	4.8	4.7
1.0	668	788	20	64.5	34.4	51.5	0.9	3.4	10.2	8.3	8.9
1.5	1322	1212	176	79.1	45	63.6	7.2	4.5	15.1	12.0	12.9
2.0	1765	1606	211	89.3	54.7	71.7	8.0	5.4	20.9	15.7	17.1
2.5	2298	2121	234	98.9	63.1	79.5	8.5	6.3	23.4	18.5	20.8
3.75	3430	3182	304	112.5	9.92	90.5	12.1	9.4	39.6	27.5	29.5
5.0	4729	4363	445	123.4	88.2	99.0	15.6	12.2	51.7	35.8	38.9
6.25	5927	5454	515	131.5	97.1	106.1	18.9	14.7	63.7	43.0	46.8
7.5	7243	6727	629	138.3	103.8	111.7	22.2	17.5	75.8	50.6	54.5
8.75	8442	2177	725	145.1	110.7	114.1	26.0	20.2	86.7	59.0	61.8
10.0	9923	9151	889	151.0	117.6	121.2	29.4	23.2	98.4	67.4	69.4
8.75	8824	8423	655	149	116.9	120.2	27.3	21.2	92.2	63.2	64.7
7.5	7592	7272	445	149	116.3	119.2	25.6	19.7	85.1	57.9	60.8
6.25	6444	6272	257	147.1	115.6	119.2	24.7	18.4	77.4	51.8	55.2
5.0	5062	2090	164	146.1	112.8	118.2	22.5	16.9	66.3	47.4	48.6

TEST NUMBER 49 (Sheet 2 of 2) DATE 5/13/66

REMARKS:\_

LVDT LOCATIONS

GAGE LOCATIONS

SYMBOLS:

10

NR - NOT RECORDED

EF - EQUIPMENT FAILURE

NC - NOT CALCULABLE

FROM DATA RECORDED

LAADS	NO. 3	psi	42.1	33.5	29.7	25.1	19.3	11.8	3.2	
IMBEDDED GAGE LAADS	NO. 2	psi	39.7	31	29.3	25.7	23.0	19.1	6.6	
IMBEDI	NO. 1	psi	57.2	44.3	39.6	31.9	24.7	13.1	1.9	
RADIAL LOADS	воттом	psi	14.7	12	10.6	9.1	7.5	5.3	3.3	
RADIA	TOP	psi	20.0	16.7	15	13.5	11.9	10.2	7.4	
TS	LVDT NO. 3	in/0.001	116.7	114.5	113.8	113.1	112.4	109.6	106.1	
<b>UISPLACEMENTS</b>	LVDT NO. 2	in/0.001	111.4	110	109	108.6	107.3	108	108	
	LVDT NO. 1	in/0.001	145.1	143.2	141.2	139.3	137.3	134.4	125.6	
L LOADS	CYLINDER	16	23	-47	-117	-140	-117	-70	-23	
MEASURED AXIAL LOADS	FIXED	16	3939	2757	2272	1818	1273	269	242	
MEAS	MOVING	16	3863	2597	3931	1598	1249	599	167	
	LOAD	K-16	3.75	2.5	2.0	1.5	1.0	0.5	0	
<b></b>			·			***************************************				

5/16/66 SAMPLE DESCRIPTION Dense Graded Ottawa Sand \_ DATE \_ TEST NUMBER \_ 50 (Sheet 1 of 2) Š

SYMBOLS:

CULABLE TA RECORDED NT FAILURE NOT RECORDED

	AFTER LOADING NC		האה יאסיי אהרטי
SAMPLE HEIGHT:	BEFORE LOADING	LVDT LOCATIONS	NC - NOT CALCI
	AFTER LOADING_NR		FROM DAI
HEIGHT OF IMBED	HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 4 inches.		
		(0 30)	
REMARKS: T	Teflon powder lubricant used		
		GAGE LOCATIONS	

	CYLINDER 16	LVDT	LVDT	LVDT			. 01.	7	
		- - - -	NO. 2	NO. 3	T0P	воттом		7 .52	МО. 3
		in/0.001	in/0.001	in/0.001	psi	psi	p\$i	psi	psi
	0	0	0	0	0	0	0	0	0
	47	43.8	31.1	34.7	8.0	0.5	4.8	3.7	7.4
	116	65.7	46	51.2	1.1	0.8	9.0	8.9	13.6
	175	80.4	58.1	62.1	1.6	1.0	13.1	8.0	19.1
1927   1678	210	90.0	8.99	70.7	1.9	1.2	17	12.8	24.8
2400 2166	233	97.5	76.8	76.8	2.2	1.4	20.7	15.8	29.7
3634 3294	305	112.5	91.3	88.4	4.0	2.3	29.5	23.1	42.3
4833 4422	443	124.2	103.8	99.7	5.6	3.3	38.1	30.7	55.2
6135   5604	489	131.5	112.4	106.1	7.4	4.2	46.4	37	65.8
7504 6862	629	138.3	119.7	113.1	9.4	4.5	55.2	44.8	9
8746 8121	669	146.1	127.3	118.8	11.9	6.7	62	51.8	89.4
10,140   9272	816	149.0	132.2	123.2	14.4	7.8	69.3	58.2	7.96
8991   8540	909	149	132.9	123.2	12.7	6.8	65.0	53.9	2.06
7706   7320	419	148	131.5	120.9	11.8	6.3	60.2	49.3	85.2
6464 6268	256	147.1	130.8	120.2	11.2	5.7	54.2	44.5	77.8
5104 4972	140	146.1	129.8	118.8	10.2	5.2	47.4	39.1	6.89

NR . NOT RECORDED SYMBOL S: LVDT LOCATIONS HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING DATE 5/16/66 TEST NUMBER 50 (Sheet 2 of 2) SAMPLE HEIGHT: BEFORE LOADING\_ SAMPLE DENSITY: BEFORE LOADING ... AFTER LOADING\_ AFTER LOADING\_ SAMPLE DESCRIPTION REMARKS.\_

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NC - NOT CALCULABLE FROM DATA RECORDED EF - EQUIPMENT FAILURE GAGE LOCATIONS

_		7	····			«——				 ٦٣
LOADS	NO. 3	psi	59.8	47.9	42.1	36.2	27.3	17.9	6.3	F390-1-68
IMBEDDED GAGE LOADS	NO. 2	psi	33.4	25.5	22.0	19.1	15.3	13.2	7.8	
IMBED	NO. 1	psi	40.5	31.4	27.5	22.7	16.6	9.2	2.0	
RADIAL LOADS	воттом	psi	4.6	3.7	3,4	3.0	2.6	2.1	1.7	
RADI	TOP	psi	8.5	7.4	6.4	6.1	5.6	5.4	4.5	
4TS	LVDT NO. 3	in/0.001	118.4	115.2	114.5	114.5	113.1	111.7	107.8	
DISPLACEMENTS	LVDT NO. 2	in/0.001	126.6	124.6	124.6	123.2	122.5	122.5	122.5	
	LVDT NO. 1	in/0.001	146.1	144.2	143.2	140.3	138.3	136.4	129.1	
r LOADS	CYLINDER	15	23	-23	-58	-70	-70	-70	-47	
MEASURED AXIAL LOADS	FIXED	2	3904	2684	2196	1769	1220	671	244	1
MEA	MOVING	19	3819	2603	2028	1606	1014	809	169	
	APPLIED LOAD	K-lb	3.75	2.5	2.0	1.5	1.0	٦.5	0	

HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 2 inches 5/17/66 (0.0787 inch > Grains > 0.0165 inch) DATE SAMPLE HEIGHT: BEFORE LOADING 6-3/32 inches SAMPLE DENSITY: BEFORE LOADING  $1.69~\mathrm{gm/cm}^3$ SAMPLE DESCRIPTION Graded Ottawa Sand NR NR TEST NUMBER 51 (Sheet 1 of 2) AFTER LOADING\_\_\_ AFTER LOADING\_

1 2 3 OVDT LOCATIONS	300
LVDT	40

GAGE LOCATIONS

Teflon powder lubricant used

REMARKS:\_\_\_\_

NC - NOT CALCULABLE FROM DATA RECORDED EF . EQUIPMENT FAILURE NR - NOT RECORDED SYMBOLS:

MOVING P				<u></u>						
	FIXED PISTON	CYLINDER	LVDT NO. 1	LVDT NO. 2	LVDT NO. 3	T0P	воттом		NO 2	NO. 3
	-P	16	in/0.001	in/0.001	in/0.001	psi	psi	psi	psi	psi
0	0	0	0	0	0	0	0	0	0	0
441	375	47	43.2	33.2	43	2.1	1.4	9	3.9	7.9
949	875	116	62.2	44.3	9.09	3.9	2.4	11.6	6.9	15.5
1390   1	1250	210	76.7	54.7	74.7	101 111	က	17.2	6.6	22.1
1864   1	1750	233	88.9	65.7	85.3	5.2	3.7	23.2	12.8	28.2
2373 2	2188	256	95.5	72.7	91.9	6.2	4.3	27.9	15.8	34.2
3644 3	3359	408	139.8	88.2	105.3	8.3	5.8	40.2	22.2	47.3
4915 4	4531	489	122.7	6.66	114.5	10.5	7.5	52.6	30.3	60.8
5763   5	5344	582	129.1	109.3	120.2	12.3	8.6	61.5	34	69.4
7500   6	6797	746	138.3	118.3	128	15.5	11.2	76.7	44.2	84.7
8238 7	7500	874	145.1	124.6	134.3	17.1	12.1	85.1	48.5	91
8 2966	8938	066	151.9	132.9	141.4	21.3	14.2	66	57.7	108.5
8339   7	7938	641	150	132.9	141.4	19.7	13.1	92.2	51.4	100.6
7068   6	6703	443	149	131.5	140.4	17.8	12.2	83.6	47.4	89.9
6000   5	5938	233	148	131.5	140.4	17.1	11.6	76.7	42.9	84.7
5085   5	5156	163	148	130.1	139.4	16.1	10.9	70.2	39.1	78.9

56.2 50 43.4 34.2 20.8 66.3 IMBEDDED GAGE LOADS NOT CALCULABLE FROM DATA RECORDED EF - EQUIPMENT FAILURE NR - NOT RECORDED NO. 2 26.9 23.3 20.8 18.2 32.5 14.5 ps. FT FT 48.9 28.4 NO. 1 59.4 42.7 37.1 ps. SYMBOLS: BOTTOM 7.4 6.7 6 4.4 RADIAL LOADS P. œ LVDT LOCATIONS GAGE LOCATIONS TOP 8.4 .<u>.</u> 13.6 12.3 10.3 10.5 10.3 LVDT NO. 3 in/0.001 129.3 138.4 134.3 134.3 133.3 131.3 127.3 DISPLACEMENTS LVDT NO. 2 in/0.001 124.6 124.6 123.9 123.2 123.2 125.3 128 HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING LVDT NO. 1 in/0.001 136.4 133.9 144.2 143.2 139.3146.1 145.1 DATE CYLINDER -47 -70 -116 -140 -140 -93 9 MEASURED AXIAL LOADS 51 (Sheet 2 of 2) FIXED PISTON BEFORE LOADING SAMPLE DENSITY: BEFORE LOADING. AFTER LOADING\_ AFTER LOADING\_ 3812 2812 1312 688 1844 2281 MOVING 2610 2034 542 1593 1017 3627 SAMPLE DESCRIPTION SAMPLE HEIGHT: TEST NUMBER APPLIED LOAD 3.546 **₹** REMARKS: 2.5 2.0 1.5 1.0 0.5

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LVDT LOCATIONS inches HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING  $\overline{4}$ (0.0787 inch > Grains > 0.0165 inch) DATE  $\frac{5/18/66}{}$ BEFORE LOADING 6-3/32 inches SAMPLE DENSITY: BEFORE LOADING  $1.69~\mathrm{gm/cm}^3$ Graded Ottawa Sand AFTER LOADING NR (Sheet 1 of 2) AFTER LOADING\_ TEST NUMBER 52 SAMPLE DESCRIPTION SAMPLE HEIGHT:

SYMBOLS.

NR - NOT RECORDED

EF - EQUIPMENT FAILURE

NC - NOT CALCULABLE

FROM DATA RECORDED

GAGE LOCATIONS

Teflon powder lubricant used

REMARKS:\_

490	MEA	MEASURED AXIAL LOADS	AL LOADS		DISPLACEMENTS	1TS	RADÍA	RADIAL LOADS	IMBED	IMBEDDED GAGE LOADS	LOADS
LOAD	MOVING	FIXED PISTON	CYLINDER	LVDT NO. 1	LVDT NO. 2	LVDT NO. 3	ТОР	воттом	NO. 1	NO. 2	NO. 3
K-1b	4	9	ą.	in/0.001	in/0.001	in/0.001	psi	î <b>s</b> C	psi	psi	psi
0	0	0	0	O	0	0	0	0	0	0	0
0.5	424	344	58	43.2	31	36.4	1.8	1.4	4.1	3.6	5.3
1.0	915	844	140	66.2	45.4	54.3	2.9	2.5	8.4	9.9	9.5
1.5	1356	1250	187	79.1	55.4	65.6	4.2	3.7	13	9.5	13.1
2.0	1864	1719	234	90.1	64	75.2	9	4.9	17.2	12	17
2.5	2339	2188	257	100.2	72.1	83.1	7.4	6.2	22.5	15.4	20.5
3.75	3526	3375	398	114.5	87.2	97.2	10.5	8.8	34	21.7	28.9
5.0	4780	4469	491	123.4	98.1	106.8	13.8	11.4	43.3	28.9	37.1
6.25	0009	5562	644	133.9	105.9	113.1	16.8	13.8	57.8	34.6	44.4
7.5	7170	6562	725	138.3	114.2	120.2	19.7	15.9	52.4	415	51.2
8.75	8644	7922	936	145.1	121.8	124.2	23.3	18.7	8.69	48.2	58.9
10.0	10,034	0006	994	149	128	131.3	8.92	21.2	80.5	54.8	67.3
8.75	8814	8125	725	149	128	131.3	25.2	20.2	76.1	50.6	62.1
7.5	7628	7266	515	148.5	128	131.3	23.6	19.2	72.4	47	57.1
6.25	6229	6094	292	148	125.9	129.3	22.5	18	63.9	42.1	50.8
5.0	2000	5031	211	146.1	124.6	128.8	20.5	16.3	56.8	36 1	43.7
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IMBEDDED GAGE LOADS NR - NOT RECORDED NO. 2 30.9 24.9 22.3 19.6 13.5 7.8 17 33.9 49.2 46.2 28.8 21.9 KO. 7 <u>P</u>8. 13 SYMBOLS: BOTTOM 14.8 11.8 10.3 8.8 6.9 4.7 RADIAL LOADS p & GAGE LOCATIONS LVDT LOCATIONS TOP 13.4 11.6 pš 18.4 80 10 15 00 LVDT NO. 3 in/0.003 123.2 121.2 122.2119.2 128.3 123.7 DISPLACEMENTS L.YDT HO. 2 im/0.001 124.6 122.8 121.8 119.7 119.7 HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING FATE 5/18/66 in/0.001 143.2 136.4 133.4 144.2 141.2 138.8 146.1 LVDT NO. 1 CYLINDER -140 -70 23 -23 -58 -117 -164 MEASURED AXIAL LOADS TEST NUMBER 52 (Sheet 2 of 2) FIXED PISTON 1b BEFORE LOADING SAMPLE DENSITY: BEFORE LOADING AFTER LOADING\_ 2844 1875 1312 781 281 2344 AFTER LOADING. 4031 2610 1085 610 102 MOVING PISTON 16 2034 1593 3797 SAMPLE DESCRIPTION SAMPLE HEIGHT: APPLIED LOAD 3.75 2.5 1.0 0.5 4-!₽ 5.5 REMARKS:

27.6 23.4 18.9

36.8

NO. 3

8.4

14

196

`o6

EF - EQUIPMENT FAILURE

NC - NOT CALCULABLE FROM DATA RECORDED

NO. 3 3.9 NC - NOT CALCULABLE FROM DATA RECORDED IMBEDDED GAGE LOADS EF - EQUIPMENT FAILURE NK - NOT RECORDED NO. 2 1.8 D S. NO. 1 10.2 ps SYMBOLS: BOTTOM 16.2 RADIAL LOADS . 5 LVDT LOCATIONS GAGE LOCATIONS TOP 17.3 pai 0 L. VDT NO. 3 in/0.001 瓦瓦 0 HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 2 inches DISPLACEMENTS LVDT NO. 2 in/0.001 (0.0787 inch > Grain Size > 0.0165 inch) सञ 0 DATE \_5/27/66 LVDT NO. 1 in/0.001 0 Load applied by hyge shock tester 4-7/32 inches SAMPLE DENSITY: BEFORE LOADING 1.78 gm/cm<sup>3</sup>  $1.79 \,\mathrm{gm/cm}^{3}$ 4-1/4 inches CYLINDER Graded Ottawa Sand 1305 4 MEASURED AXIAL LOADS AFTER LUADING\_\_\_\_ FIXED PISTON AFTER LOADING\_ SAMPLE HEIGHT: BEFORE LOADING. 2145 0 MOVING 3181 9 SAMPLE DESCRIPTION 53 TEST NUMBER APPLIED LOAD REMARKS:\_ K-16

pa.

0

F390-1-68 NO. 3 psi NC - NOT CALCULABLE FROM DATA RECORDED IMBEDDED GAGE LOADS 20 0 EF - EQUIPMENT FAILURE NR - NOT RECORDED NO. 2 psi 0 15 NO. 1 psi 0 15 SYMBOLS: BOTTOM RADIAL LUADS psi 0 <u>~</u> LVDT LOCATIONS GAGE LOCATIONS **T0P** ps: 30 2 0 L.VDT NO. 3 in/0.001 0 121 DISPLACEMENTS HEIGHT OF IMBEDDED GAGES ABOVE BOTTOM OF SAMPLE BEFORE LOADING 2 inches LVDT NO. 2 in/0.001 135 (0.6737 inch > Grain Size > 0.0165 inch)0 DATE 6/2/66 LVDT NO. 1 in/0.001 109 BEFORE LOADING 6-7/32 inches Load applied by hyge shock tester SAMPLE DENSITY: BEFORE LOADING  $1.69~\mathrm{gm/cm}$ CYLINDER **Graded Ottawa Sand** 1900 0 REMARKS. Teflon lubricating layer used 2 MEASURED AXIAL LOADS NC RH FIXED PISTON AFTER LOADING. AFTER LOADING. 7800 MOVING 9900 0 4 SAMPLE DESCRIPTION 56 SAMPLE HEIGHT: TEST NUMBER APPLI2D Load ₹. Б

NC · MOT CALCULABLE FRUM DATA RECORDED IMBEDDED GAGE LOADS EF - EQUIPMENT FAILURE NR - NOT RECORDED 114.5 43.4 NO. 2 , s NO. 1 114.5 44.5 psi SYMBOLS: BOTTOM 12.5 5.7 RADIAL LOADS ρšί Scale LVDT LOCATIONS tester; first data line is peak loading, second line is steady GAGE LOCATIONS **TOP** psi Off 40 in/0.001 LVDT NO. 3 0 133 129 REMARKS. Teflon lubricating layer used; load applied by hyge shock DISPLACEMENTS HEIGHT OF IMBEDDED GAGES ABOVE BO'LTOM OF SAMPLE BEFORE LOADING 2 INCHES L VDT NO. 2 in/0.001 155 150 0 (0.0787 inch > Grain Size > 0.0165 inch) DATE  $\frac{6/3/67}{}$ in/0.001 LVDT NO. 1 144 138 0 5-15/16 inches BEFORE LOADING 6-1/8 inches SAMPLE DENSITY: BEFORE LOADING 1.68 gm/cm CYLINDER Graded Ottawa Sand 1360 468 MEASURED AXIAL LOADS AFTER LOADING NC 14,350 5200 FIXED PISTON AFTER LOADING... 5920 14,500 MOVING SAMPLE DESCRIPTION 57 state TEST NUMBER SAMPLE HEIGHT APPLIED LOAD **₹**-15

165.0 70.5

0

ХO. 3 psi

NO. 3 52.6 22.0 IMBEDDED GAGE LOADS psi 0 NC - NOT CALCULABLE FROM DATA RECORDED EF - EQUIPMENT FAILURE NR - NOT RECORDED NO. 2 0 33.5 11.3 28. NO. 1 ps 25.0 65.0 0 SYMBOLS: BOTTOM 33.0 0 60.4 , t d RADIAL LOADS L VDT LOCATIONS GAGE LOCATIONS TOP \operation 20 ps. सञ 되 0 30 in/0.001 LVDT NO. 3 42 34 0 HEIGHT OF IMBEDDED GAGES ABOVE BOTTCM OF SAMPLE BEFORE LOADING 2 inches DISPLACEMENTS LVDT NO. 2 in/0.001 REMARKS. Load applied by hyge shock tester; first data line is 0 55 50 (0.0787 inch > Grain Size > 0.0165 inch) in/0.001 LVDT NO. 1 peak loading, second line is steady state 50 42 0 BEFORE LOADING 6-1/32 inches  $1.78~\mathrm{gm/cm}^3$ CYLINDER Graded Ottawa Sand 936 5880 <u>•</u> MEASURED AXIAL LOADS NR NR SAMPLE DENSITY: BEFORE LOADING ... AFTER LOADING \_\_\_ FIXED PISTON AFTER LOADING\_ 0 4660 10,950 <u>.0</u> MOVING 0 5590 15,300 2 SAMPLE DESCRIPTION SAMPLE HEIGHT: APPLIED Load **7**-15

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DATE <u>6/3/66</u>

20

TEST NUMBER